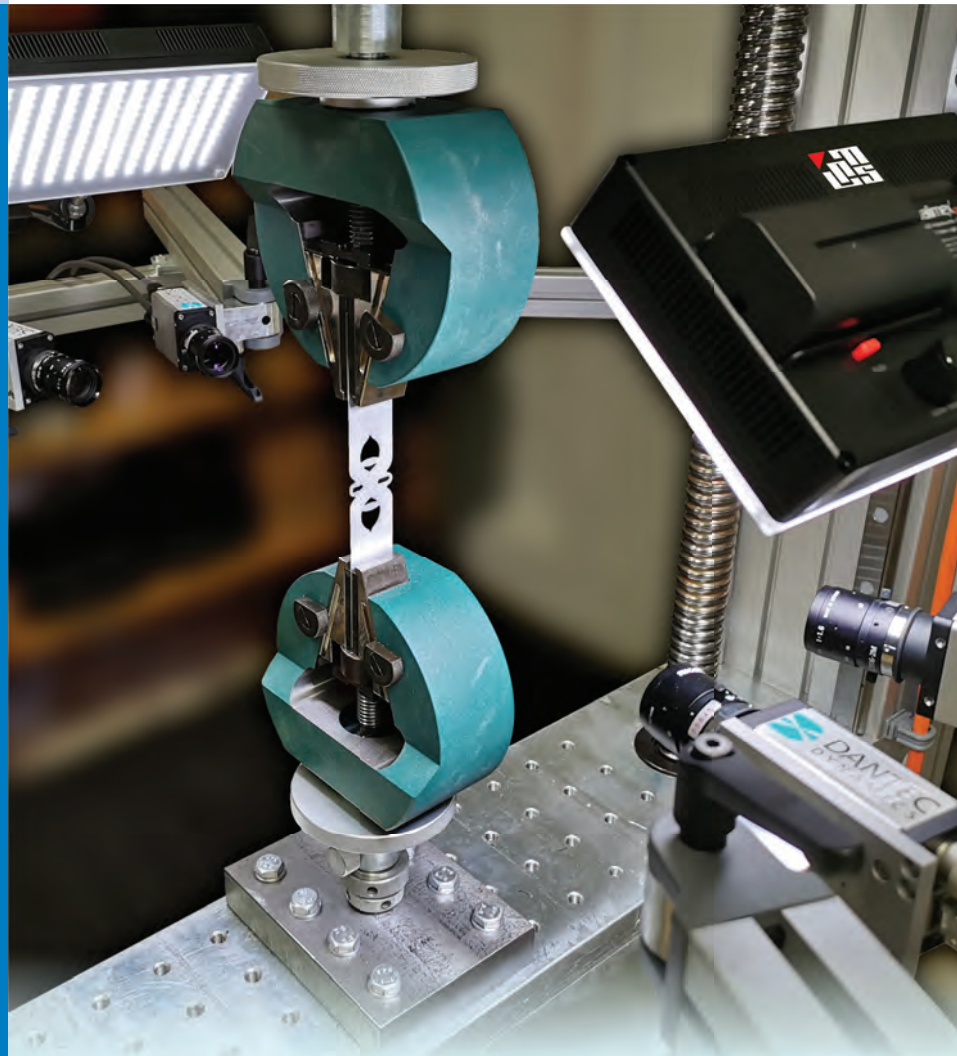


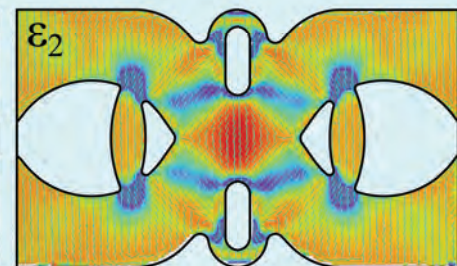
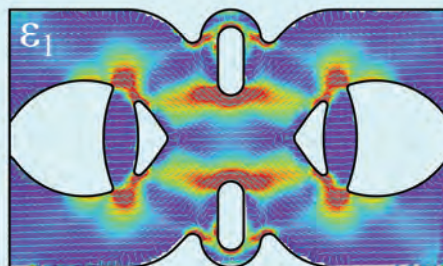


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The international journal publishes original and (mini)review articles covering the concepts of materials science, mechanics, kinematics, thermodynamics, energy and environment, mechatronics and robotics, fluid mechanics, tribology, cybernetics, industrial engineering and structural analysis.

The journal follows new trends and progress proven practice in the mechanical engineering and also in the closely related sciences as are electrical, civil and process engineering, medicine, microbiology, ecology, agriculture, transport systems, aviation, and others, thus creating a unique forum for interdisciplinary or multidisciplinary dialogue.

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Cover:
The cover image shows a digital image correlation (DIC) measurement system, which can acquire heterogeneous strain fields (shown on the bottom). By using an inverse identification procedure, constitutive model parameters can be determined by comparing the measured strain fields with the calculated ones. However, prior to the procedure, proper spatial alignment of both data needs to be ensured.

Image courtesy:
University of Ljubljana,
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Contents

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Papers

Andraž Maček, Janez Urevc, Miroslav Halilovič: Flat Specimen Shape Recognition Based on Full-Field Optical Measurements and Registration Using Mapping Error Minimization Method	203
Abdulbasit Mohammed, Hirpa G. Lemu, Belete Sirahbizu: Determining Optimum Rotary Blade Design for Wind-Powered Water-Pumping Systems for Local Selected Sites	214
Mingxing Han, Yinshui Liu, Yitao Liao, Shucui Wang: Investigation on the Modeling and Dynamic Characteristics of a Novel Hydraulic Proportional Valve Driven by a Voice Coil Motor	223
Mohsen Soori, Mohammed Asmael: Virtual Minimization of Residual Stress and Deflection Error in the Five-Axis Milling of Turbine Blades	235
Mário Santos, Jaime Santos: Ultrasonic Scattering Attenuation in Nodular Cast Iron: Experimental and Simulation Studies	245
Ivan Okorn, Marko Nagode, Jernej Klemenc: Operating Performance of External Non-Involute Spur and Helical Gears: A Review	256

Flat Specimen Shape Recognition Based on Full-Field Optical Measurements and Registration Using Mapping Error Minimization Method

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In the paper, an alignment methodology of finite element and full-field measurement data of planar specimens is presented. The alignment procedure represents an essential part of modern material response characterisation using heterogeneous strain-field specimens. The methodology addresses both the specimen recognition from a measurement's image and the alignment procedure and is designed to be applied on a single measurement system. This is essential for its practical application because both processes, shape recognition and alignment, must be performed only after the specimen is fully prepared for the digital image correlation (DIC) measurements (white background and black speckles) and placed into a testing machine. The specimen can be observed with a single camera or with a multi-camera system. The robustness of the alignment method is presented on a treatment of a specimen with a metamaterial-like structure and compared with the well-known iterative closest point (ICP) algorithm. The performance of the methodology is also demonstrated on a real DIC application.

Keywords: full-field measurements, digital image correlation (DIC), specimen shape recognition, surface registration, iterative closest point (ICP)

Highlights

- A methodology is proposed for aligning FEA data and the full-field measurement data of planar specimens.
- The methodology offers both the shape recognition and alignment process to be performed on a single full-field measurement system.
- The main advantage of the proposed methodology is its robustness.
- The performance of the methodology is presented on synthetic as well as real DIC data.
- The measured specimen can be observed with a single camera or a multi-camera system.

0 INTRODUCTION

Advanced constitutive models allow precise adjustment of the material mechanical response to specific loading conditions. Their flexibility is a result of a large number of free parameters. However, as the number of parameters increases, information gathered from standard experiments becomes insufficient or a huge amount of different experiments is required [1] and [2]. An alternative to the approach is taking into account the full-field kinematic information. Such data are normally acquired through digital images from where the displacement fields can be calculated using the digital image correlation (DIC) techniques. The approach gives rise to the development of complex specimen shapes [3], making it possible to identify material parameters through a single experiment [4]. One of the challenges, still not sufficiently resolved and addressed in this work, is the alignment of planar specimens between the modelling (numerical) data and experimental DIC data.

Full-field measurements offer a huge amount of information on the specimen surface (several 10'000), especially compared to classical extensometers.

However, their dependency on the sought material parameters generally cannot be explicitly determined, and therefore direct material identification cannot be executed. Researchers resort to the use of inverse identification techniques, where the finite element model updating (FEMU) [5] represents one of the established methods with extensive research work performed on the full-field measurements [3] (e.g. DIC [6] response calculation or by employing different optimization methods (e.g. genetic algorithm [7] or simulated annealing [8])). The method is based on an iterative comparison between the measured and calculated specimen's responses. The optimal values of material parameters are then determined by minimizing their discrepancy.

However, before finite element model (FEM) results and DIC data can be compared, coordinate systems of both sets of data need to be aligned. The latest research shows that such alignment is essential for reliable identification of material parameters in the case of specimens with non-homogeneous strain field where sharp strain gradients occur. Fehervary et al. [9] examined material parameter fitting results of planar tests when the sample orientation was unknown

or deviated. The study showed that after a certain threshold of misalignment, reliable parameters can no longer be found. The authors also concluded that the level of threshold seemed to be material dependent. Lava et al. [10] studied two different methods used to compute full-field error maps between experimental DIC data and FEA data to, among others, investigate the effects of processing parameters, model form error (such as incorrect boundary conditions) and mesh alignment. The authors showed that even a small misalignment can have a surprisingly large effect on the strain error maps and exposed the necessity to develop robust and precise methods of alignment. Similarly, Ruybalid et al. [4] performed various virtual test cases to assess the performance of FEMU and integrated digital image correlation (IDIC) method when subjected to different error sources, among others, misalignment of the specimen. Both methods are shown to be sensitive to misalignment. Namely, the increase of the misalignment further increases the error on the identified material parameters.

It seems interesting that despite the importance of properly addressing the alignment of DIC and FEM data and considering the numerous publications addressing material characterization using planar specimens, not much information can be found on how researchers align both data sets in their works. However, as pointed out in a recent article from Polyga (a developer of 3D scanners with more than 10 years of experience), scanning flat objects can be particularly difficult even for experienced 3D scanning technician [11]. The process of aligning DIC in FEM data set generally consists of two steps, specimen's edge detection and alignment (point set registration). It is important to note that the procedure needs to be performed on the same optical system as DIC measurements so that the DIC and FEM data can then be compared, which adds to the complexity of the problem.

One of the best-known methods for point sets registration is the iterative closest point (ICP) algorithm introduced by Besl and McKey [12]. The algorithm consists of the closest point search and a minimization of the matching error, applied iteratively to the two surfaces to be matched [13]. Many variants of the algorithm have been introduced since, affecting the algorithm at different stages [14] and [15], e.g. the selection of points, matching, rate of convergence, etc. However, despite the widespread of the ICP method, its convergence in the general case of initial misalignment cannot be guaranteed [16]. Also, optical full-field displacement measuring methods most

commonly cannot accurately measure the specimen shape.

To address the problem of aligning full-field measurements with FEM data, Bruno et al. [17] utilized a linear transformation matrix to map the location of the calculation point into the measurement picture pixel position. The projection matrix was determined using user input coordinates of the calculation point and the pixel positions of three arbitrary points. The spatial position of all three points must be known a priori and all of them must be observed by the measurement. A similar approach was employed by Silva [18] who transformed the model coordinate system by recognizing that some reference points observed in the image have known numerical coordinates. Both calibration procedures are easy to implement but with the state of the art specimen designs, which exhibit smooth shapes and no dominant features [19] they become increasingly difficult to be used and prone to errors.

In the present work, a robust methodology is presented for aligning full-field measurements (e.g. DIC) and modelling (e.g. FEM) data of flat specimens. The procedure addresses both the contour recognition of a specimen as well as registration. Namely, for a successful DIC measurement, the specimen must be covered with a speckle pattern consisting of a white background and black speckles. In practice, this represents a problem because the distinction between the specimen and the background can quickly be lost. We determine the specimen's geometry by taking two consecutive images with changing the background illumination. From their comparison, the background can be subtracted, leaving just the specimen's geometry. The result is a black-white image of the measured specimen (white) and the background (black). To perform the alignment, the CAD model is also projected onto the image as black and white pixels. By minimizing the intersection between both images, i.e. the measured specimen and numerical model, both data sets are aligned.

In the following, the methodology is presented first in the case of 2D and then 3D measurements. The performance of the method is demonstrated on two examples. In the first example, a synthetic one, we demonstrated the method on a problem of aligning a specimen with a metamaterial-like structure. In the example, we also analyse the effect of misalignment and compare the performance of the method with the ICP. In the second example, the method is demonstrated on real DIC data where its ability to handle the presence of noise and mismatches in the geometry is also presented. Discussion of the

results and the performance of the method is given in the Discussion section, which is followed by the conclusion.

1 METHODS

The method will be presented first in the case of 2D DIC measurements, where the specimen is observed with a single camera, located normal to its front surface (Fig. 1a). Afterwards, the method will be generalized for the case of 3D DIC measurement. Although in the latter case, additional coordinate transformations are needed to take into account the specimen's perspective (Fig. 1b), the alignment procedure is in both cases the same.

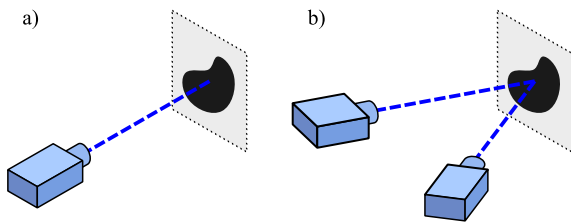


Fig. 1. Camera arrangement for a) 2D, and b) 3D DIC measurement

1.1 2D measuring System

Specimen recognition. Let us denote the numbering of pixels in the acquired (DIC) image in a form of a set:

$$P = \{(u, v) \in \mathbb{N} \times \mathbb{N} \mid u \leq n \text{ and } v \leq m\}, \quad (1)$$

with the size of the image being $n \times m$ pixels.

To determine the set of pixels which represent the specimen surface, two images of the same specimen are needed. The first image is a regular one used for the measurement itself. The background on the acquired image is usually dark and cannot be clearly distinguished from the specimen. For the second image, we brighten up the background using an illumination panel, as shown schematically in Fig. 2 by the two left-most images (corresponding to specimen recognition).

The region of the specimen, which we wish to determine, is defined by a set $S_m \subset P$, which is a set of pixels, whose difference in grayscale values between both images is below a threshold value ε :

$$S_m = \{(u, v) \in P \mid |g_1(u, v) - g_2(u, v)| < \varepsilon\}, \quad (2)$$

where $g_1(u, v)$ and $g_2(u, v)$ represent grayscale values of the $(u, v)^{\text{th}}$ pixel in the individual image, respectively.

Alignment. First, positions of pixels (u, v) of the specimen's shape (set S_m) need to be transformed into a physical location associated with specimen dimensions (X, Y) , see Fig. 2. This is performed with a calibration procedure, which can be written in a form of an arbitrary function c

$$c: \mathbb{N}^2 \rightarrow \mathbb{R}^2; (u, v) \mapsto (X, Y). \quad (3)$$

By assuming that the position of the camera is normal to the specimen surface and neglecting the optical distortions [20], the calibration only scales (u, v) as

$$\begin{Bmatrix} X \\ Y \end{Bmatrix} = c(u, v) = k \begin{Bmatrix} v \\ n-u \end{Bmatrix}, \quad (4)$$

with k being the scaling factor.

The relation between coordinates (X, Y) and the modelling space coordinates (x, y) can be written in a form of a mapping function f

$$f: \mathbb{R}^2 \rightarrow \mathbb{R}^2; (X, Y) \mapsto (x, y), \quad (5)$$

where the mapping, in general, carries out translational and rotational rigid body transformation only. For a planar case, we can write the mapping as

$$\begin{Bmatrix} x \\ y \end{Bmatrix} = f(X, Y) = \begin{bmatrix} \cos \alpha & -\sin \alpha \\ \sin \alpha & \cos \alpha \end{bmatrix} \begin{Bmatrix} X \\ Y \end{Bmatrix} + \begin{Bmatrix} \Delta x \\ \Delta y \end{Bmatrix}, \quad (6)$$

with α , Δx and Δy being the angle of rotation, translation in x and y direction, respectively. All three parameters $(\alpha, \Delta x, \Delta y)$ are unknowns and must be determined in such a manner that the modelling specimen, being in (x, y) space, coincides with the measured one, being in (X, Y) space.

So far, from the calibration and mapping procedure, one can determine each pixel location in the modelling space as $(x, y) = (f \circ c)(u, v)$. If we denote the specimen region in the (x, y) space as Ω_s (see Fig. 2), the set of pixels that are located inside the region Ω_s , denoted as a set $S_s \subset P$, follows as

$$S_s = \{(u, v) \in P \mid (f \circ c)(u, v) \in \Omega_s\}. \quad (7)$$

By comparing both images, the mismatch between the recognized measured and modelled specimen shape is determined as the symmetric difference between the two pixel sets S_m and S_s

$$E = S_m \Delta S_s. \quad (8)$$

Thus, E contains all the pixels of both sets S_m and S_s except for the ones in their intersection. The number of those pixels, i.e. the cardinality of the error set $|E|$, represents the measure of the mismatch level.

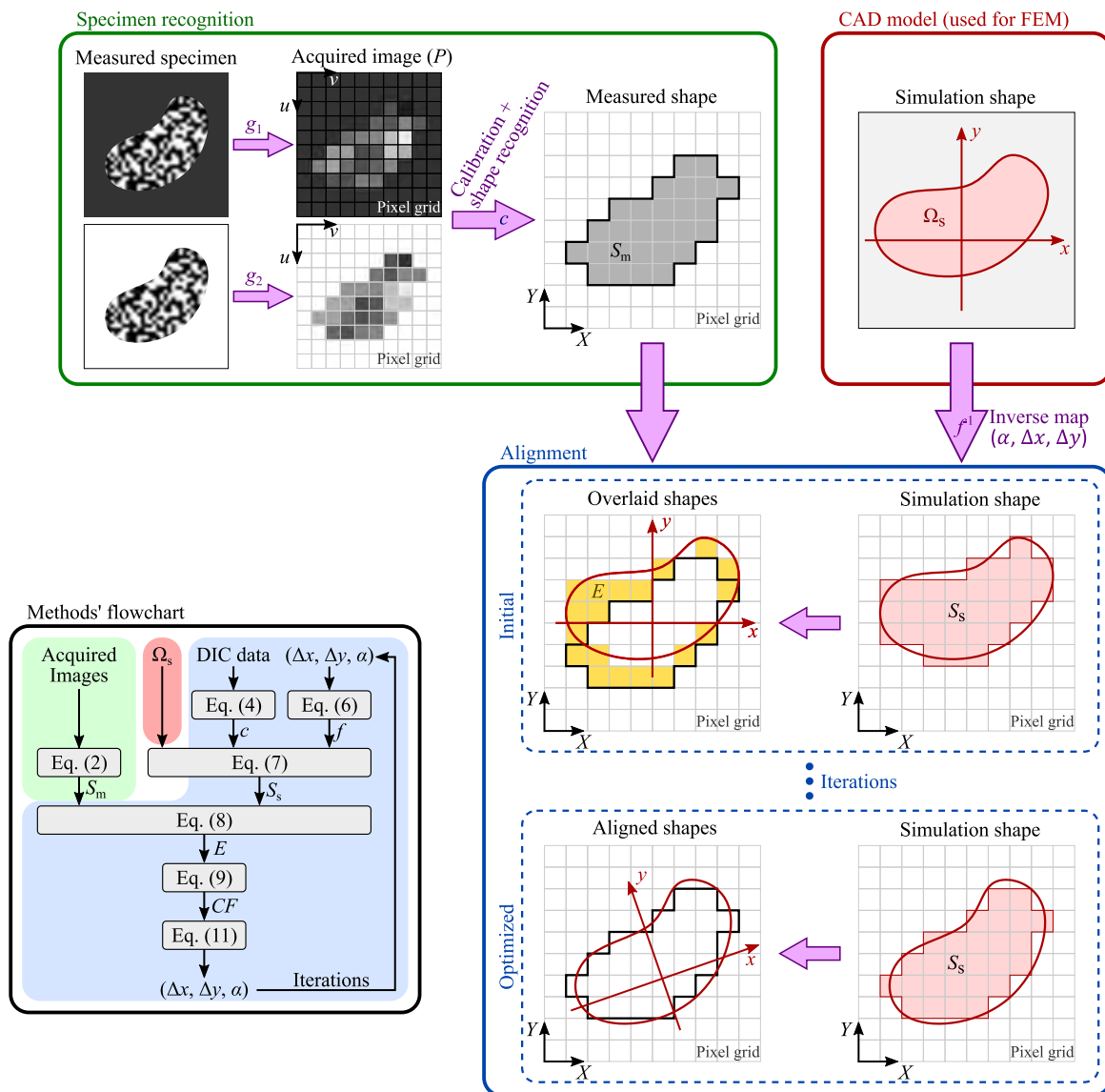


Fig. 2. The proposed alignment method

If the shapes are perfectly aligned, there will be no mismatched pixels, thus the cardinality of the error set would be zero.

The unknown mapping parameters $(\alpha, \Delta x, \Delta y)$ are determined as a solution to the following unconstrained optimization problem. Let us define the cost function $CF(\alpha, \Delta x, \Delta y)$ as

$$CF(\alpha, \Delta x, \Delta y) = |E|^2, \tag{9}$$

which needs to be minimized

$$\min CF(\alpha, \Delta x, \Delta y). \tag{10}$$

The result of the iterative optimization procedure (e.g. steepest descent method, Levenberg-Marquard algorithm) is a set of optimal values of mapping parameters $\{\hat{\alpha}, \hat{\Delta x}, \hat{\Delta y}\}$ that minimize the cardinality of the error set

$$\{\hat{\alpha}, \hat{\Delta x}, \hat{\Delta y}\} = \arg \min_{\{\alpha, \Delta x, \Delta y\}} CF(\alpha, \Delta x, \Delta y). \tag{11}$$

Once the optimal values of parameters are obtained, the measured results can be mapped into the modelling space via Eq. (6) as

$$\begin{Bmatrix} \hat{x} \\ \hat{y} \end{Bmatrix} = f(X, Y) = \begin{bmatrix} \cos \hat{\alpha} & -\sin \hat{\alpha} \\ \sin \hat{\alpha} & \cos \hat{\alpha} \end{bmatrix} \begin{Bmatrix} X \\ Y \end{Bmatrix} + \begin{Bmatrix} \hat{\Delta x} \\ \hat{\Delta y} \end{Bmatrix}. \tag{12}$$

1.2 3D Measuring System

The 3D measuring system consists of several digital cameras, each of them observes the specimen from its direction, as presented in Fig. 1b. The result of a 3D DIC measurement is a point cloud given in a measurement coordinate system $(\tilde{X}, \tilde{Y}, \tilde{Z})$, Fig. 3, which in general is not aligned with the specimen. In the case of planar specimens, a best-fit plane is normally constructed over the point cloud. The coefficients of the best-fit plane are determined using linear regression. Consequently, the location of the measured points in 3D $(\tilde{X}, \tilde{Y}, \tilde{Z})$ can be expressed with the 2D location on the best-fit plane (X, Y) . The reduction to 2D is essential to enable the use of equations derived in the previous subsection.

Calibration procedure c_i is performed for each camera with the purpose to map the pixels of the acquired images into the (X, Y) coordinate system. Each camera acquires its image with the corresponding pixel set P_i , $i = \{1, 2, \dots, N\}$ (analogously to Eq. (1)) with N being the number of cameras.

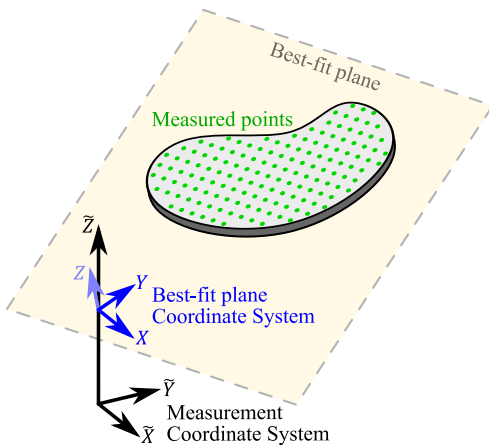


Fig. 3. Best-fit plane

The calibration procedure for each camera c_i is approximated using a linear transformation

$$\begin{cases} X \\ Y \end{cases} = c_i(u, v) = \begin{bmatrix} k_{uX,i} & k_{vX,i} \\ k_{uY,i} & k_{vY,i} \end{bmatrix} \begin{Bmatrix} u \\ v \end{Bmatrix} + \begin{Bmatrix} n_{X,i} \\ n_{Y,i} \end{Bmatrix}; \quad (13)$$

$(u, v) \in P_i$,

where k_* and n_* represent the constants of a two-dimensional linear function. Constants are obtained for each camera individually by employing two linear regressions. First regression calculates $k_{uX,i}$, $k_{vX,i}$ and $n_{X,i}$ by considering the known pixels' positions (u, v) and the corresponding position on the X axis of the best-fit plane. The second regression analogously

calculates $k_{uY,i}$, $k_{vY,i}$ and $n_{Y,i}$ by considering the known pixels' positions (u, v) and the corresponding position on the Y axis of the best-fit plane.

The mapping f of measured points on the best-fit plane (X, Y) to the modelling space (x, y) is performed as presented in the 2D case via Eq. (6). Sets $S_{m,i}$, $S_{s,i}$ and E_i , which again represent sets of pixels of the measured specimen, modelling specimen and the mismatch between the two, respectively, are constructed for each camera individually as already presented.

The level of the mismatch considering all cameras is expressed as the sum of all individual levels of mismatches squared

$$CF(\alpha, \Delta x, \Delta y) = \sum_{i=1}^N |E_i|^2, \quad (14)$$

with $CF(\alpha, \Delta x, \Delta y)$ being the cost function. As in the 2D case, optimal values of mapping variables $\{\hat{\alpha}, \hat{\Delta x}, \hat{\Delta y}\}$ are determined by minimizing CF . Measured data are then mapped onto the modelling space via Eq. (12).

2 EXPERIMENTAL INVESTIGATION

The performance of the method is presented on two examples. For a practical purpose, both examples are presented in the following along with the corresponding results. Discussion of results is then provided in Section 3. Minimization of the cost function (Eq. (10)) was in both cases performed using the steepest descent method. Derivatives of the cost function were approximated using the finite difference method.

In the first example, the synthetic one, we demonstrated the method on a problem of aligning a specimen with a metamaterial-like structure. The purpose of the experiment is to present the advantage of the proposed alignment algorithm in comparison with the well-known ICP algorithm. The effect of misalignment on the performance of methods is also studied.

In the second example, the entire methodology is presented on real DIC data (i.e. shape recognition and alignment), where the ability of the method to handle the presence of noise and mismatches in the geometry is also tested. The example presents a heterogeneous strain field specimen proposed for the calibration of plastic anisotropy.

2.1 Synthetic Experiment

The geometry of the specimen is shown in Fig. 4. The shape consists of cut-out slots exhibiting an even pattern, mimicking the shape of metamaterial [21].

The specimen in Fig. 4 also represents the reference (modelling) geometry. The measured geometry is obtained by introducing rigid translation and rotation to the reference one [10].

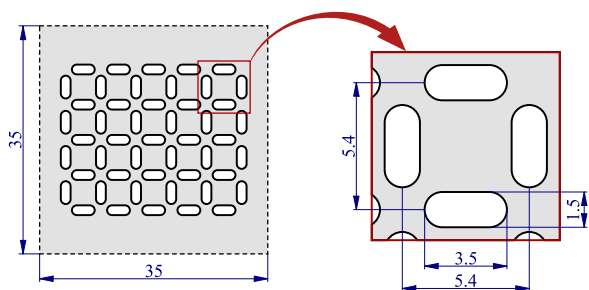


Fig. 4. Investigated synthetic specimen (dimensions in mm)

The effect of initial rotation and translation was examined to test the robustness of the method, mimicking multiple measurement setups. The translation was assumed equal in the horizontal x and vertical y direction (for the sake of brevity, to reduce the number of cases). Analysed cases include all possible combinations of rotations for angle α , ranging from 0° to 90° by a step 15° , and translations $\Delta x = \Delta y$, ranging from 0 mm to 2 mm with a step of 0.5 mm. An example of the initial misalignment ($\alpha = 2^\circ$, $\Delta x = \Delta y = 0.5$ mm), presents Fig. 5a. The corresponding error image, i.e. pixels from the error set E , Eq. (8), is presented in Fig. 5b in white.

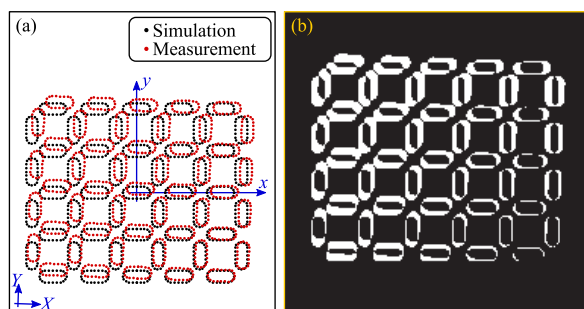


Fig. 5. Example of the analysed case:

- a) initial misalignment between simulation and measurement,
- b) the corresponding error image E , Eq. (8)

Besides the proposed method, the ICP was also employed in this example for comparison. Although many different versions of the ICP exist, one of the most basic closed-form versions of the algorithm was

assumed. Data used for the ICP are those presented in Fig. 5a. The closest point search was performed using the grid closest point and no false matches rejection was assumed.

Results. The performance of both methods is demonstrated by presenting the number of iterations that were needed to align each initial misalignment.

Results of both methods are presented in Fig. 6. Initial rotational misalignment α is displayed on the vertical axis and the translational Δx , Δy on the horizontal axis. Colours represent the number of iterations used for a successful alignment. Cases where the alignment was not achieved (exceeding 100 iterations or converged to an inappropriate position) are displayed in white.

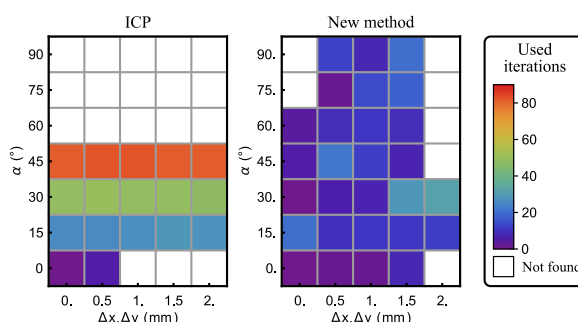


Fig. 6. Performance and convergence of ICP and the proposed alignment method

The proposed method managed to align the specimen in almost all the analysed cases of the initial misalignment. On the other hand, the ICP failed to converge or converged to a wrong minimum at initial misalignment rotations greater than 45° . With the increase of the initial rotation, the number of iterations considerably increased for the ICP, reaching over 80 iterations. The proposed method needed approx. 15 iterations on average for a successful alignment, irrespective of the initial rotation.

2.2 Application to Real DIC Data

Practical application of the method is demonstrated on a specimen presented in [22] (Fig. 7), designed to induce a biaxial strain-stress state. Due to the specimen's diverse geometry and no distinct features (such as sharp corners or round holes), the problem also represents a good alignment test problem.

Measurements were performed with a DIC measuring system Q-400 Dantec Dynamics GmbH, (Ulm, Germany). The measuring setup is presented in Fig. 8. We utilized 4 digital cameras, which in pairs acquire images of the front and

backside of the specimen, representing a case of a 3D DIC measurement. Such setup enables biplane measurement, where we simultaneously measure the front and back specimen surface in a common coordinate system. The calibration procedure was performed using a special two-sided calibration target. The purpose of such a set-up is enhanced characterization of material mechanical behaviour [22].

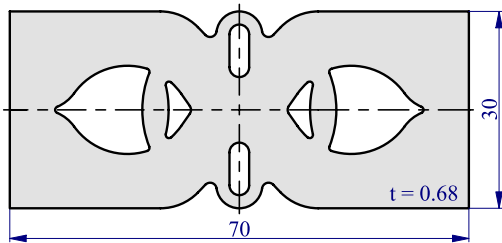


Fig. 7. Investigated specimen (dimensions in mm) [22]



Fig. 8. Measuring setup utilizing the digital image correlation

The DIC method was carried out with Istra 4D software. Technical details of the measuring system and the adopted DIC settings are summarized in Table 1.

The shape of the measured specimen was determined as presented in section Methods. Two images of the specimen were acquired by changing the background lightening as presented in Figs. 8 and 9. The first image was acquired using the typical lighting setup, presented on the left-hand side of Fig. 10a. For the second image, we illuminated the specimen from the backside, making the background on the acquired image brighter. The acquired image is presented on the right-hand side of Fig. 10a. The shape of the specimen is obtained from both images using Eq. (2).



Fig. 9. Image acquisition using a backlight

Table 1. Details of the measuring system

Company	DANTEC Dynamics GmbH, (Ulm, Germany)
Model	Q-400
Cameras	Manta G-507 (4 pieces)
Image resolution	2464 × 2056 pixel
Objective focal distance	35 mm
Field of view	approx. 65 mm × 40 mm
Patterning technique	mat white spray paint base coat with black speckles
Pattern feature size	approx. 3 pixel
DIC software	DANTEC Dynamics, Istra 4D (ver. 4.6)
Facet size	19 pixel
Grid spacing	12 pixel
Spatial smoothing	none
Temporal smoothing	none
Number of acquired data points	20400

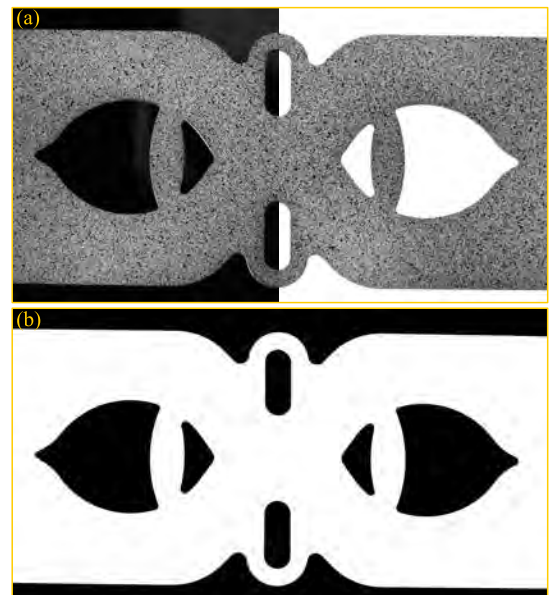


Fig. 10. Acquisition of specimen geometry: a) left-hand side: specimen image using typical lighting (Fig. 8) and right-hand side: using the backlight (Fig. 9), b) the obtained specimen geometry

Results. The determined shape of the measured specimen is presented in Fig. 10b. From the image, one could easily further determine the specimen contour, which most alignment methods use for the alignment procedure. In the proposed method, however, the entire shape is assumed, presented in Fig. 10b in white.

The measured and modelling specimen are together presented in Fig. 11a, given in the measured specimen's coordinate system. The measured shape is presented with the experimentally obtained image and the modelling shape with yellow dots. The corresponding error between both shapes is presented in the form of an error image, Fig. 11b. White pixels correspond to the error set E , obtained using Eq. (8), representing the regions of misalignment.

The initial value of the cost function CF , corresponding to the square number of white pixels in Fig. 11b, is approx. 10^{12} px^2 , Eq. (9). The value was then significantly reduced by the optimization procedure in only 5 iterations and the procedure successfully converged in 28 iterations. The final value of the cost function was reduced to approx.

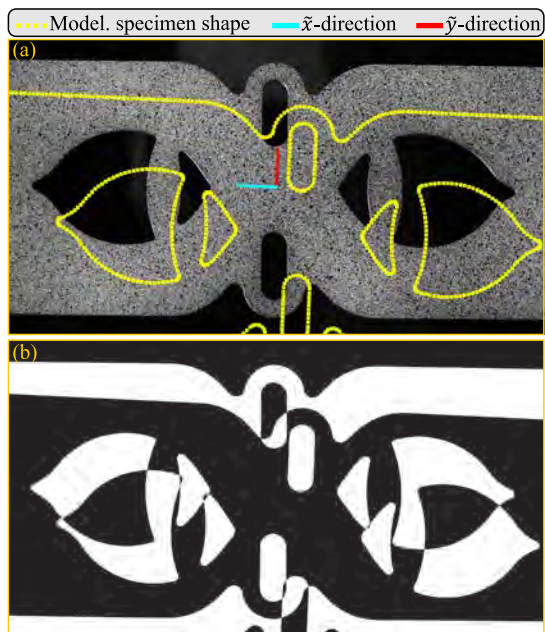


Fig. 11. The initial position of the measured and modelled specimen (the latter shown with yellow dots) in the measuring coord. system, a) presentation of both shapes, b) the corresponding error image of the method

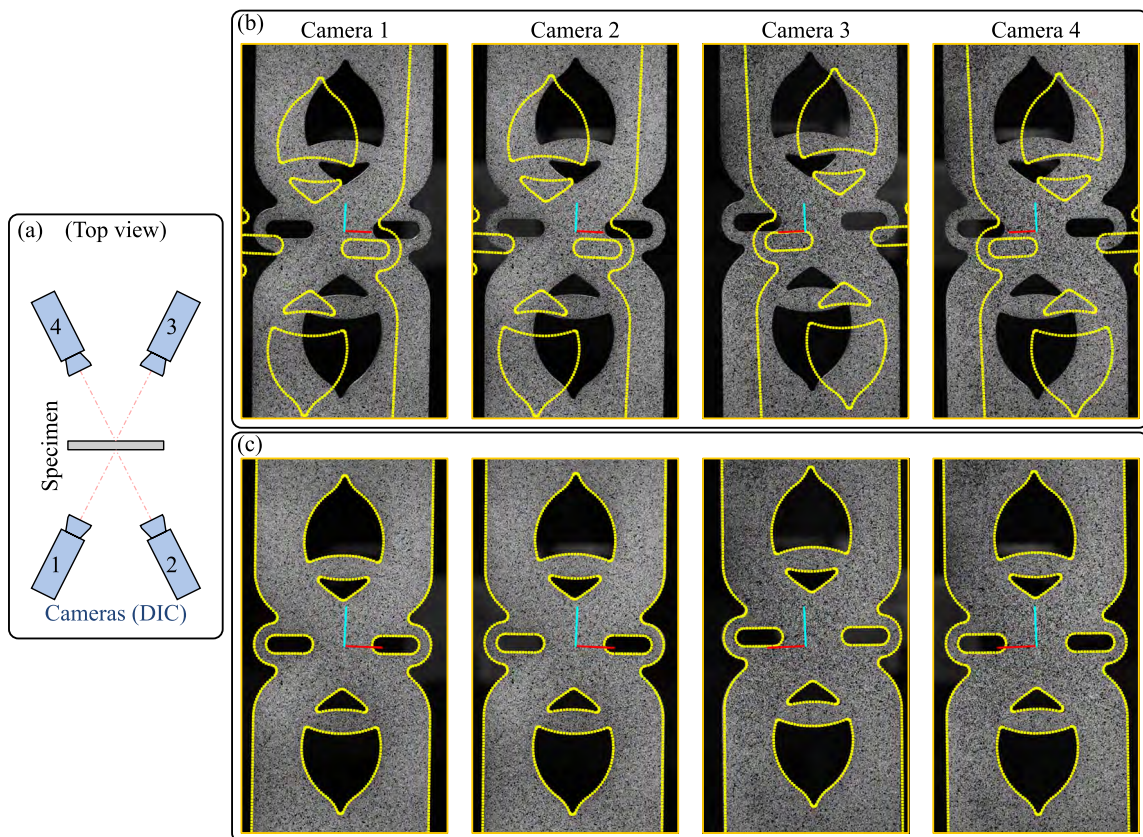


Fig. 12. Alignment of the specimen: a) camera arrangement, b) the initial misalignment and c) the optimal alignment

10^6px^2 with the optimal values of the mapping parameters determined to be: $\hat{\alpha} = 2.108^\circ$, $\hat{\Delta x} = 4.907 \text{ mm}$, $\hat{\Delta y} = 3.091 \text{ mm}$.

The final result of alignment is shown graphically in Fig. 12. The corresponding Fig. 12a schematically presents the experimental set-up. The initial misalignment of the measured and modelling specimen shape is presented in Fig. 12b, shown in the measuring coord. system. The final alignment of specimens is presented in Fig. 12c. Although it is difficult to quantify the level of the alignment, practically no difference between the aligned shapes can be observed.

Handling of the artifacts. To test the ability of the method in handling geometrical mismatches (due to machining or manufacturing tolerances) we purposely misplaced the holes in the measured specimen in the first case, see the upper row in Fig. 13, Case A, where the artifacts are marked with red on the ideal shape shown in white. In the second case, Case B, the roundings on the outer contour were purposely modified to some extent on the measured specimen as if they were not properly machined. The initial position of the modelled specimen is shown with magenta dots. As can be seen in the lower row of the figure, the alignment was performed successfully in both cases.

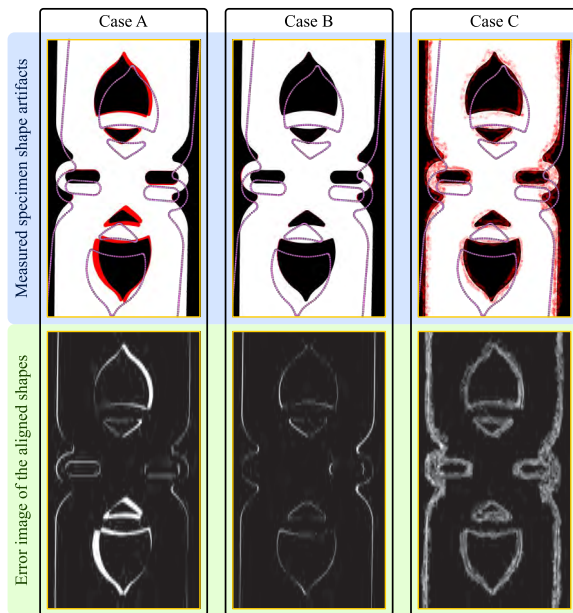


Fig. 13. Handling of the artifacts; upper row: the assumed artifacts, marked with red on the ideal specimen shape (white), initial modelled specimen shape position, marked with magenta, and lower row: the alignment results; Cases A and B analysed the mismatches in the measured geometry and Case C analysed the presence of noise (outliers)

The ability of the method to handle noise is analysed in Case C, where noise (outliers) are added along the entire contour of the measured specimen. As can be seen in the lower row of the figure, the method managed to successfully perform the alignment. We have also analysed the assumed cases with the ICP method. In Cases A and B, ICP performed the alignment as the proposed method whereas in Case C, the ICP method diverged.

3 DISCUSSION

The purpose of this study was to present a new methodology for aligning full-field measurements. More specifically, the method deals with the alignment of planar specimens, which is a field becoming increasingly popular due to its potential when combining full-field measurements with material characterization. As presented in [22], by using the full-field measurement techniques and monitoring the specimen loads, it is possible to characterize the anisotropic material behaviour from a single heterogeneous strain field specimen by employing inverse identification techniques (eq. finite element model updating, virtual fields method). However, as pointed out in the introduction, the outcome of the material characterization process crucially depends on the accuracy of the alignment between the measured and modelling data.

In the first example, the method was applied to a synthetic experiment. The shape of the specimen resembled a structure of a metamaterial. The example was chosen due to its periodic structure which can cause difficulties in the aligning procedure. We analyse the initial misalignment effect and the performance of the method compared with the ICP algorithm. The results demonstrated the robustness of the proposed method, which managed to successfully perform the alignment even at 90° of the initial rotation. This was not the case for ICP, which managed to perform the alignment up to 45° of the initial rotation only. We need to point out, however, that only the most basic version of the closed-form ICP was employed in this work. There are numerous modifications of the method [23] that could perform better. On the other hand, the performance of the proposed method could also be enhanced, such as by using pixels weights or by modifying the optimization algorithm. By employing such modifications, it is possible to successfully perform the alignment for the entire region of initial misalignments in Fig. 6 (data omitted). However, as is the case with the ICP, the convergence of the proposed

alignment algorithm cannot be guaranteed in a general case.

In the majority of cases on which we tested alignment methods, the performance of the ICP and the proposed method was comparable. In general, the advantage of the ICP is its computation efficiency. The method uses singular value decomposition (SVD) for the computation of alignment parameters whereas in the proposed approach an optimization approach is assumed. However, the benefit of the method is its robustness. As demonstrated with the first example, the initial mismatch did not considerably affect the number of iterations whereas they significantly increase in the case of the ICP. This property of the proposed approach comes from using the entire specimen region for the error estimation whereas in the ICP only the contour of a specimen is assumed.

The accuracy of the proposed algorithm on the one hand depends on the quality of the measuring image and on the other on the accuracy of the calibration. The mapping between the observed and modelled geometry is a composite of two mappings. First mapping c_1 of images, provided by the cameras, onto the unified plane takes place, subsequently followed by the mapping f of these points onto the space of the modelled specimen. It is important to note that the alignment algorithm addresses only the latter because the former is defined during a calibration procedure of cameras, needed for any full-field measurement system.

In the second example, the specimen recognition and the alignment process were both demonstrated. It can be seen from the acquired specimen's images (Fig. 11) that a simple threshold effect for the contour recognition could not be employed. Due to the speckle pattern, there are similar values of brightness in the background as well as on the specimen. For the presented approach of shape recognition, however, this was not an issue. The alignment of the specimen was in the example performed by using the proposed approach. However, ICP or any other method could be employed as well. From analysing the example (data omitted), similar performance was obtained for both the ICP and the proposed method, with ICP being computationally more efficient. Methods were also analysed for dealing with image artifacts, such as noise and geometrical mismatches due to machining or manufacturing tolerances. In the case of handling the noise, the proposed method turned out to be superior, otherwise, it is known that the ICP algorithm is susceptible to such artifacts [24].

In comparison to alignment procedures published in the field of material characterisation [17] and

[18], the benefit of the proposed approach is that it needs no user input for the shape recognition or for the alignment procedure and it is fairly easy to be implemented. Although we cannot claim that the presented approach for solving the shape recognition and the alignment problem is novel since in both areas there are numerous publications, we found no studies in the field of material characterisation that would address both problems and join them in a form of a methodology.

4 CONCLUSIONS

An integral part of advanced material characterisation by using full-field measurement techniques is the alignment between the FEA data and the experimental DIC data. Despite numerous publications in the field of treating planar specimens, studies that address the subject are scarce.

In the paper, a methodology is presented that enables the alignment of data from a single measuring system. In practice, the alignment approach needs to be addressed once the specimen is placed in the measuring system, more precisely, it needs to be addressed by the measuring system used for the DIC itself. However, because DIC does not recognize the pattern near an object's contour, the location of the contour is not exactly known. With the presented methodology we address both problems, specimen shape recognition, and the alignment procedure. The practical application of the method is presented on two examples, which demonstrate the robustness of the method, its comparison with the ICP algorithm, and its application on real DIC data.

Although the results show that the methodology manages to successfully perform the alignment, it is difficult to actually quantify its accuracy (from both the shape recognition and the alignment process). Such an analysis can be performed by means of computing full-field error maps [10]. For the present methodology, this remains to be performed.

5 ACKNOWLEDGEMENTS

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Determining Optimum Rotary Blade Design for Wind-Powered Water-Pumping Systems for Local Selected Sites

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The design of a windmill rotor is critical for harnessing wind energy. In this work, a study is conducted to optimize the design and performance of a rotor blade that is suitable for low wind conditions. The windmills' rotor blades are aerodynamically designed based on the SG6043 airfoil and wind speed data at local selected sites. The aerodynamic profile of the rotor blade that can provide a maximum power coefficient, which is the relation between real rotor performance and the available wind energy on a given reference area, was calculated. Different parameters, such as blade shapes, chord distributions, tip speed ratio, geometries set angles, etc., were used to optimize the blade design with the objective of extracting maximum wind power for a water pumping system. Windmill rotor of 10.74 m, 7.34 m, and 6.34 m diameter with three blades were obtained for the selected sites at Abomsa, Metehara, and Ziway in south-east Ethiopia. During the rotary blades performance optimization, blade element momentum (BEM) theory and solving iteration by MATLAB® coding were used.

Keywords: boundary element method, rotary sizing, wind power, hydraulic power, power coefficient, water pumping system

Highlights

- Optimum rotary blade for wind power at low wind speed has been designed based on SG603 airfoil and analysed for three selected local sites.
- The blade design is optimized with the objective of extracting maximum wind power for each selected site for a water-pumping system.
- The relation between real rotor performance and the available wind energy on the selected given reference areas is calculated to find the optimum aerodynamic profile of the blades.
- Blade element momentum theory has been used and iterated in MATLAB to identify the parameters for the performance optimization.

0 INTRODUCTION

Currently, significant global progress to benefit from the need for renewable energy by converting the energy of the wind to a mechanical form by designing and constructing diverse forms of windmills exists [1]. The main interest in wind energy is that the conversion process does not release carbon emissions to the environment, and it leads to less consumption of resources. In order to mitigate the side effects of fossil fuels, the use of renewable energy sources has become extremely important. In this respect, wind energy is considered one of the most promising renewable energy sources, which forces of nature can continuously generate, and the rotor blade design is decisive in optimally harvesting wind energy.

Studies on wind turbine rotor blade design have been continuously conducted based on the performance of aerodynamic analysis. Supreeth et al. [2] used a heuristic approach to study rotary blade design for a fixed pitch and small-scale horizontal axis wind turbine. The sectional chord and twist angle distribution for the idealized, optimized, and linearized blades were analytically determined. Prasad and Virupaxi [3] designed the rotary blade of windmills

for irrigational purposes in Bagur of Hosadurga district (Karnataka, India). Various parameters were optimized to achieve maximum power coefficient. Lopez-Lopez et al. [4] studied the effect of the blade slenderness and the wind speed on the dynamics and instability of wind turbine blade under large deflection. They used a simple cantilever beam model and employed a Galerkin approach validated with experimental results. Lee and Shin [5] also improved an optimization framework for blade aerodynamic design under realistic conditions.

The design parameters, such as angle of twist, chord distribution, maximum chord and the objective function relation, were obtained. Shubham et al. [6] designed and optimized the micro wind turbine blades at a rated wind speed of 8.4 m/s using QBlade, with which the SG6043 airfoil was selected, and simulation was done on a 1.2 m long blade. Chan [7] reported the design of a 1.2 kW wind turbine blade for rural application for Mandalay Hill of Mandalay City (Myanmar) at a rated wind velocity of 7 m/s. The blade airfoil, angle of pitches and distribution of sectional chord along the blade span were determined. Deepak et al. [8] designed and optimized the blade of a small wind turbine by considering various factors,

such as tip loss, hub loss, drag coefficient, and wake. By using blade element momentum (BEM) methods, power performance was simulated. Muhsen et al. [9] optimized the rotor blade parameters that affect blade design for small turbines with low performance due to low Reynolds number. In the optimization processes, the performance of the final designed blade performance was investigated by employing the blade element momentum theory. Norona and Krishna [10] designed a 300 W micro-horizontal axis wind turbine using BEM theory on SG6040 and SG6041 airfoils. The BEM procedure was codified in MATLAB software, and a 3 m diameter wind turbine was analysed with QBlade. In general, the literature study indicates that the efficient conversion of the wind energy directly to a mechanical form requires closer study of the local wind data and designing rotor blades that are tailormade for the wind condition.

This study focused on designing optimum rotary blades for a wind-powered water-pumping system for local selected sites at Abomsa, Metehara and Ziyay (Ethiopia). This design of the blades is based on the design of an aerodynamic type of SG6043 airfoil reported by Giguefe and Selig [11] and Osei et al. [12] and analysis of collected wind speed data of the sites [13] from the National Meteorological Agency (NMA). The blade parameters, including chord distribution length, angle of twist, tip speed ratio, power coefficient of the blade, and attack angle, were determined. These blades' parameters were optimized to attain maximum power coefficients by using BEM theory procedures, and different iterations were solved by MATLAB coding.

1 DESIGN METHODS AND PROCEDURES

The general procedures followed to design the rotor blades of the windmills for water pumping application are summarized in Fig. 1.

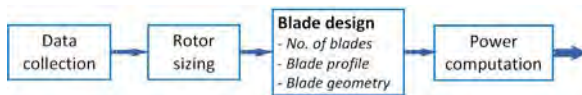


Fig. 1. Design procedure of rotary blade of wind turbines

The weather data were collected from selected local sites: Abomsa, Metehara and Ziyay, all from the eastern part of the Oromia regional state (Ethiopia). The next step is developing new designs and configurations of the rotor blades (rotor blade sizing) that can efficiently transform the energy in the wind to direct mechanical energy to drive water pumps. Fig. 2 describes the design steps that must be taken

to determine the optimum rotary size of windmills to be used at those particular sites and for a particular purpose.

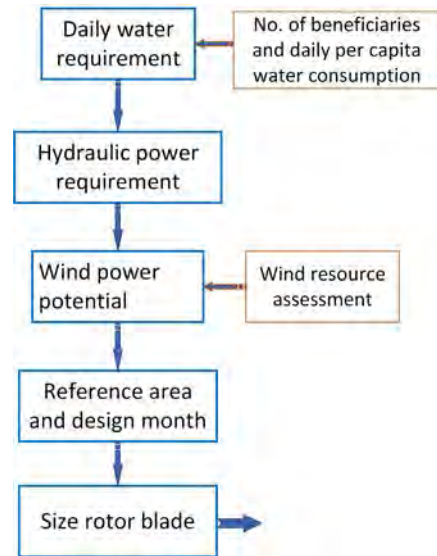


Fig. 2. Illustration of the design steps for sizing of windmill rotor for water-pumping system

Daily water requirement (Q_D) depends on the total number of beneficiaries (N_B) of the site and the daily water consumptions per capita (q_{pp}), expressed as Eq. (1).

$$Q_D = N_B \cdot q_{pp} \tag{1}$$

Hydraulic power requirement (P_{hyd}) parameter is a function of the daily water requirement (Q_D) and total pumping head (H) as given by Eq. (2) [14] to [16], where the total pumping head includes the static water head (H_{St}) and friction losses (H_{fr}), considering head losses in the pipe, velocity head losses, and other minor losses.

$$P_{hyd} = 0.113452 \cdot Q_D \cdot H, \tag{2}$$

where $H = H_{St} + H_{fr}$. The availability of wind power potential on a monthly basis is essential to evaluate as given in Eq. (3) [14] to [18].

$$P_{wind} = \frac{1}{2} \rho_{air} \cdot V^3. \tag{3}$$

Reference area (A_r) is related to the rotor area needed to capture sufficient power from the wind as given in Eq. (4) [14] to [18].

$$A_r = \frac{P_{hyd}}{P_{wind}}. \tag{4}$$

1.1 Blade Design Procedures

After the rotary blade size of horizontal axis windmills was determined, the following steps were considered in rotary blade design.

- Determining the minimum C_d/C_l ratio: This is important to determine the value of design lift and angle of attack corresponding to a minimum value of C_d/C_l ratio.
- Determining design lift coefficient and design angle of attack (α_d): From airfoil data corresponding to minimum C_d/C_l ratio, the value of lift coefficient and angle of attack are found.
- Choosing design tip speed ratio (λ) and the number of blades (B): These parameters can be chosen from tip speed ratio (TSR) vs the number of blade table depending on the form of operation.
- Determining geometries of optimum blades: When rotor blade size is determined, then it is divided into 10 to 20 equal segments or elements; for each unit element, the blade geometries such as local tip speed ratio $\lambda(r_i)$, relative wind angle $\psi(r_i)$, chord distribution $c(r_i)$, pitch angle $\theta(r_i)$, twist angle $\phi(r_i)$, and local solidity $\sigma(r_i)$ can be computed as shown in Fig. 3.
- Linearization of the twist angle and blade chord length: Using the optimum blade shapes as a guide, a blade shape that promises to be a good approximation is selected. Since the chords as well as the blade twist vary in a non-linear manner along the blade, it is possible to linearize the chords and the twist angles using Eq. (5).

$$\begin{aligned} c(r_i) &= a_1 r_i + a_2, \\ \phi(r_i) &= a_3 (R - r_i), \end{aligned} \tag{5}$$

where a_1 , a_2 and a_3 are constant coefficients obtained from the chosen chord distributions and twist angles.

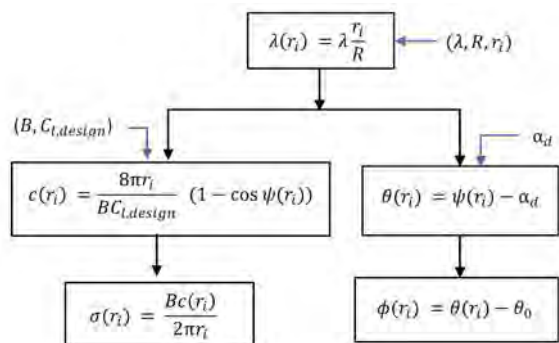


Fig. 3. Computation procedures to determine blade geometries

1.2 Rotor Blade Performance

After the chords and twist angles are linearized, the geometry of the rotary blade has a practical shape. The performance study of the designed rotor blade is conducted by using BEM theory [19]. Obtaining axial $a(r_i)_j$, and angular induction factors $\acute{a}(r_i)_j$ of the designed blade is the base of rotor blade performances. It is possible to determine the inflow angle using the axial and angular induction factors, as follows (Eq. (6)).

$$\psi(r_i) = \tan^{-1} \left(\frac{1 - a(r_i)_j}{(1 + \acute{a}(r_i)_j) \lambda(r_i)} \right). \tag{6}$$

To describe the lifting power reduction by airflow around the blade tip, the tip loss correction factor $F(r_i)$ for each i^{th} section is calculated based on Prandtl's method [20]:

$$F(r_i) = \left(\frac{2}{\pi} \right) \cos^{-1} \left[e^{-\left(\frac{\frac{B}{2} \left(\frac{1-r_i}{R} \right)}{\frac{r_i}{R} \sin \psi(r_i)} \right)} \right]. \tag{7}$$

The axial and tangential induction factors for each sectional i^{th} element at iteration j^{th} are then calculated using Eq. (8). If the axial induction factor is greater than 0.4, the Froude momentum principle is no longer accurate. Thus, Bangga [21] suggested correction of the axial induction factor, if $a(r_i)_j > a_c$, using Eq. (9).

$$\begin{aligned} a(r_i)_j &= \left(\frac{4F(r_i) \sin^2 \psi(r_i)}{\sigma_{r,i} (C_{l,i} \cos \psi(r_i) + C_{d,i} \sin \psi(r_i))} + 1 \right)^{-1}, \\ \acute{a}(r_i)_j &= \left(\frac{4F(r_i) \cos \psi(r_i) \sin \psi(r_i)}{\sigma(r_i) (C_{l,i} \sin \psi(r_i) - C_{d,i} \cos \psi(r_i))} - 1 \right)^{-1}, \end{aligned} \tag{8}$$

$$a(r_i)_j = \frac{1}{2} \left\{ 2 + Z(r_i) - \sqrt{(Z(r_i)(1 - 2a_c) + 2)^2 + 4(Z(r_i)a_c^2 - 1)} \right\}, \tag{9}$$

where a_c is commonly about 0.2 and $Z(r_i)$ is defined as:

$$Z(r_i) = \frac{4F(r_i) \sin^2 \psi(r_i)}{\sigma(r_i) (C_{l,i} \cos \psi(r_i) + C_{d,i} \sin \psi(r_i))}. \tag{10}$$

Finally, the local thrust coefficient $C_T(r_i)$ and power coefficient $C_P(r_i)$ can be computed from the following relations.

$$C_T(r_i) = \sigma(r_i) \left(1 - a(r_i)_j\right)^2 \frac{(C_{l,i} \cos \psi(r_i) + C_{d,i} \sin \psi(r_i))}{F(r_i) \sin^2 \psi(r_i)},$$

$$C_P(r_i) = \frac{8}{\lambda^2} F(r_i) \lambda^3(r_i) a(r_i)_j \left(1 - a(r_i)_j\right) \left(1 - \frac{C_d}{C_l}\right) \cot \psi(r_i). \quad (11)$$

2 ANALYSIS OF RESULTS

The rotary blades have been designed according to data collected by NMA of Ethiopia from three selected sites (Abomsa, Metehara, and Ziway, Ethiopia). The data was collected for one year (October 2018 to September 2019) at 10 m height. Table 1 shows important data determined from the three selected

sites, and total water demand and pumping required can be calculated by estimating the daily water consumption per capita to be 20 l/person in Ethiopia [14] and [15]. The monthly mean wind speeds were extrapolated at 20 m heights (hub heights), and monthly specific wind power potentials for the three selected sites were evaluated (Table 2).

Table 1. Important parameters for three selected sites

Parameters	Abomsa	Metehara	Ziway
No. of beneficiaries [-]	4086	4191	3867
Total demand [l]	81720	83820	77340
Pumping required [m ³ /day]	81.72	83.82	77.34
Total head [m]	87	79	74
Tip speed ratio, (λ) [-]	3	3.5	4
Number of blades, (B) [-]	3	3	3

Based on the number of beneficiaries and total water demands, hydraulic power can be determined,

Table 2. Available mean wind speeds and specific wind power for three selected sites

Year	Month	Abomsa			Metehara			Ziway		
		V_m at 10 [m/s]	V_m at 20 [m/s]	P_{wind} [W]	V_m at 10 [m/s]	V_m at 20 [m/s]	P_{wind} [W]	V_m at 10 [m/s]	V_m at 20 [m/s]	P_{wind} [W]
2018	Oct	2.82	3.42	24.59	2.93	3.56	27.58	3.82	4.64	61.12
	Nov	2.81	3.41	24.33	2.72	3.30	22.06	3.92	4.76	66.04
	Dec	2.99	3.63	29.31	2.63	3.19	19.95	3.56	4.32	49.47
2019	Jan	3.82	4.64	61.12	2.63	3.19	19.95	3.67	4.46	54.20
	Feb	3.63	4.41	52.44	2.92	3.55	27.30	3.05	3.70	31.11
	Mar	3.27	3.97	38.34	2.53	3.07	17.76	2.91	3.53	27.02
	Apr	2.33	2.83	13.87	2.65	3.22	20.40	2.66	3.23	20.64
	May	2.76	3.35	23.05	2.75	3.34	22.80	3.01	3.65	29.90
	Jun	2.58	3.13	18.83	3.39	4.12	42.71	4.14	5.03	77.80
	July	2.42	2.94	15.54	3.36	4.08	41.59	3.20	3.89	35.93
	Aug	2.01	2.44	8.90	3.23	3.92	36.95	3.64	4.42	52.88
	Sept	2.16	2.62	11.05	2.75	3.34	22.80	2.72	3.30	22.06

Table 3. Hydraulic power, reference area and rotor diameter for three selected sites

Year	Month	Abomsa			Metehara			Ziway		
		P_{hyd}	A_r	D_r	P_{hyd}	A_r	D_r	P_{hyd}	A_r	D_r
2018	Oct	806.3	32.8	6.5	750.9	27.2	5.9	649.1	10.6	3.7
	Nov	806.3	33.1	6.5	750.9	34.0	6.6	649.1	9.8	3.5
	Dec	806.3	27.5	5.9	750.9	37.7	6.9	649.1	13.1	4.1
2019	Jan	806.3	13.2	4.1	750.9	37.7	6.9	649.1	12.0	3.9
	Feb	806.3	15.4	4.4	750.9	27.5	5.9	649.1	20.9	5.2
	Mar	806.3	21.0	5.2	750.9	42.3	7.3	649.1	24.0	5.5
	Apr	806.3	58.1	8.6	750.9	36.8	6.9	649.1	31.5	6.3
	May	806.3	35.0	6.7	750.9	32.9	6.5	649.1	21.7	5.3
	Jun	806.3	42.8	7.4	750.9	17.6	4.7	649.1	8.3	3.3
	July	806.3	51.9	8.1	750.9	18.1	4.8	649.1	18.1	4.8
	Aug	806.3	90.6	10.7	750.9	20.3	5.1	649.1	12.3	4.0
	Sept	806.3	73.0	9.6	750.9	32.9	6.5	649.1	29.4	6.1

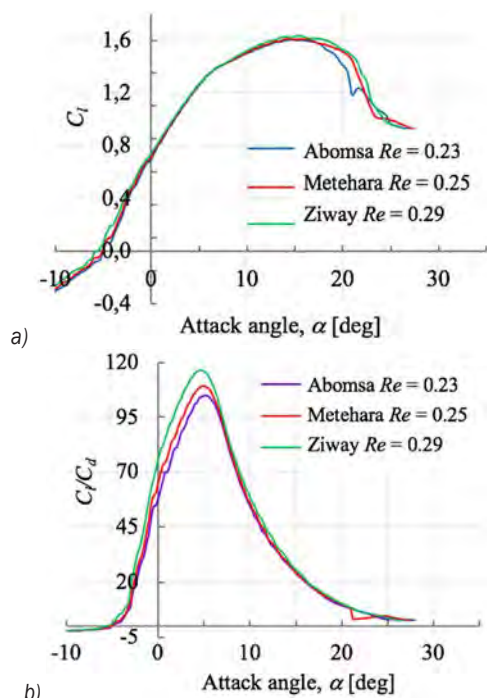


Fig. 4. Lift coefficient a) C_l and b) C_l/C_d of SG6043 with Reynolds numbers of three sites

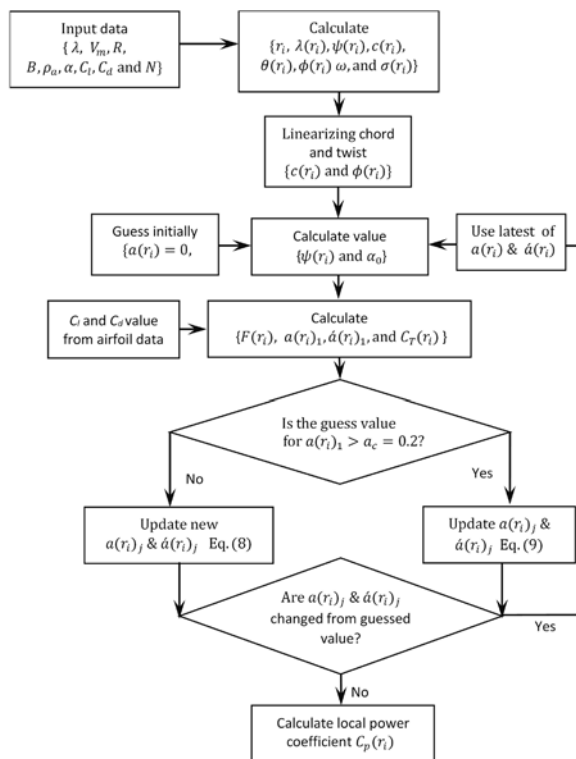


Fig. 5. Flow diagram of performing rotor blades design

which is constant through the year, the reference area and rotor diameter for the three sites were calculated and summarized in Table 3. The rotor size is chosen on the basis of “design month”, which is the month in which the water demand is highest in relation to the wind power resources, i.e., the month when the system will be most heavily loaded [14] and [16].

Therefore, according to the data given in Table 3, August, March and April are the design months for Abomsa, Metehara and Ziway, respectively.

For three rotor blade design, Giguéfe and Selig [11] and Osei [12] recommended airfoil type SG6043 for low Reynolds numbers because this type of airfoil operates in regional low wind speed. The designed C_l

Table 4. Geometric properties of optimum and linearized rotor blades for Abomsa site

r_i / R	Optimal blade				Linearized blade			
	$\psi(r_i)$	$\theta(r_i)$	$\phi(r_i)$	$c(r_i)$	$\psi(r_i)$	$\theta(r_i)$	$\phi(r_i)$	$c(r_i)$
0.07	52.46	47.46	40.17	0.39	24.83	19.83	20.79	0.77
0.13	45.47	40.47	33.18	0.59	23.34	18.34	19.30	0.73
0.20	39.36	34.36	27.07	0.68	21.86	16.86	17.82	0.70
0.27	34.23	29.23	21.94	0.69	20.37	15.37	16.33	0.67
0.33	30.00	25.00	17.71	0.66	18.89	13.89	14.85	0.64
0.40	26.54	21.54	14.25	0.63	17.40	12.40	13.36	0.61
0.47	23.69	18.69	11.40	0.59	15.92	10.92	11.88	0.58
0.53	21.34	16.34	9.05	0.54	14.43	9.43	10.39	0.54
0.60	19.37	14.37	7.08	0.51	12.95	7.95	8.91	0.51
0.67	17.71	12.71	5.42	0.47	11.46	6.46	7.42	0.48
0.73	16.30	11.30	4.01	0.44	9.98	4.98	5.94	0.45
0.80	15.08	10.08	2.79	0.41	8.49	3.49	4.45	0.42
0.87	14.03	9.03	1.74	0.38	7.01	2.01	2.97	0.39
0.93	13.10	8.10	0.81	0.36	5.52	0.52	1.48	0.35
1.00	12.29	7.29	0.00	0.34	4.04	-0.96	0.00	0.32

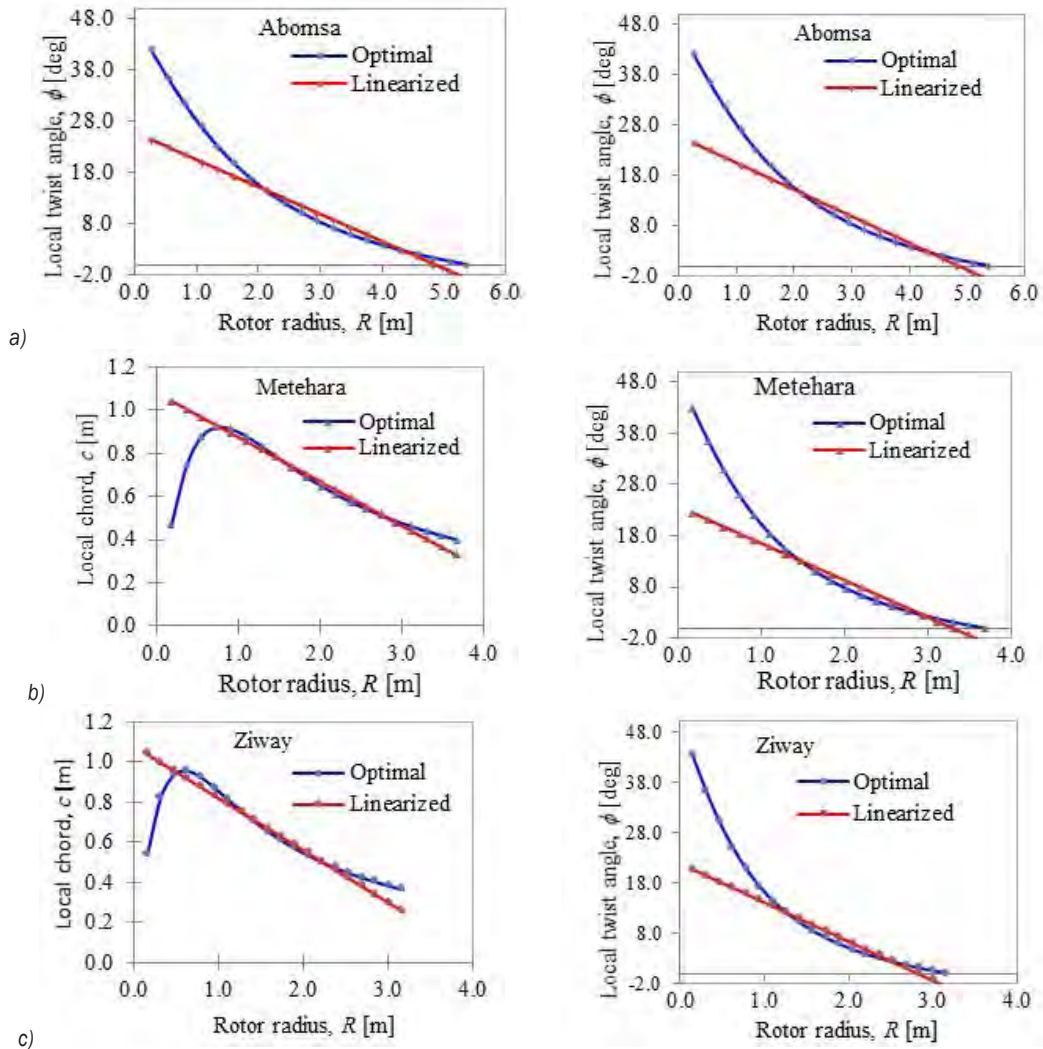


Fig. 6. Local chord and twist angle comparison of optimal and linearized blades for three sites, a) Abomsa, b) Metehara, and c) Ziway

Table 5. Geometric properties of optimum and linearized rotor blade for Metehara site

r_i / R	Optimal blade				Linearized blade			
	$\psi(r_i)$	$\theta(r_i)$	$\phi(r_i)$	$c(r_i)$	$\psi(r_i)$	$\theta(r_i)$	$\phi(r_i)$	$c(r_i)$
0.07	51.24	46.24	40.61	0.43	23.04	18.04	18.98	0.73
0.13	43.32	38.32	32.69	0.63	21.68	16.68	17.62	0.70
0.20	36.67	31.67	26.04	0.69	20.33	15.33	16.27	0.66
0.27	31.32	26.32	20.69	0.67	18.97	13.97	14.91	0.63
0.33	27.07	22.07	16.44	0.63	17.62	12.62	13.56	0.60
0.40	23.69	18.69	13.06	0.59	16.26	11.26	12.20	0.57
0.47	20.98	15.98	10.35	0.54	14.91	9.91	10.85	0.54
0.53	18.79	13.79	8.16	0.49	13.55	8.55	9.49	0.50
0.60	16.98	11.98	6.35	0.45	12.19	7.19	8.13	0.47
0.67	15.47	10.47	4.84	0.42	10.84	5.84	6.78	0.44
0.73	14.19	9.19	3.56	0.39	9.48	4.48	5.42	0.41
0.80	13.10	8.10	2.47	0.36	8.13	3.13	4.07	0.38
0.87	12.16	7.16	1.53	0.34	6.77	1.77	2.71	0.35
0.93	11.35	6.35	0.72	0.32	5.42	0.42	1.36	0.31
1.00	10.63	5.63	0.00	0.30	4.06	-0.94	0.00	0.28

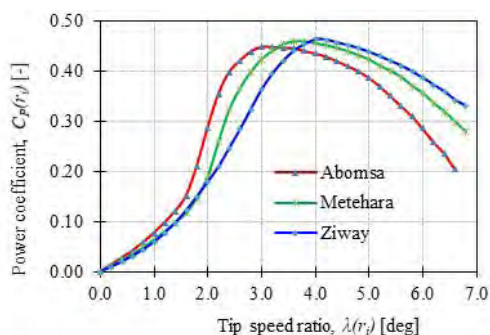


Fig. 7. The vs. of rotor performed blades for three selected sites

and C_l/C_d graphs characteristics of SG6043 with Re of 0.23×10^6 , 0.25×10^6 and 0.29×10^6 are evaluated, as illustrated in Fig. 4, using QBlade, for Abomsa, Metehara, and Ziway, respectively. Fig. 5 shows the general flow diagram for determining blade geometries and performing the designed blades by iterating axial and tangential induction factors.

The blade geometries are determined with the number of segments or elements (N) equal to 15.

The local chord distributions, geometries set angles, and solidities of optimal and linearized blades for the three selected sites are determined and summarized in Tables 4, 5 and 6. The comparison of blade chord distributions and sectional angles of twist for optimal and linearized blades for the three selected sites are illustrated in Fig. 6.

Based on the BEM flow diagram of iteration in Fig. 5, each local axial and tangential induction factors have converged after many iterations by MATALB coding and the coefficients of performance are illustrated as shown in Fig. 7 for these three local sites. The geometry profiles (shape) of designed and optimized rotor blades are modelled by QBlade software, and the resulting blades are shown in Fig. 8. As can be observed from this figure, the maximum coefficient of performance values of $C_p(r_i)$ are 0.448, 0.459 and 0.463 at the maximum tip speed ratio of $\lambda(r_i)$ 3.0, 3.5 and 4.0 for the three sites Abomsa, Metehara and Ziway respectively. The geometry shape of optimal, linearized and optimized designed blades modelled by QBlade software and the resulting blades are shown in Fig. 8.

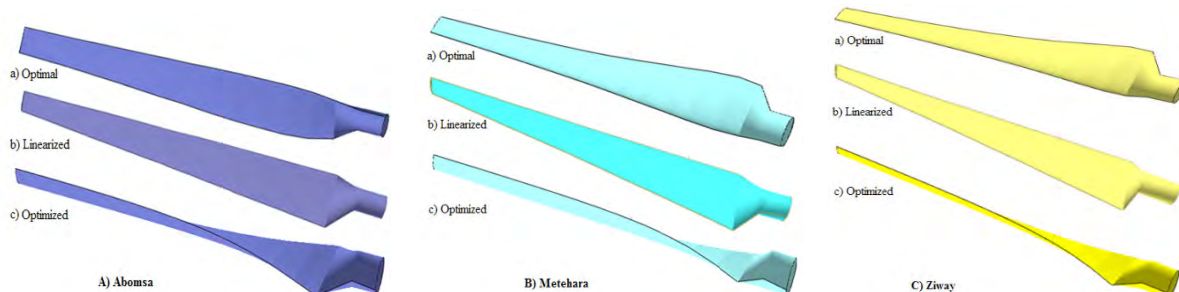


Fig. 8. Geometry shape of optimal, linearized and optimized blades modelled by QBlade for A) Abomsa, B) Metehara and C) Ziway sites.

Table 6. Geometric properties of Optimum and linearized rotor blades for Ziway site

r_i / R	Optimal blade				Linearized blade			
	$\psi(r_i)$	$\theta(r_i)$	$\phi(r_i)$	$c(r_i)$	$\psi(r_i)$	$\theta(r_i)$	$\phi(r_i)$	$c(r_i)$
0.07	50.05	45.05	40.69	0.47	21.41	16.41	17.29	0.67
0.13	41.29	36.29	31.93	0.66	20.17	15.17	16.05	0.64
0.20	34.23	29.23	24.87	0.69	18.94	13.94	14.82	0.61
0.27	28.77	23.77	19.41	0.65	17.70	12.70	13.58	0.58
0.33	24.58	19.58	15.22	0.60	16.47	11.47	12.35	0.55
0.40	21.34	16.34	11.98	0.54	15.23	10.23	11.11	0.52
0.47	18.79	13.79	9.43	0.49	14.00	9.00	9.88	0.49
0.53	16.74	11.74	7.38	0.45	12.76	7.76	8.64	0.46
0.60	15.08	10.08	5.72	0.41	11.53	6.53	7.41	0.43
0.67	13.70	8.70	4.34	0.38	10.30	5.30	6.18	0.39
0.73	12.55	7.55	3.19	0.35	9.06	4.06	4.94	0.36
0.80	11.57	6.57	2.21	0.32	7.83	2.83	3.71	0.33
0.87	10.73	5.73	1.37	0.30	6.59	1.59	2.47	0.30
0.93	10.00	5.00	0.64	0.28	5.36	0.36	1.24	0.27
1.00	9.36	4.36	0.00	0.26	4.12	-0.88	0.00	0.24

3 CONCLUSION

In this study, the optimum rotary blade for a wind-powered water pumping system for local three selected sites has been designed and analysed. From these designs, the optimal blade parameters are determined, linearized, and optimized. The geometry profiles of optimized blades are modelled using QBlade software for the three local selected sites. The key results of the design and analysis work show that the maximum coefficient of performance values $C_p(r_i)$ are 0.4512, 0.4587, and 0.4627 at the maximum tip speed ratio $\lambda(r_i)$ values of 3.0, 3.5 and 4.0 for the three sites Abomsa, Metehara, and Ziway, respectively. Furthermore, the average wind speeds at a height of 20 m are 3.50 m/s, 3.60 m/s and 4.60 m/s, respectively.

This article is part of a PhD research whose further work will focus on developing the design model of the blade with the obtained design specification and building a prototype wind turbine to test at the three locations.

4 NOMENCLATURE

$a(r_i)$	Axial induction factor, [-]
$\acute{a}(r_i)$	Angular induction factor, [-]
a_1, a_2, a_3	Constant coefficient, [-]
A_r	Reference area, [m ²]
$c(r_i)$	Chord distribution, [-]
$C_{d,i}$	Local drag coefficient, [-]
C_d	Drag coefficient, [-]
$C_{l,i}$	Local lift coefficient, [-]
C_l	Lift coefficient, [-]
$C_p(r_i)$	Local power coefficient, [-]
$C_T(r_i)$	Local thrust coefficient, [-]
$F(r_i)$	Blade tip loss factor, [-]
H	Total pumping head, [m]
H_{fr}	Friction losses, [m]
H_{st}	Static water head, [m]
N_B	Total number of beneficiaries [-]
P_{hyd}	Hydraulic power requirement [W]
P_{wind}	Wind power potential [W/m ²]
Q_D	Daily water requirement, [m ³ /day]
q_{pp}	Water consumption per capita, [l/day]
R	Rotor blade radius, [m]
r_i	Local radius, [m]
V_m	Wind velocity, [m/s]
α_d	Design attack angle, [deg]
$\theta(r_i)$	Pitch angle, [deg]
$\lambda(r_i)$	Local tip speed ratio, [-]
λ	Tip speed ratio, [-]
ρ_{air}	Air density, [kg/m ³]
$\sigma(r_i)$	Local solidity, [m ⁻¹]

$\phi(r_i)$	Twist angle, [deg]
$\psi(r_i)$	Relative wind angle, [deg]

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Investigation on the Modeling and Dynamic Characteristics of a Novel Hydraulic Proportional Valve Driven by a Voice Coil Motor

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As the key control component of the water hydraulic systems, the water hydraulic proportional valve has a significant influence on the control performance of the systems. Due to the poor viscosity and lubricity of water, the valve spool resistance is large and non-linear. In this study, a novel fast-response water hydraulic proportional valve is presented. The actuator of the valve adopts a voice coil motor (VCM), which has the advantages of fast response, high control precision and small volume. In order to realize the fast control of the valve, a lever amplifier is designed to obtain enough actuation force. A detailed and precise non-linear mathematical model of the valve considering both the valve's structural parameters and VCM electromagnetic characteristics is developed. A comprehensive performance simulation analysis has been carried out, mainly divided into an electromagnetic simulation, an analysis of the characteristics of the lever magnifier, and a dynamic performance simulation of the valve. The simulation results show that the adjusting time is about 28 ms, and the maximum overshoot is about 5 %. The step response rise time is about 15 ms. The test rig of the valve and VCM have been built. The test results of the prototype show that the optimal stroke range of VCM is 4 mm to 15 mm. The maximum overshoot of the valve is around 10 %; the adjusting time is about 30 ms in the opening process and 35 ms in the closing process. The test results prove that the valve has good static and dynamic control performance.

Keywords: proportional control valve, voice coil motor, optimization design, dynamic performance

Highlights

- A novel water hydraulic proportional valve with fast response is proposed.
- The actuator of the valve adopts a voice coil motor (VCM), and a lever amplifier is designed to obtain enough actuation force to realize the fast and accurate control.
- The detailed and precise mathematical models of the valve have been developed, and the comprehensive optimization design has been carried out.
- The test rig has been built, and the results indicate that the step response time of 0 % to 100 % stroke is about 30 ms.

0 INTRODUCTION

Compared with oil hydraulic systems, water hydraulic systems could be a good solution for the environmental and safety problems [1]. Water hydraulic systems are widely used in steel and glass production, ocean exploration, food and medicine processing, and coal mining [2] to [4]. The water hydraulic control valve is one of the most important components of hydraulic systems, and its dynamic performance has an important impact on the performance of hydraulic systems [5]. However, it is quite difficult to design a high-performance water hydraulic proportional valve due to the low viscosity and oxidative corrosion of water. Many problems, such as poor lubrication and large friction, sealing, and leakage, need to be solved in the design process. Some studies [6] have shown that while the oil directional control sliding valve worked perfectly, there is a strengthening oscillatory movement of the same water valve spool, caused by non-optimal valve material combination (steel/steel) in water-lubricated conditions.

The current research indicates that the dynamic performance of electro-mechanical devices is one of the key factors affecting the response speed of the control valve [7]. Considering that the high-speed on/off solenoids have fast response, some water hydraulic control valves are designed with them as the actuators [8] and [9]. However, they are usually used in high-speed switching valves with small flow capacity due to the small actuation force and stroke. Park et al. [10] and Park [11] proposed a water pressure proportional valve with the high-speed switching valve as a pilot stage, and the experiments show that the maximum flow rate can reach 9 l/min and the pressure drop is 7 MPa. This kind of valve can realize proportional control by pulse width modulation (PWM) signal but cannot achieve a good control effect. Piezoelectric actuators have become a solution for improving the valve dynamic performance because of their high response speed and high output force. However, their stroke usually is very small even at a large applied voltage, and the properties are greatly affected by temperature [12] and [13]. These drawbacks restrict

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the valve opening (which in turn limits maximum flow rate), and it is difficult to adapt to the working environment. Compared with the high-speed on/off solenoids and piezoelectric actuators, a voice coil motor (VCM) has the obvious advantages of fast response and large stroke-actuation force. Therefore, as a high-performance actuator, a voice coil motor has attracted the attention of researchers. Dahoon et al. [14] designed a high force voice coil motor. VCM usually can provide 0.7 N to 1000 N actuation force and 0 mm to 100 mm stroke. The motion frequency of VCM can reach 1000 Hz. It is widely used in medical, aviation, vibration platform control, etc. A comparison of these existing electro-mechanical devices shows that VCM is an appropriate choice for electro-mechanical actuators in water hydraulic control valve.

As an ideal electric-mechanical conversion device, VCM can be used to quickly and accurately control the movement of the valve spool. Li et al. [15] and [16] proposed a pneumatic servo-valve with control pressure up to 20 MPa, which was driven by VCM. Experiments show that the valve has good control performance, and the step response frequency can reach 100 Hz. Zhang et al. [17] proposed a direct-drive water hydraulic control valve driven by a voice coil motor, and the rated flow is 14 l/min. The test results show that the control valve has the performance of approximate linear proportional control. In contrast, other researchers are committed to improving the dynamic performance of the valve by optimizing structural parameters. Liu et al. [18] studied a novel throttle poppet valve based on the hydraulic feedback principle. Filo et al. [19] proposed a novel control valve that consists of a throttle valve and a differential valve controlled by the pressure difference between the supply line and the return line to increase the speed of cylinder piston rod movement. Xu et al. [20] studied the mathematical modelling and simulation of a novel hydraulic variable valve time system. Simic and Herakovic [21] pointed out that non-optimized

hydraulic valve geometry is the main cause for many problems related to response time, actuation force and energy consumption. The results of the simulation analyses show that the axial component of the flow forces could be reduced significantly by optimizing the geometry of the valve spool and housing.

Due to the poor viscosity and lubricity of the water medium, the conventional hydraulic valve structure and the optimization design method will be not suitable for the water hydraulic proportional valve. In this study, a novel fast-response water hydraulic proportional valve that driven by VCM is presented. A comprehensive performance simulation analysis has been carried out, which mainly includes electromagnetic simulation of VCM, characteristics analysis of lever magnifier and dynamic performance simulation of the valve.

1 STRUCTURE AND PRINCIPLE OF THE VALVE

The water hydraulic valve proposed in this study uses a lever amplifier to amplify the VCM actuation force (as shown in Fig. 1). A smaller VCM can be used to drive the spool to move quickly. The traditional oil hydraulic control valve mostly adopts the slide valve structure. Due to its small sealing clearance, the working fluid cleanliness level has a significant impact on the valve performance [22]. The existing research shows that water hydraulic control valve with non-gap sealing structures, such as the seat valve (ball valve, disc valve, poppet valve etc.), can effectively solve the leakage problem caused by the low viscosity of water and improve the anti-pollution ability of the valve. The valve orifice adopts a seat valve structure, which is mainly composed of seat and ball. The displacement sensor detects the spool position, and the displacement signal will be fed back to the PID controller for closed-loop control.

The water hydraulic control valve with non-gap sealing structures such as seat valve (ball valve, disc

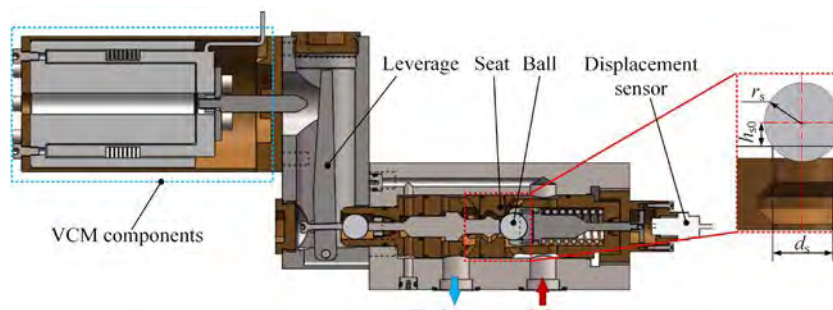


Fig. 1. Structure of the water hydraulic proportional valve

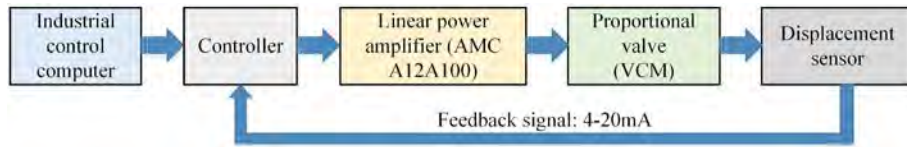


Fig. 2. The valve control schematic

valve, poppet valve etc.) can effectively solve the leakage problem caused by the low viscosity of water and high working pressure [7] and [11]. The valve orifice adopts a seat valve structure, which is mainly composed of valve seat and ball. The displacement sensor detects the spool position, and the displacement signal will be fed back to the controller for closed-loop control. The main parameters of the prototype are shown in Table 1.

Table 1. The main parameters of the valve

Parameter	Symbol	Value	unit
Fluid density	ρ	1000	kg/m ³
Flow coefficient	C_d	0.82	/
Valve seat diameter	d_s	10	mm
Valve ball diameter	r_s	12.7	mm
Return spring stiffness	k_{sv}	18	N/mm
Valve opening	x_v	0 to 1.2	mm

The closed-loop control has been used to improve the control precision and anti-interference ability of the valve. The control schematic is shown in Fig. 2. The control system is composed of a PC, controller, linear power amplifier (linear motor driver, AMC A12A100), displacement sensor and the water hydraulic proportional valve, etc. The PC sends out the control command, and the controller compares the control command with the feedback displacement signal. Through the comparison operations, $\pm 10V$ control signal will be output to the VCM linear driver (AMC A12A100). The linear driver linearly amplifies the control signal into 0 A to 10 A current to drive VCM so as to drive the spool to move until reaching the specified position. At that point, the valve spool is finally kept at the specified position.

2 THE VALVE MODEL

As the intermediate structure of transfer force and displacement, the lever amplifier significantly influences the performance of the valve. Based on the VCM model and the valve model, the influence of lever amplifier parameters on the dynamic performance is studied in detail.

2.1 VCM model

Fig. 3 shows the principle of VCM, and Table 2 shows the main VCM parameters. It is composed of the support, coil, magnet yoke, soft iron and permanent magnet. The real permanent magnetic material and soft iron material are N40H and DT3, respectively. The permanent magnet provides a uniform and constant magnetic field inside the motor. According to Lorentz force law, the Lorentz force will be produced when the coil is electrified in the magnetic field. The Lorentz force is the actuation force for the VCM and drives the coil to move.

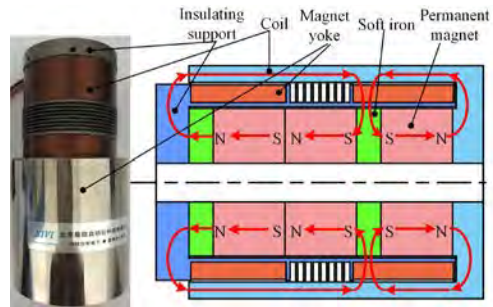


Fig. 3. VCM composition and arrangement of internal magnetic poles

The arrangement of the inner magnetic pole is shown in Fig. 3. A total of three permanent magnets are designed and installed in VCM. Among them, the two permanent magnet poles at the bottom are assembled in reverse to form a strong magnetic gathering effect at the working air gap. This will also help to reduce magnetic leakage and improve magnetic induction strength. Then the electromagnetic efficiency will be greatly improved.

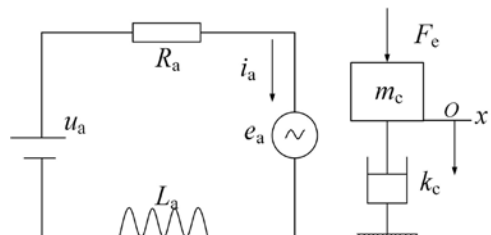


Fig. 4. The VCM equivalent electrical circuit and equivalent dynamic model

Table 2. The main parameters of VCM

Parameter	Value	unit
Maximum voltage	24.5	V
Maximum current	5.8	A
Continuous actuation force	51.5	N
Peak actuation force	160	N
Stroke	0 to 18	mm
Force constant	27.5	/

When the coil is electrified, Lorentz force will be produced and drive the coil to move. The dynamic model of VCM can be simplified as a mass damping mechanical motion system. Therefore, VCM can be equivalent to the coil equivalent circuit model and mass damping motion model, as shown in Fig. 4. According to the force balance equation and voltage balance equation of VCM, the mathematical model can be obtained as Eq. (1).

$$\begin{cases} F_c = B_\sigma l_c N i_a \\ u_a = e_a + i_a R_a + L_a \frac{di_a}{dt}, \\ e_a = B_\sigma l_c \dot{x}_c \\ F_c = m_c \ddot{x}_c + k_c \dot{x}_c \end{cases} \quad (1)$$

where F_c is the electromagnetic force of VCM [N]; k_c is the damping coefficient [N/(m/s)]; i_a is the current in the coil [A]; u_a is the voltage of the coil armature [V]; e_a is the back electromotive force generated by the coil motion [V]; L_a is the coil inductance [H]; m_c is the mass of the coil [kg].

2.2 Valve Dynamic Model

Compared with the poppet valve, the ball valve can realize a self-centring function. Therefore, it has better sealing performance than the poppet valve. In order to obtain good sealing performance, the valve adopts the

structure of the ball valve. The flow rate of the valve can be calculated with the following equations:

$$\begin{cases} q_v = C_d A(x_v) \sqrt{\frac{2\Delta p_s}{\rho}} \\ A(x_v) = \pi d_s h_{s0} \frac{(1 + \frac{x_v}{2h_{s0}}) x_v}{\sqrt{\left(\frac{d_s}{2}\right)^2 + (h_{s0} + x_v)^2}} \end{cases}, \quad (2)$$

where C_d is flow coefficient of valve orifice; Δp_s is the pressure difference of the valve [MPa]; ρ is the water density [kg/m³]; d_s is the diameter of the valve seat [m]; x_v is the valve opening [m]; h_{s0} can be calculated with the following formula:

$$h_{s0} = \sqrt{r_s^2 - \left(\frac{d_s}{2}\right)^2}. \quad (3)$$

When the spool is moving, the valve will be subjected to the comprehensive effect of flow force, spring force, friction force, inertia force, and viscous resistance (as shown in Fig. 5). VCM needs to provide enough driving force to overcome these resistances so that the valve spool has enough acceleration to meet the requirements of rapid response. Assuming that the opening direction of the valve is positive, the kinetic equation can be established as follows:

$$\begin{cases} m_s \ddot{x}_v = F_s + \frac{\pi p_s}{4} (d_2^2 - d_5^2) + \\ \frac{\pi p_k}{4} (d_5^2 - d_2^2) - F_{f1} - F_{fv1} - F_{ks1}, \\ F_{ks1} = k_{sv} (l_{v0} + x_v) \\ F_{f1} = \sum 1.57 \mu_i d_i \sqrt{4r_o^2 - 0.25\Delta h^2} \Delta p_o \\ F_{fv1} = \rho q_v v \cos \alpha_o \end{cases}, \quad (4)$$

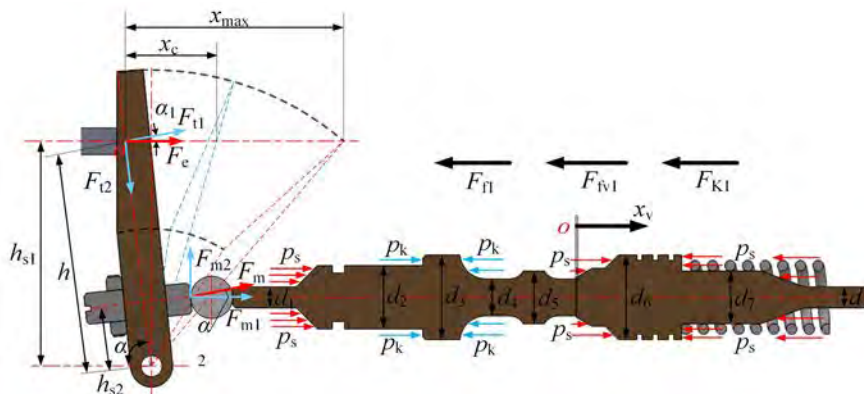


Fig. 5. The force analysis of the valve

where m_s is equivalent mass of the valve moving parts [kg]; p_s is the inlet pressure [MPa]; d_i is the diameter of the valve stem ($i = 1, 2, 3, 4, 5$) [m]; p_k is the pressure acting on the stem (MPa); F_{fl} is the friction force [N]; F_{ks1} is then spring force [N]; F_{fv1} is flow force (N); k_{sv} is spring stiffness [N/m]; l_{v0} is initial compression of spring [m]; μ_f is the friction coefficient of o-ring; r_o is the o-ring radius [m]; Δh is O-ring compression [m]; Δp_o is difference pressure acting on o-ring [MPa]; q_v is the flow rate [l/min]; v is the valve orifice flow velocity [m/s]; α_0 is the valve orifice jet angle [°]. Where F_s is the driving force acting on the ball, and it can be got by:

$$F_s = \begin{cases} F_e \cos(\arctan(\cot \alpha - \frac{x_c}{h_{s1}})) \frac{h_{s1}}{h_{s2}} & x_c \leq h_{s1} \cot \alpha \\ F_e \cos(\arctan(\frac{x_c}{h_{s1}} - \cot \alpha)) \frac{h_{s1}}{h_{s2}} & x_c > h_{s1} \cot \alpha \end{cases}, \quad (5)$$

In the same way, the relationship between the stroke of VCM and the stroke of the valve spool can be obtained:

$$x_v = \begin{cases} \frac{x_c h_{s2}}{(\cos(\arctan(\cot \alpha - \frac{x}{h_{s1}})) h_{s1})} & x_c \leq h_{s1} \cot \alpha \\ \frac{x_c h_{s2}}{(\cos(\arctan(\frac{x}{h_{s1}} - \cot \alpha)) h_{s1})} & x_c > h_{s1} \cot \alpha \end{cases}, \quad (6)$$

where F_e is Lorentz force of VCM [N]; α is the swing angle of leverage [°]; h_{s1} is the centre distance of VCM [m]; h_{s2} is the centre distance of ball [m]; x_c is the coil displacement [m].

3 SIMULATION ANALYSIS

Based on the mathematical model of the valve, the simulation model can be established in the MATLAB/Simulink and Maxwell 3D, respectively. The influence of the lever parameters on the performance of the output actuation force and displacement is compared and analysed, following which the best amplification coefficient is obtained.

3.1 Electromagnetic Simulation

The simulation model of VCM is established in Maxwell 3D, as shown in Fig. 6. Since the VCM is of symmetrical structure, half of the model can be used for simulation calculation to improve the calculation

efficiency. The main electromagnetic simulation parameters of VCM is shown in Table 3.

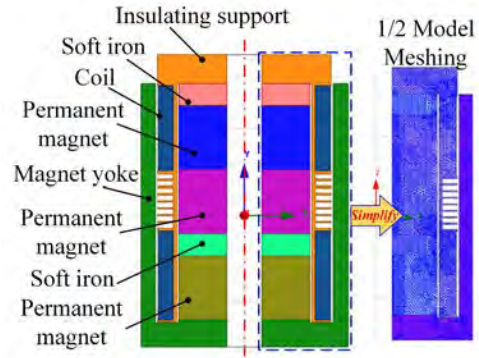


Fig. 6. VCM FEM model

Table 3. Electromagnetic simulation parameters

Name	Soft iron	Permanent magnet	Coil
Parameter	Material: 10#	Material: N40H; Coercivity: 970 kA/m; Remanence: 1.3 mT	Copper wire diameter: 0.5 mm; Turns - number: 550; Inside diameter of coil: 41 mm; Outside diameter of coil: 50 mm; Coil length: 50.6 mm

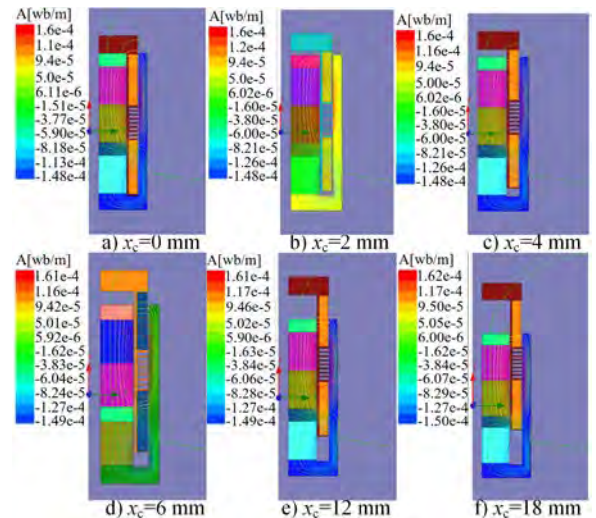


Fig. 7. Electromagnetic simulation of VCM in the whole stroke,

- a) $x_c = 0$ mm, b) $x_c = 2$ mm, c) $x_c = 4$ mm,
- d) $x_c = 6$ mm, e) $x_c = 12$ mm, f) $x_c = 18$ mm

The magnetic field distribution can be obtained by the electromagnetic simulation, as shown in Fig. 7. The simulation results indicate that there will be a small amount of magnetic leakage at the top of the coil and the outside of the magnetic steel, respectively. The effect of magnetic concentration is obvious when the two permanent magnets are assembled in reverse,

and the majority of the Lorentz force is generated here. In the whole stroke of VCM, the maximum vector magnetic potential increases slightly from 1.6×10^{-4} WB/m to 1.6171×10^{-4} WB/m; this ensures the stability of the actuation force output of the voice coil motor in the stroke.

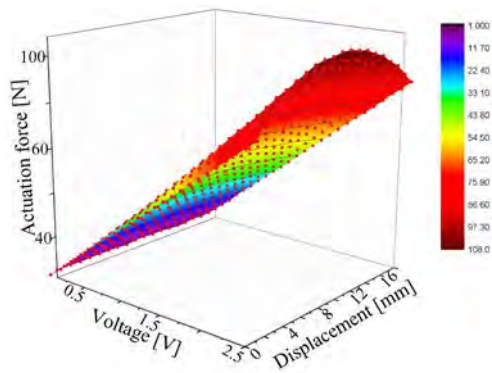


Fig. 8. Simulation results of VCM actuation force

Fig. 8 shows the simulation results of the VCM actuation force. The Lorentz force F_c with the variation of parameters (U_c, x_c) has been shown in Fig. 8, of which the coil current i_a is determined by coil control voltage U_c and x_c is the coil displacement. With the movement of the coil, the maximum actuation force is generated in the middle section, and the minimum actuation force is generated at both sides of the stroke. The distribution of electromagnetic actuation force is small on both sides and large in the middle section. When the continuous actuation force is 51.5 N, the current is about 1.673 A to 1.804 A.

3.2 Characteristics Analysis of Lever Magnifier

When VCM pushes the lever to drive the spool, the vertical distance between the VCM push rod and the lever fulcrum will change with the rotation of the lever. When the lever is perpendicular to

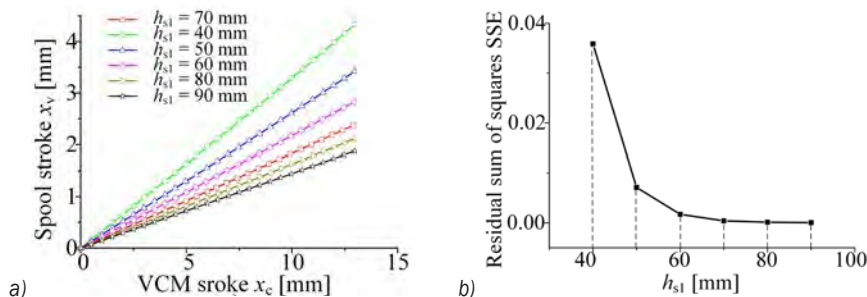


Fig. 9. Relationship between a) VCM stroke and spool stroke; and b) residual sum of squares

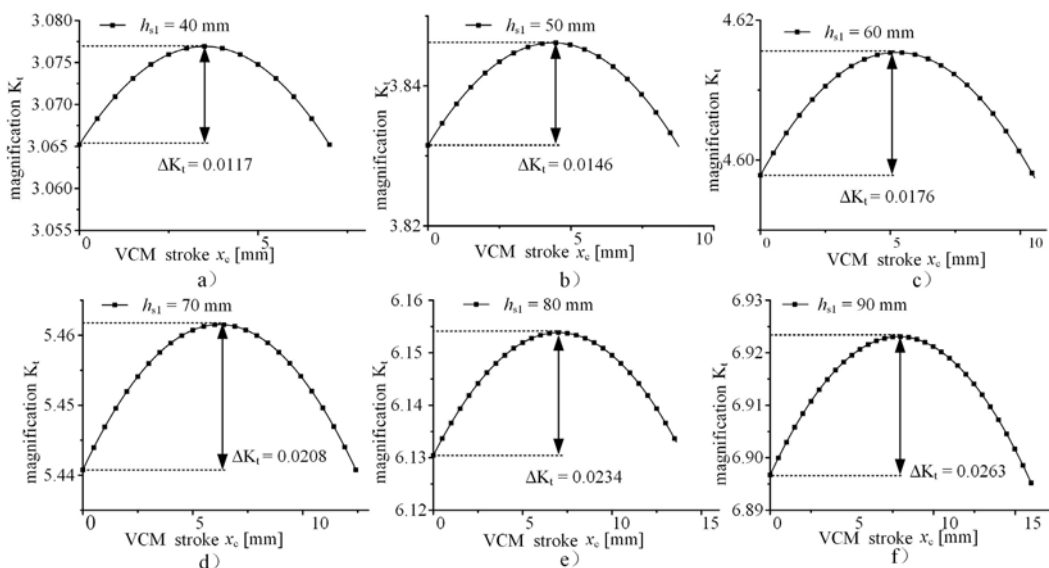


Fig. 10. Relationship between VCM stroke and magnification;

a) $h_{s1} = 40$ mm, b) $h_{s1} = 50$ mm, c) $h_{s1} = 60$ mm, a) $h_{s1} = 70$ mm, a) $h_{s1} = 80$ mm, a) $h_{s1} = 90$ mm

the horizontal direction, the magnification is the maximum. The variation of magnification will lead to the non-linearity of actuation force and stroke that acts on the spool. Therefore, it is necessary to select the reasonable distance h_{s1} to obtain the linear output performance.

Fig. 9a shows the relationship between VCM stroke and spool stroke when h_{s1} is 40 mm to 90 mm. Fig. 9b is the residual sum of squares (SSE) of the spool stroke in Fig. 9a. With the increase of h_{s1} , SSE decreases gradually. When $h_{s1} = 70$ mm, a good linear relationship can be obtained. The magnification is between 5.44 and 5.46, and the fluctuation amplitude of magnification is 0.0208, as shown in Fig. 10.

The results show that with the increase of amplification, the amplification fluctuation (ΔK_t is the amplification fluctuation, as shown in Fig. 10) will gradually increase from 0.0117 to 0.0263. Based on the comprehensive comparison and analysis, the optimized value of h_{s1} is 70 mm. and the optimal magnification can be obtained.

4 EXPERIMENTS AND DISCUSSION

A prototype of the water hydraulic proportional valve is developed. The dynamic performance test rig of

the valve and VCM have been built, respectively. The dynamic and static performances of VCM and the valve are tested.

4.1 Experimental Study on the Performance of VCM

As shown in Fig. 11a, the actuation force test rig of VCM is mainly composed of support frame, force sensor, displacement sensor, step motor, ball screw, adjustable linear power supply, computer, etc. Fig. 11b shows the schematic diagram of the test system. The industrial computer sends control commands to the stepper motor driver and VCM linear driver through the acquisition card, respectively. The stepper motor drives the linear motion platform to move with VCM through the ball screw. The force sensor is fixed on the base and remains stationary. Therefore, by moving the linear motion platform and adjusting the control signal, the VCM actuation force under different stroke can be tested. The displacement sensor acquires the displacement signal, and the force sensor records the VCM actuation force signal every 1 mm interval.

The results presented in Figs. 12 and 13 show the actual test results of VCM actuation force and the difference between simulation and test results. With the movement of the coil, the maximum actuation force

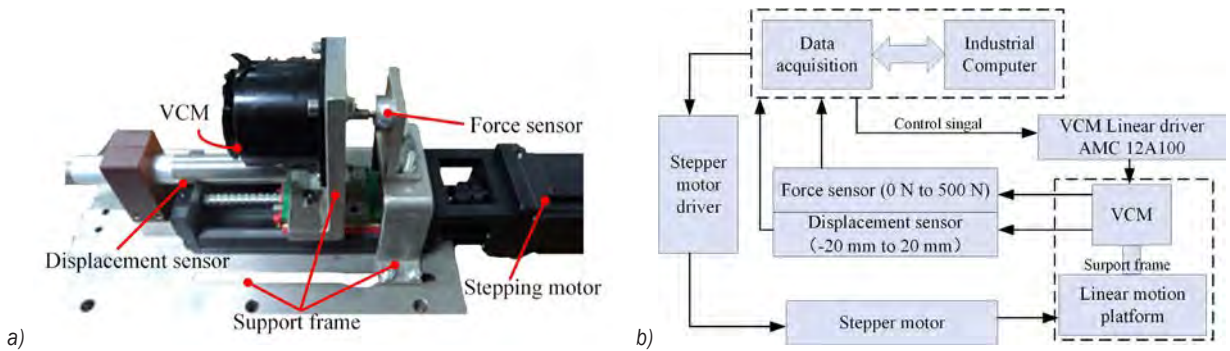


Fig. 11. VCM actuation force testing rig; a) Testing rig of VCM, b) Schematic diagram of test system

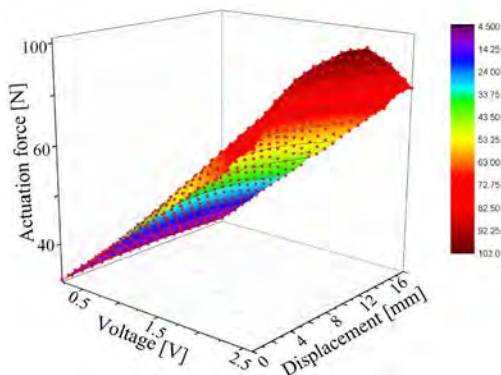


Fig. 12. VCM actuation force test results

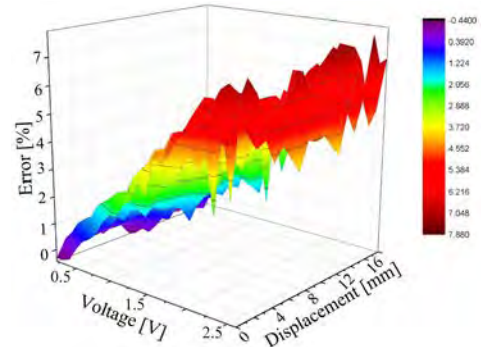


Fig. 13. Difference between simulation and test of VCM actuation force

is generated in the middle section, and the minimum actuation force is generated at both sides of the stroke. The distribution of electromagnetic actuation force is small on both sides and large in the middle section. The distribution characteristics of actuation force are essentially consistent with the simulation results. The simulation value is slightly larger than the real test value, and the maximum difference between them is nearly 8 N (the maximum error is 8.4 %). The distribution of error is irregular and significantly fluctuates with the increase of actuation force and displacement. However, in general, the error increases with the increase of actuation force. Both simulation results and test results show that the actuation force decreases in the initial stroke (0 mm to 4 mm) and the end stroke (15 mm to 18 mm). Therefore, the optimal stroke range is 4 mm to 15 mm.

4.2 Experimental Study on Dynamic Performance of the Valve

As shown in Fig. 14, the test system is mainly composed of pump, relief valve, pressure gauges, pressure transducer, displacement transducer, throttle valve, hydraulic multimeter, etc. During the test, the water pump is used to generate rated pressure and rated flow for the prototype valve. The test pressure can be set by adjusting the relief valve.

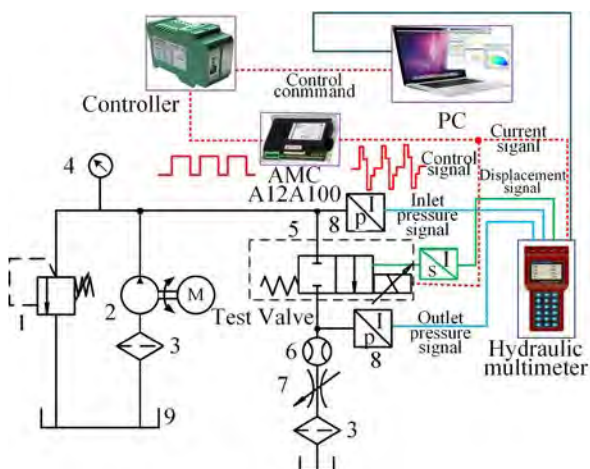


Fig. 14. Principle of the hydraulic test system; 1. Relief valve, 2. Water pump, 3. Filter, 4. Pressure gage, 5. Water hydraulic valve, 6. Flowmeter, 7. Throttle valve, 8. Pressure transducer, 9. Water tank

Pressure sensors are installed at the inlet and outlet of the valve. The two different ranges of the pressure sensors are 0 MPa to 20 MPa and 0 MPa to 35 MPa, respectively. The output signals of the pressure

sensors are both 4 mA to 20 mA. The range of the eddy current displacement sensor is 0 mm to 2 mm. The output signal of the eddy current displacement sensor is 4 mA to 20 mA. The pressure sensors are used to measure the pressure at the inlet and outlet of the valve. The pressure difference (Δp) can be obtained. The eddy current displacement sensor is installed on the valve and used to measure the valve spool stroke (x_v). The PC sends the control commands to the controller. The controller sent control signals into the linear driver (AMC A12A100), and then the control signals will be linearly amplified to drive VCM, in order to control the movement of the valve spool. The displacement signal and pressure signals can be collected in real time with a hydraulic multimeter.

According to the above principle of the hydraulic test system, the test rig is built (as shown in Fig. 15, which is composed of hydraulic power supply, the valve test bench, DC power supply and electrical control system. The linear DC power supplies power to the controller, VCM and sensors. The electric control system realizes the closed-loop control position.

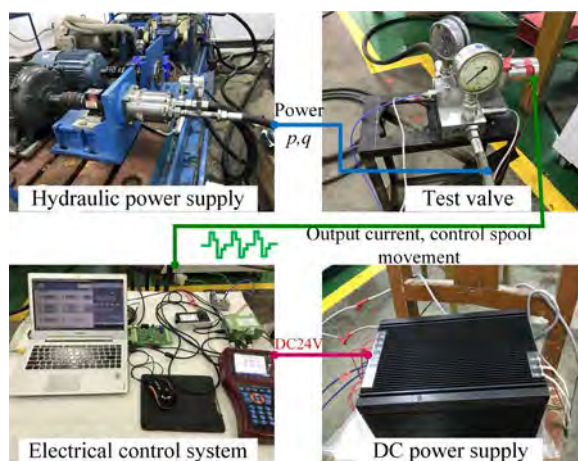


Fig. 15. The valve dynamic performance test system

As shown in Fig. 16, the step response (the spool displacement of 1.0 mm, 0.5 mm and 0.1 mm) is compared with the corresponding simulation results. The simulation results show that the step response adjusting time of the valve is about 28 ms and the maximum overshoot is about 5 %. The step response rise time is about 15 ms. Furthermore, the test results indicate that the adjusting time of the opening process is about 30 ms, and the adjusting time of the closing process is about 35 ms. The test results show that there is no obvious delay in the process of opening and closing. It takes more time for the valve spool to close than to open. The actuation force is provided

by VCM when the valve is opening. As the return spring closes the valve when the VCM returns. This is the main reason for the difference between the opening and closing time. The simulation results are in good agreement with the experimental results. An approximately 10 % overshoot occurs when the spool displacement is 1.0 mm, while there is no overshoot when the spool displacement is 0.5 mm and 0.1 mm. The rising time is about 12 ms to 15 ms. The valve has good dynamic performance, and the valve opening and closing process is controlled smoothly.

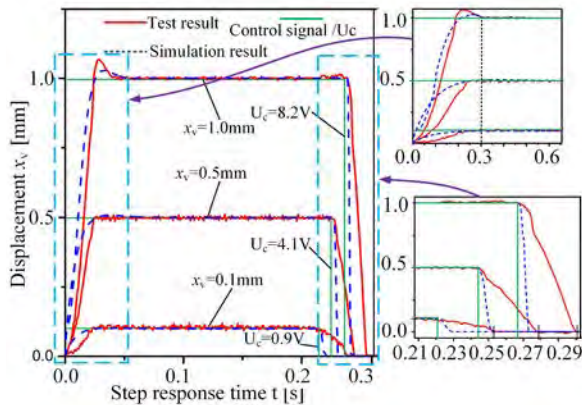


Fig. 16. Step response test and simulation

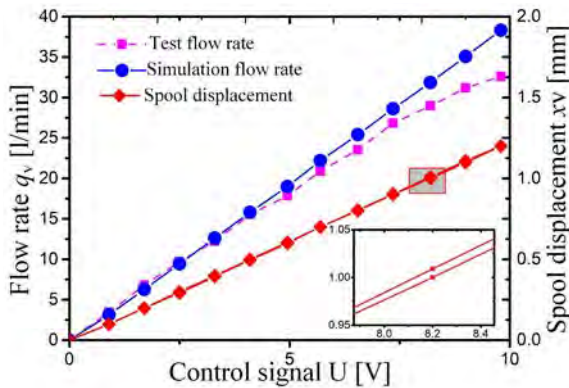


Fig. 17. Control signal and flow characteristics

The control signal-spool displacement curve and control signal-flow characteristic curve are shown in Fig. 17. During the test, the inlet pressure of the valve is kept at 1 MPa by adjusting the relief valve. The outlet is connected to the water tank, and the pressure difference between the inlet and outlet of the valve is kept at 1 MPa. The maximum displacement of the valve spool is 1.2 mm in the range of 0 V to 10 V control signal. The control signal has the good linear relationship with the displacement of the valve spool, and the position hysteresis of the valve is less than 1 %. By comparing the experimental flow rate

with the simulation results, it can be found that the test flow rate is in good agreement with the simulation flow rate when the control signal is less than 6.5 V. However, the greater the flow rate, the higher the deviation. When the control signal is more than 6.5 V, there is a certain deviation between them, which is mainly caused by the machining error of the valve orifice and the pressure fluctuation of the test system. Generally, the control signal of the valve has a good linear relationship with the flow rate.

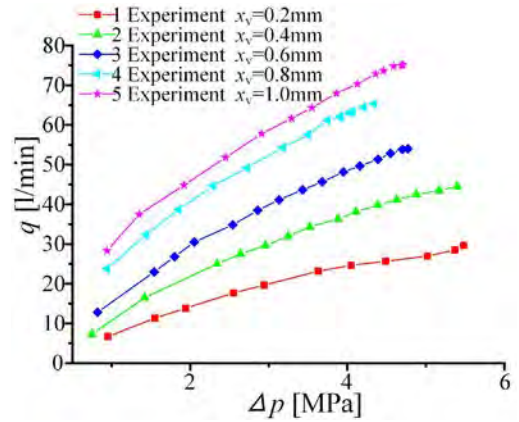


Fig. 18. q and Δp characteristic

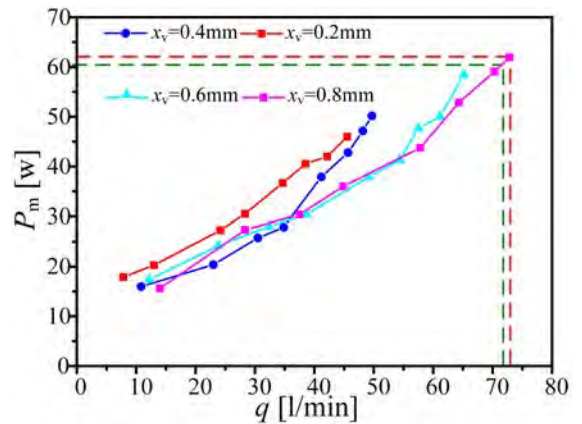


Fig. 19. The valve input power characteristics

Fig. 18 shows the pressure difference (Δp) and flow rate (q_v) characteristic under different valve opening ($x_v = 0.2$ mm, 0.4 mm, 0.6 mm, 0.8 mm, 1.0 mm) in detail. There is irregular change when Δp is over 4 MPa at the valve opening $x_v = 0.2mm$. This is probably caused by the machining errors at the valve orifice. In general, the results indicate that the Δp and q_v characteristics curve is smooth and linear. The valve has good control performance. The relationship between the flow rate and VCM power under different valve opening is shown in Fig. 19. With the increase

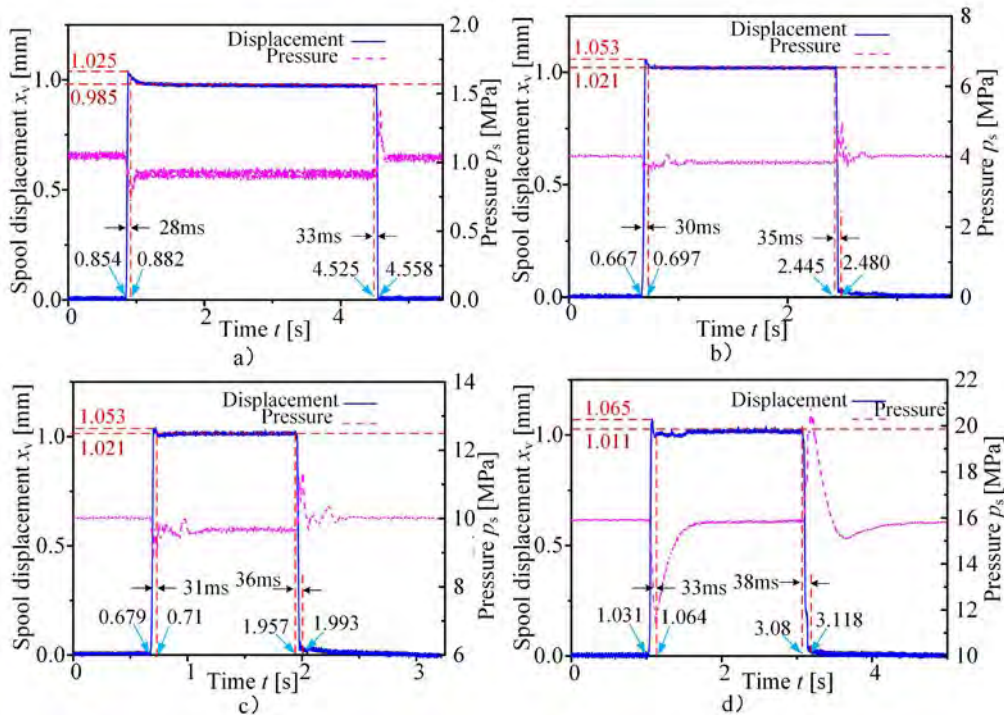


Fig. 20. Step response under different pressure, a) $\Delta p = 1$ MPa, b) $\Delta p = 4$ MPa, c) $\Delta p = 10$ MPa, d) $\Delta p = 16$ MPa

Table 4. Step response characteristics comparison

Manufacturer	MOOG	Tested valve	ATOS	Danfoss
Model	D633/D634	Prototype	QVKZOR-A*-10	VOH30PE
Type	Direct drive servo valve	Proportional valve actuated by VCM	Proportional valve actuated by solenoid	Proportional valve actuated by solenoid
Hydraulic medium	Mineral oil	Water	Mineral oil	Water
Max Pressure	35 MPa	25 MPa	21 MPa	14 MPa
Response time (0 % to 100 % Stroke)	≤ 12 ms ≤ 20 ms	30 ms	45 ms	≤ 150 ms

of the flow rate, the power required to drive the spool will also increase. When the flow rate is 70 l/min, the corresponding VCM power is 60 W, which can meet the requirements of long-term stable and reliable operation of the valve.

The step response test of the valve under different pressures is shown in Fig. 20. During the test, the inlet pressure of the valve is maintained at 1 MPa, 4 MPa, 10 MPa and 16 MPa by adjusting the relief valve, while the outlet is connected with the water tank. Thus, the outlet pressure of the valve is always kept at 0. According to the test results, it can be seen that the maximum overshoot will increase slightly with the increase of the inlet pressure. Nevertheless, the valve can maintain the rapid response performance. The maximum overshoot is approximately 10 %. In general, the adjusting time of the valve is about 30

ms in the opening process and 35 ms in the closing process.

Table 4 shows the comparison between the new water hydraulic valve and the other proportional valves in the market. The step response characteristics of the new valve presented in this paper are better than those of the similar type of oil hydraulic proportional valve. While compared with the servo valve, there is a gap in the dynamic performance. However, compared with the traditional water hydraulic proportional valve (such as Danfoss VOH30PE), the new water hydraulic valve actuated by VCM has better dynamic performance. The results prove that VCM is a good solution instead of solenoids for the hydraulic control valve.

5 CONCLUSIONS

In this study, a water hydraulic proportional valve with fast response is proposed. Given that VCM has the advantages of high speed and high control accuracy, it has been used as the electrical-mechanical conversion device for the valve. Due to the poor viscosity and lubricity of the water medium, the valve will need a large force to push the spool. Thus, a lever amplifier is used to amplify the VCM actuation force. The position feedback closed-loop control has been used to improve the dynamic performance and anti-interference ability of the valve.

A detailed mathematical model of the valve has been developed. The simulation models are established in the MATLAB/Simulink and Maxwell 3D platform, respectively. A comprehensive optimization design method has been proposed. The test rig of the valve and VCM have been built. The dynamic and static performances of VCM and the valve have been tested. Both the VCM actuation force test and simulation results show that the optimal stroke range is 4 mm to 15 mm. According to the dynamic response test, the maximum overshoot of the valve is approximately 10 %, the adjusting time is about 30 ms in the opening process, and 35 ms in the closing process. The test results prove that the valve can maintain a fast response speed under different pressures. Compared with the traditional water hydraulic proportional valve, the new water hydraulic valve actuated by VCM significantly improved the dynamic performance. VCM is a good solution instead of solenoids for the water hydraulic proportional valve. The water hydraulic proportional valve designed in this paper has good static and dynamic control performance.

6 ACKNOWLEDGEMENTS

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Virtual Minimization of Residual Stress and Deflection Error in the Five-Axis Milling of Turbine Blades

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To simulate and analyse the real machined parts in virtual environments, virtual machining systems are applied to the production processes. Due to friction, chip forming, and the heat produced in the cutting zone, parts produced using machining operation have residual stress effects. The machining force and machining temperature can cause the deflection error in the machined turbine blades, which should be minimized to increase the accuracy of machined blades. To minimize the residual stress and deflection error of machined parts, optimized machining parameters can be obtained. In the present research work, the application of a virtual machining system is presented to predict and minimize the residual stress and deflection error in a five-axis milling operations of turbine blades. In order to predict the residual stress and deflection error in machined turbine blades, finite element analysis is implemented. Moreover, to minimize the residual stress and deflection error in machined turbine blades, optimized parameters of machining operations are obtained by using a genetic algorithm. To validate the research work, experimentally determining residual stress by using a X-ray diffraction method from the machined turbine blades is compared with the finite element results obtained from the virtual machining system. Also, in order to obtain the deflection error, the machined blades are measured by using the CMM machines. Thus, the accuracy and reliability of machined turbine blades can be increased by analysing and minimizing the residual stress and deflection error in virtual environments.

Keywords: virtual machining, residual stress, deflection error, parameter optimization, turbine blade

Highlights

- The application of a virtual machining system is presented to predict and minimize the residual stress and deflection error in the five-axis milling operations of turbine blades.
- In order to minimize the residual stress and deflection error in machined turbine blades, an optimization technique based on genetic algorithms is used.
- The accuracy and reliability of machined turbine blades can be increased by analysing and minimizing the residual stress and deflection error in virtual environments.

0 INTRODUCTION

The residual stress in machined parts can be generated as a result of mechanical, thermal, and chemical effects in chip forming of metal cutting operations. The generated residual stress in the machined parts can impair the performance of components, such as fatigue life, corrosion resistance, and part distortion in actual working conditions. Due to the cutting forces and cutting temperature, the machined blades have deflection errors, which can cause inaccuracy in the machined turbine blades. As a result, the residual stress and deflection error should be analysed and decreased to increase the accuracy and reliability of parts produced using machining operations.

To improve the precision of machined parts, the evaluation of surface error characteristics in thin-walled constructs during peripheral milling has been studied by Wimmer and Zaeh [1]. The process parameter optimization of thin-wall machining for wire arc additive manufactured parts is investigated by Grossi et al. [2] to decrease the deformation error in the thin-walled manufactured components. The method of error compensation during the milling operations

of flexible thin-wall parts is presented by Ratchev et al. [3] to reduce the deflection error of machined parts. To improve the accuracy of thin-walled machined parts, an adaptive toolpath methodology for three-axis milling is presented by Grossi et al. [4]. The finite element method (FEM) based cutting velocity selection for thin-walled part machining is presented by Scippa et al. [5] to enhance the precision of machined components. Finite-Element modelling of workpiece vibrations is analysed by Bolsunovskiy et al. [6] to optimize the machining parameters and increase accuracy in the milling operations of thin-walled components.

Jiang et al. [7] analysed the effects of cutting forces and cutting zone temperature to the residual stress of machined components. The application of response surface methodology in obtaining the optimized machining parameters, such as depth of cut and spindle speed, are investigated by Masmiaati et al. [8] in order to minimize the residual stress, cutting force, and surface roughness in the end milling of S50C medium carbon steel. Mohammadpour et al. [9] investigate the effects of machining parameters on the residual stress of machined parts in milling

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operations. To analyse the influence of the machining parameters, such as cutting speed as well as feed rate, on the surface roughness and residual stresses in produced parts using milling operation, the finite element method is used by Zhang and Wu [10].

To decrease the residual stress in machined parts using turning operations, the effects of cutting tool parameters, such as tool nose radius and tool wear on residual stresses, are investigated by Lin et al. [11]. Cutting tool parameters, such as tool edge radius, are investigated by Yang et al. [12] in order to decrease the residual stress in machined parts. To study the effects of cutting-edge radius and cutting forces on residual stresses of machined parts, a finite element model is developed by Nasr et al. [13].

To decrease the residual stress in machined parts and effects of cutting tools materials, machining parameters, such as cutting speeds and depth of cut to residual stress of machined parts, are investigated by Arunachalam et al. [14]. The influence of cutting tool parameters, such as tool nose radius on surface integrity and residual stresses, are investigated by Sharman et al. [15] to decrease the residual stress in machined parts. A variable depth-of-cut milling strategy for thin-walled workpiece is proposed by Yan et al. [16] in order to decrease the deflection error in the machined turbine blades. A five-axis adaptive flank milling of flexible thin-walled parts based on the on-machine measurement is developed by Huang et al. [17] to modify the cutting tool paths in terms of deflection error minimization. The time domain flutter analysis based on the blade element momentum (BEM) theory for bend-twist coupled large composite wind turbine blades is investigated by Shakya et al. [18] to increase the performances of wind turbines in actual working conditions. The blisk vibration in aircraft engine using similitude models is analysed by Luo et al. [19] to predict and decrease the vibration of the engine systems in the virtual environments. To analyse and modify the machining operations in virtual environments, virtual machining systems and applications are presented by Soori et al. [20] to [24]. To analyse and develop the process of part manufacturing using welding operations, a review in recent development of friction stir welding process is presented by Soori et al. [25]. To increase efficiency in process of part production using the friction stir spot welding, effects of machining parameters to the mechanical properties of aluminium sheet alloys is presented by Nasir et al. [26].

According to the analysis of previous published papers, the area of residual stress as well as deflection error prediction and minimization in milling

operations by using virtual machining systems has been insufficiently explored. Moreover, to minimize the residual stress and deflection errors in machined parts, the applications of the virtual machining system is not studied.

In this research work, the application of a virtual machining system is presented to predict and minimize the residual stress and deflection error in a five-axis milling operations of turbine blades. The cutting forces as well as cutting temperatures for each position of cutting tool along machining paths are calculated in order to obtain residual stress and deflection error of blades due to machining operations by using the finite element analysis (FEA). The optimization technique of a genetic algorithm is used to calculate the optimized machining parameters in terms of residual stress and deflection error minimization. To measure the residual stresses on the surface and in-depth of the machined blades, the X-ray diffraction technique is used in the study. The coordinate measuring machines (CMM) machines are used to obtain the deflection error in the machined blades. Finally, the obtained data are compared by using diagrams.

1 CUTTING FORCE MODEL

To calculate the cutting forces in the five-axis machining operations, the cutting force model developed by Zhang et al. [27] is applied.

The XYZ planes are defined in order to describe the tool motion process, workpiece geometry and tool path in machining operations. Also, X_r, Y_r, Z_r is created in order to describe the tool rotation coordinate system in machining operations. Thus, the cutting forces in the differential format, which is applied to the j^{th} axial disk element of the i^{th} tooth at rotation angle $\phi_{i,j}(t)$, can be presented as:

$$\begin{cases} dF_{T,i,j}(t) = (K_{ts}h(i, j, t)db + K_{tp}db)W(\theta(\phi_{i,j}(t))) \\ dF_{R,i,j}(t) = (K_{rs}h(i, j, t)db + K_{rp}db)W(\theta(\phi_{i,j}(t))), \\ dF_{A,i,j}(t) = (K_{as}h(i, j, t)db + K_{ap}db)W(\theta(\phi_{i,j}(t))) \end{cases} \quad (1)$$

where $h(i,j,t)$ is the instantaneous uncut chip thickness of the j^{th} axial disk element of the i^{th} tooth at t moment and the db is the axial height of the cutting disk element. Moreover, the coefficients of shearing specific cutting force in different directions as tangential, radial, and axial can be shown as:

$$k_{qs} = a_{qs} + b_{qs}e^{c_{qs}h(t)} \quad (q = t, r, a).$$

Also, the coefficients of ploughing specific cutting force in different directions as tangential, radial, and axial direction can be presented as:

$$k_{qp} = a_{qp} + b_{qp} e^{c_{qp}h(t)} \quad (q = t, r, a).$$

To determine whether the condition of disk element is in or out of cutting operation, the $W(\theta)$ can be presented as:

$$w(\theta(\phi_{i,j}(t))) = \begin{cases} 1 & \theta_{st,i,j} \leq \theta(\phi_{i,j}(t)) \leq \theta_{ex,i,j} \\ 0 & \text{others} \end{cases}, \quad (2)$$

where $\theta_{ex,i,j}$ stand for entry and exit angle of j^{th} disk element on i^{th} tooth respectively, as shown in Fig. 1 [27].

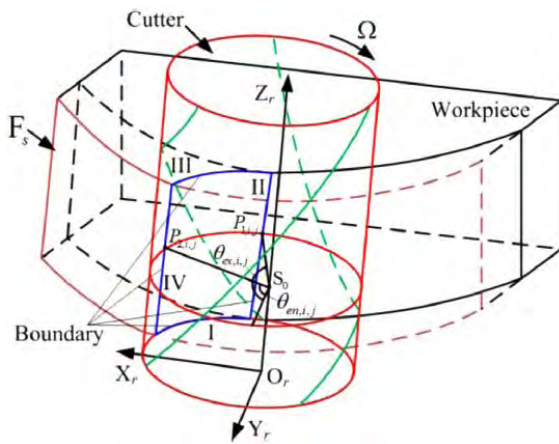


Fig. 1. Entry and exit angle model of cutting tool [27]

Thus, the overall cutting forces that are applied to the t moment can be calculated by summing three cutting forces acting on disk element within axial depth of cut as is shown in Eq. (3).

$$\begin{bmatrix} F_{c,x}(t) \\ F_{c,y}(t) \\ F_{c,z}(t) \end{bmatrix} = \sum_i \sum_j Rot \begin{bmatrix} dF_{T,i,j}(t) \\ dF_{R,i,j}(t) \\ dF_{A,i,j}(t) \end{bmatrix}, \quad (3)$$

where Rot is the cutter rotation matrix. With the aid of transformation relationship matrix in the five-axis machine tool, the total cutting force components acting on the workpiece can be obtained by:

$$\begin{bmatrix} F_{w,x}(t) \\ F_{w,y}(t) \\ F_{w,z}(t) \end{bmatrix} = -RotC \cdot RotB \begin{bmatrix} F_{c,x}(t) \\ F_{c,y}(t) \\ F_{c,z}(t) \end{bmatrix}, \quad (4)$$

where $RotC \cdot RotB$ is rotation matrix for B -axis and C -axis.

2 CUTTING TEMPERATURE MODEL

In the machining operations, the chip formation process generates heat in the cutting zone. To model the cutting temperature regarding the cutting parameters, the response surface methodology (which is statistical) and mathematical techniques can be applied. The experimental values of cutting temperature corresponding to the control variables of machining parameters are incorporated into the response surface methodology in terms of cutting temperature model modification [28]. As a result, The equation of the cutting temperature regarding the cutting parameters based on response surface methodology is presented by Shaw [29] as,

$$\lambda = C_\lambda V^{b_1} f_z^{b_2} a_e^{b_3} a_p^{b_4}, \quad (5)$$

where λ is temperature [$^{\circ}\text{C}$], C_λ is temperature coefficient determined by workpiece material, machine tool, and cutting tool geometry parameter. The $b_1, b_2, b_3,$ and b_4 are exponents influencing the machining parameters V , cutting speed (m/min), f_z , feed rate (mm/rotation), a_e , radial feed [mm] and a_p , axial feed [mm].

To obtain the experimental values of cutting temperature and calculate the cutting temperature during machining operation, the tool-workpiece thermocouple device with three brushes is used. The electromotive force produced in the tool-workpiece thermocouple circuit during machining is proportional to the temperature difference between the tool and room temperature. Extruded round bars (101 mm \times 2000 mm) of the aluminium alloys 1350-O and 7075-T6 were used as the workpiece. The cutting tool used in the experiment is carbide bar (310 \times 10 \times 4) mm K15 grade, and the tool holder is (25 \times 20 \times 150) mm made of steel. The cutting speed 600 m/minute, feed rate 0.3 mm/rotation and axial depth of cut 3 mm are considered as machining parameters of the experiments. The average of the chip-tool temperatures measured by the tool workpiece thermocouple system (average of two replicates) are calculated using analysis of variance of the 2^k factorial design as input variables. The coefficients of the temperature model are then calculated using a statistic analysis based on multiple nonlinear regression of the chip-tool interface temperature over 32 experiments. As a result, the equations of the cutting tool and workpiece temperatures in the machining operations of Al alloys based on multiple nonlinear regression analysis are presented by Santos et al. [30] as:

$$\lambda_t = 116.503V^{0.211}f_z^{0.181}a_e^{0.0464}a_p^{0.00391}, \quad (6)$$

$$\lambda_w = 43.319V^{0.365}f_z^{0.244}a_e^{0.0423}a_p^{0.019}, \quad (7)$$

where λ is temperature [°C], V is cutting speed [m/min], f_z is feed rate [mm/rotation], a_e and a_p are radial feed and axial feed [mm], respectively.

3 DEFLECTION ERROR ANALYSIS OF THIN-WALL WORKPIECE IN MACHINING OPERATIONS

The machining operations of thin-walled parts have deflection errors as a result of cutting forces as well as cutting temperature. The workpiece deflects to a new position due to cutting forces and cutting temperatures which creates dimensional errors and inaccuracies in the process of machining operations.

The corresponding surface dimensional error can be presented as Eq. (8) [31],

$$e_p = \delta_{t,p} + \delta_{f,p}, \quad (8)$$

where, $\delta_{t,p}$ and $\delta_{f,p}$ are the normal projections of the cutting force and cutting temperature-induced deflection error for the Point P, respectively.

For the convenience of study, the distance between the initial surface to be machined and the desired machined surface is named as the nominal radial depth of cut notated by R_N . In the actual milling process, to ensure that the surface dimensional error does not violate the tolerance, R_A is often specified to be different from $R_N \neq R_A$. In this case, Eq. (9) is no longer applicable to the calculation of surface dimensional error and must be corrected as [31]:

$$e_p = \delta_{t,p} + \delta_{f,p} + R_N - R_A. \quad (9)$$

Note that R_N and R_A are the nominal and specified radial depth of cut, respectively.

4 VIRTUAL MACHINING SYSTEM

The Visual Basic programming language is used in this study to develop the presented virtual machining system. The developed virtual machining system can obtain the cutting forces regarding the cutting tool details and parameters of machining operations. Thus, cutting forces at each position of the cutting tool along machining paths can be calculated.

The software calculates the cutting temperature regarding the cutting process parameters at each position of the cutting tool along machining paths. Then, the software is linked to the Abaqus R2016X FEM analysis software to analyse the residual stress and deflection error due to machining operations.

The calculated cutting forces and cutting temperature at each position of cutting tool along the machining paths in turbine blade machining are input to Abaqus software to be used for residual stress and deflection error calculations. As a result, the residual stress and deflection error due to cutting forces and cutting temperatures at each position of the cutting tool can be calculated and presented. A flowchart and the strategy of the virtual machining system in cutting force calculation, residual stress and deflection error prediction are shown in Fig. 2.

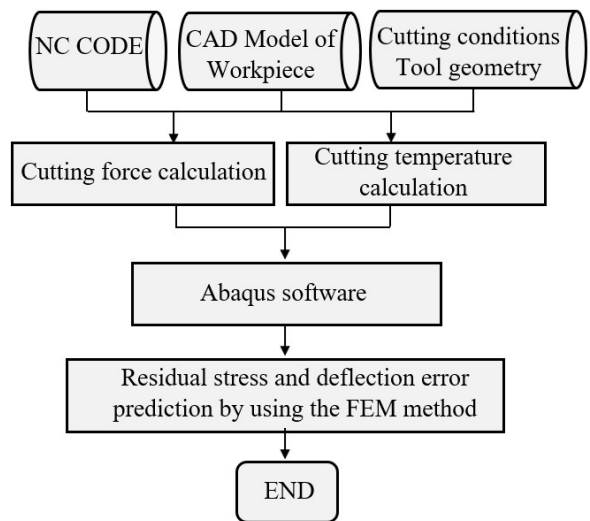


Fig. 2. Flowchart of the virtual machining system

To obtain the optimized machining parameters in terms of residual stress and deflection error minimizations, the genetic algorithm is applied. The genetic algorithm is an advanced optimization method based on natural selection by which the fittest individuals are selected for reproduction to produce offspring of the next generation. The natural process of evolution and set of chromosomes is introduced in terms of optimization process. The initial population for the optimization process is created using the binary encoding process. To provide evaluation criteria in the optimization process and rank the chromosomes in the population, the fitness function is calculated as is presented in Eq. (10) [32].

$$F(x) = \frac{1}{1 + f(x)}, \quad (10)$$

where fitness function and objective function are $F(x)$ and $f(x)$, respectively. The main operators of the algorithms are reproduction, crossover, and mutation. The operators are applied to the initial population of

the optimization process in order to provide a faster convergence to the optimized parameters.

The mathematical equation of the surface roughness in end milling operations is presented in Eq. (11) [33].

$$R_a = 318 \frac{f^2}{4d}, \quad (11)$$

where f is the feed rate, and d is the diameter of the cutting tool. Moreover, the time of machining operation should be minimized in order to decrease the cost of machined parts. The mathematical equation of machining time is presented in Eq. (12) [33].

$$t_m = \frac{k}{f}, \quad (12)$$

where k is the distance of cutting tool to reach to the operational zone and f is feed rate.

Also, the cutting tool life should be maximized in the optimization process in order to decrease the cost of machining operation. The cutting tool life can be shown as Eq. (13) [33].

$$T_L = \left(\frac{60}{Q} \right) \left[\frac{C \left(\frac{G}{5} \right)^{\frac{1}{m}}}{V(A)^w} \right], \quad (13)$$

where Q is the contact proportion of cutting edge with workpiece per revolution, C is 33.98 for the high-speed steel (HSS) tools and 100.05 for the carbide tools, $g = 0.14$, V is cutting speed [mm/min], $w = 0.28$, m is 0.15 for HSS tools while it reaches a maximum of 0.30 for carbide tools, G and A are slenderness ratio and chip cross-section, which can be shown as Eqs. (14) and (15), respectively [33].

$$G = \frac{a}{f}, \quad (14)$$

$$A = a \cdot f, \quad (15)$$

where a is a depth of cut and f is feed rate in machining operations.

According to Eqs. (6) and (7), the cutting temperature can be decreased by reducing the feed rate and spindle speed. Also, according to Eq. (11), the surface roughness can be decreased by reducing the feed rate. However, the machining time will be increased by reducing the feed rate according to Eq. (12). Moreover, increasing the spindle speed can decrease the cutting forces according to Eq. (4) and cutting tool life according to Eq. (13). It is clear that the presented mathematical models of the cutting

forces, cutting temperature, time of machining operations, surface roughness, feed rate and cutting tool life are related, which should be considered as an optimization problem. The objective function of the optimization process is minimizing the residual stress and deflection error by calculating the optimized machining parameters. To obtain the optimized machining parameters in order to minimize the residual stress and deflection error using the minimized cutting forces and cutting temperature in machined turbine blade, the Matlab programming language is used. As a result, the optimized feed rate and spindle speed are obtained in order to minimize the residual stress as well as deflection error in the machined parts. The complete methodology of the study is shown in Fig. 3.

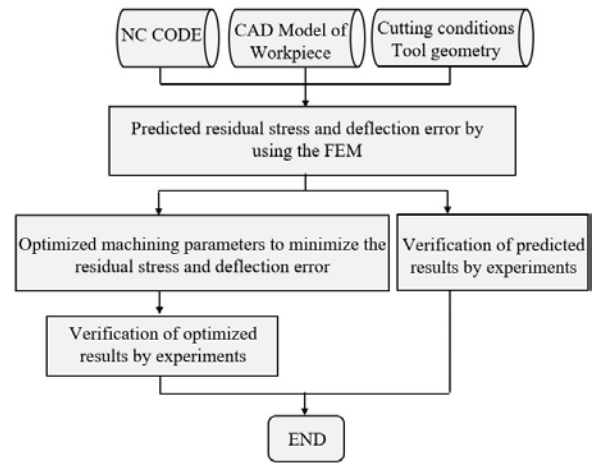


Fig. 3. Methodology of the study

5 VALIDATION AND COMPARISON

In order to validate the presented virtual machining system in the study, real turbine blades are machined by using the A 5-axis computer navigated control (CNC) milling machine tool Kondia HM 1060. The turbine blade material is AL 2618, which is used in the compressor section of gas turbines. Then, the test turbine blades are machined by using the five-axis Kondia HM 1060 CNC machine tool. The cutting tool used in the experiment is a carbide flat end mill with 10 mm diameter, helix angle 30°, flute number 4, overall length 60 mm and flute length 30 mm. The feed rate and spindle speed are 100 mm/min and 1000 rpm, respectively. The machining process of the turbine blade is shown in Fig. 4.

To calculate the cutting forces in virtual environments, the cutting force model of five-axis

CNC machine tools presented by Zhang et al. [27] is used in this study.



Fig. 4. Machining process of the turbine blades

In order to calculate the coefficients of specific cutting force, the average of cutting forces for twenty slot milling tests with 1.5 mm axial depth of cut using 5-axis CNC milling machine tool Kondia HM 1060 are measured with a Kistler dynamometer. The spindle rotating speed is 5000 rpm and the feed per tooth and feed rate are 0.5 mm and 100 mm/min, respectively. The cutting tool overhang is 60 mm. As result, the specific cutting force coefficients are obtained as Fig. 6.

The real machined Al turbine blades are shown in Fig. 5.



Fig. 5. The real machined turbine blades

The cutting forces and cutting temperatures for each position of the cutting tool along machining paths are obtained by using the virtual machining system. Then, the cutting forces and cutting temperatures are

entered into Abaqus software to obtain the residual stress and deflection error of machined blades. Thus, the calculated residual stresses by Von Mises yield criterion due to machining operation of turbine blade in MPa unit are shown in Fig. 7.

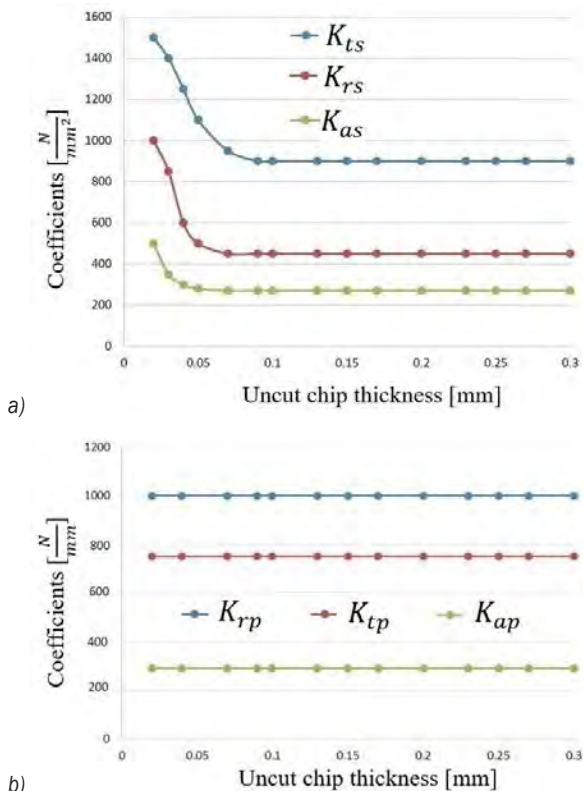


Fig. 6. Specific cutting force coefficients, a) K_{LS} , K_{rs} , K_{as} and b) K_{rp} , K_{tp} , K_{ap}



Fig. 7. The calculated residual stresses by the Von Mises yield criterion due to machining operation of turbine blade in MPa unit

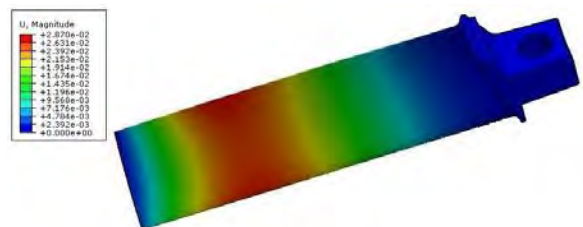


Fig. 8. The calculated deflection error of machined turbine blade

The calculated deflection error of the machined turbine blade by using the FEM method is presented in Fig. 8.

Residual stresses of the machined blade are measured by using an X-ray diffraction method. In order to obtain the residual stress in the machined turbine blades, two-circle Bragg Brentano-Diffractometer of the type Philips X'PERT MPD is used by Crk_{α} radiation with 35 kV and 35 mA. The $\sin 2\psi$ method is the most widely used methodology for stress assessment. Several XRD measurements are performed by considering the different tilts angle of ψ . In this experiment, the tilting angles are set at 0° , 13.8° , 18.7° , 24.3° , 29.4° , 32.2° , 34.8° , 36.5° , and 40.4° . The inter-planar spacing or two-theta peak is calculated in order to obtain a plot or a curve for the measured data. As a result, the stress can be determined using the obtained plot by measuring the line gradient and with simple knowledge of the material's elastic properties. In order to measure the residual stress in the machined turbine blades, the X-ray diffraction method is used, as shown in Fig. 9.



Fig. 9. The X-ray diffraction method to measure the residual stress of machined turbine blade

The residual stresses in the depth of machined blade are measured and presented in the Fig. 10.

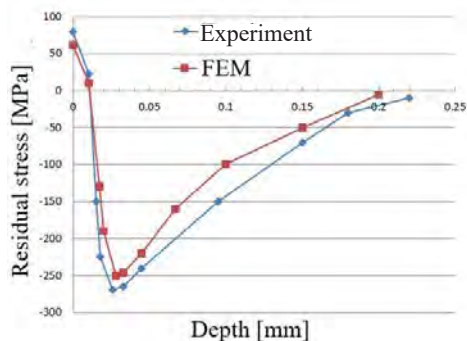


Fig. 10. Measured residual stress in depth of produced blade

A 90.4 % compatibility is obtained in the comparison of the results of the experimental test and FEM simulation. Optimized machining parameters are calculated using the optimization techniques developed in the study based on the genetic algorithm. To measure the cutting forces in the machining operation, the Kistler dynamometer is used. To provide input data for the optimization process, the measured cutting forces and cutting temperature, cutting condition and cutting tool geometry are entered in the optimization algorithm. In the optimization process, a population size of 28 is selected with the iterated for 228 generations. Also, the probability of crossover of 0.85 and mutation of 0.001 are selected. As a result, the feed rate 87 mm/min, radial feed a_e 0.2 mm and axial feed a_p 12 mm and the spindle speed 1269 rpm are obtained. Measured residual stress in the depth of machined blade using optimized machining parameters is shown in Fig. 11.

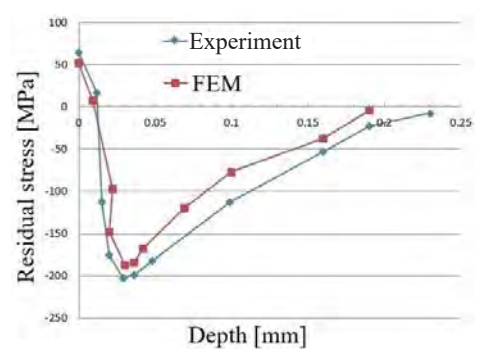


Fig. 11. Measured residual stress in depth of machined blade using optimized machining parameters

In order to measure the deflection error in the machined turbine blades, the CMM machine is used. The used probe in the error measuring is the Renishaw RSH 250 probe while its repeated accuracy is $1 \mu\text{m}$ in touching directions. The sweep scan method is used to obtain the surface condition of the machined blade.

A NURBS surface "patch" is generated for each of the sweep scans in order to obtain the deflection error in the machined turbine blades. Then, to obtain the deflection error, sections are taken along the generated surfaces at any required height. The amount of error between the CAD model of turbine blade and fitted NURBS surfaces from the measured data of the CMM machine are calculated in each section of generated surface as the deflection error of machined turbine blade.

The process of surface generation and deflection error calculation due to cutting forces and cutting

temperature for the machined blade are shown in Fig. 12.

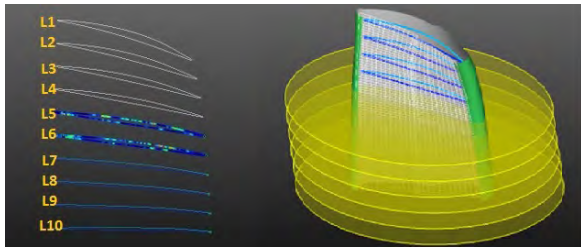


Fig. 12. The deflection error calculation from the CAD model of turbine blade and generated NURBS surfaces of machined turbine blades

As a result, the measured and predicted deflection errors in the machined turbine blade without and with the optimized machining parameters for the L7 line in Fig. 12 is presented in Fig. 13.

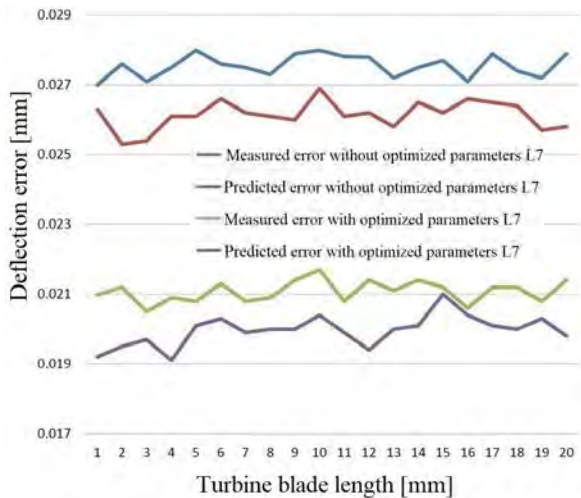


Fig. 13. The measured and predicted deflection errors in the machined turbine blade without and with optimized machining parameters for the L7 line in the Fig. 12

The predicted deflection error due to the cutting forces and cutting temperatures without optimized machining parameters for the L7 line in Fig. 12 is presented in Fig. 14.

The optimized machining parameters can decrease the unnecessary cutting forces and the cutting temperature in the machining operation. Therefore, a reduction in the residual stress of produced blades using optimized machining parameters can be achieved, which can increase accuracy as well as reliability of machined turbine blades. Moreover, the deflection error due to machining forces and cutting temperature is minimized to increase the accuracy of machined turbine blades.

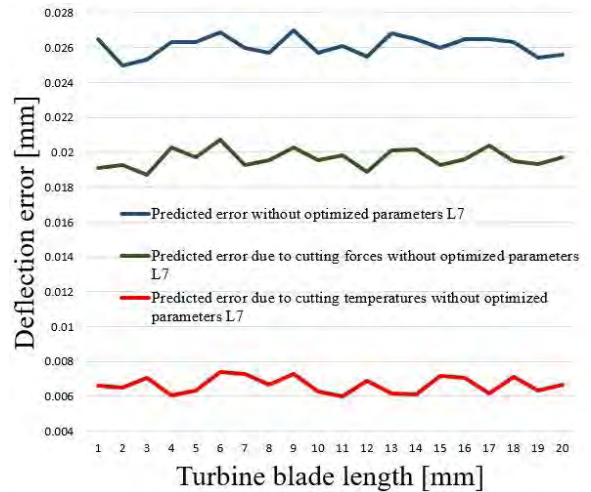


Fig. 14. The predicted deflection error due to the cutting forces and cutting temperatures without optimized machining parameters for the L7 line in the Fig. 12

6 CONCLUSIONS

To predict and minimize residual stress and deflection error in a five-axis turbine blade milling operations, a virtual machining system is developed in the study. The system can calculate the cutting forces and cutting temperature at each position of the cutting tool along machining paths with regard to the parameters of machining operations, geometries, and material properties of cutting tools as well as workpieces. Then, FEA is used to obtain residual stress and deflection error of the turbine blades in machining operations. The real turbine blades are produced and experimentally tested using a five-axis CNC milling machine tool in order to validate the developed system in the study. Then, the residual stress in the machined turbine blades are measured using X-ray diffraction machines. To measure the deflection error in the machined blades, the CMM machine is used. Finally, the obtained results from the virtual machining system and experimental tests are compared to present the rate of compatibility in the developed system. Therefore, a 90.4 % and 94.3 % compatibility are obtained in comparison to the experimental and virtual machining system results for the residual stress and deflection error, respectively.

To minimize the residual stress and deflection error in machining operation, optimized machining parameters using the genetic algorithm is obtained. As a result, 24.2 % and 23.9 % reductions in residual stress can be obtained in the real machined turbine blade and virtual machining system results, respectively. Moreover, 26.3 % and 24.1 % reductions

in the deflection error of real and virtual machining systems results are respectively measured by using the optimized machining parameters in machining operations.

The accuracy and the reliability of machined turbine blades can be increased by using the application of the virtual machining system in the minimization of residual stress and deflection error due to machining operations. Furthermore, the amount of the deflection errors due to cutting forces and cutting temperatures can be accurately predicted in virtual environments using the FEM analysis. The system developed in this study can be used in the machining operations of free-form surfaces in aircraft airfoils by using the five-axis CNC machine tools in order to decrease the residual stress of machined parts. These are the concepts of future research works for the authors.

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Ultrasonic Scattering Attenuation in Nodular Cast Iron: Experimental and Simulation Studies

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This work evaluates the ultrasonic scattering attenuation of structures with complex scatterer distributions via experimental and simulation studies. The proposed approach uses experimental attenuation knowledge to infer the scatterer size and its concentration in the studied structures, which are important for the effective construction of simulated models. The MATLAB k-Wave toolbox has been used to implement the simulator. Several cast-iron samples have been used to demonstrate the importance of simulation in the characterization of such structures. First, the scattering attenuation was evaluated using the Truell and Papadakis models, and then the results were compared with experimental ones. Emphasis was given to the Papadakis approach because it takes into account the scatterer size distribution. It is demonstrated that both analytical models provide results that are far from the experimental ones. The developed simulator for the studied samples led to a predictive model, in which the attenuation was proportional to the fifth power of the scatterer size, and the corresponding formulation is close to the one proposed by the analytical models.

Keywords: modelling, anisotropy, pulse-echo, simulation, ultrasonic attenuation

Highlights

- Experimental attenuation in cast iron samples was carried out.
- Scattering attenuation theoretical models do not apply to complex nodular cast-iron structures.
- Simulation models as a strategy to predict the experimental performance of cast iron.
- A k-Wave simplified simulation model is used to characterize complex structures.

0 INTRODUCTION

There are many applications for nodular cast iron due to its castability, high thermal conductivity, and good mechanical properties, specifically tensile strength and ductility. The mechanical properties of a metal greatly depend on the microstructure; in the case of nodular cast iron, which is produced by adding, shortly before solidification, a small amount (lower than 0.04 %) of substances such as magnesium or cerium are present. These substances give rise to the growth of nodular graphite, whose shape and distribution are of fundamental importance in the behaviour of the metal [1] to [3]. Thus, the non-destructive evaluation of such structures is very important for the identification of the nodularity and matrix phases.

Ultrasonic characterization offers great advantages when compared with destructive metallographic methods. The interaction of ultrasound waves with the material microstructure can be evaluated, measuring the acoustic parameters, including velocity and attenuation. Two ultrasound attenuation mechanisms are generally identified: absorption and scattering. Absorption is related to thermal conduction loss, hysteresis, and a viscous loss mechanism [4]. Scattering is due to heterogeneities such as grain boundaries, voids, inclusions, second-phase particles or porosity [5] to [9]. This attenuation

mechanism is commonly accepted as the most important in heterogeneous materials, such as the case of cast iron [10] to [13]. It is important to take into account the fact that the scattering effects of the matrix grains are too small when compared to the nodular scattering effects and can be ignored [6], [7], [14] to [16].

Several authors have extensively studied scattering attenuation. At the beginning of the last century, Rayleigh presented a scattering formula [17], later adapted by Mason and McSkimin [18] and [19]; the case of polycrystalline aluminium. Huntington [20] used a stochastic theory to explain the scattering effects in polycrystalline structures. Lifshitz and Parkhomovskii [21] proposed a theory that considers the mode conversion at the grain boundaries. Moreover, a great contribution was made by Papadakis [5], [6], [22] to [25] with several published works related to that topic. The author classified the scattering in three classical regimes depending on the relation between the grain size (D) and the wavelength (λ): (1) Rayleigh regime (for $\lambda \gg D$), where the attenuation is proportional to the fourth power of frequency; (2) stochastic regime (for $\lambda \approx D$), where the attenuation is proportional to the square of frequency; (3) and geometrical regime (for $\lambda \ll D$), where the attenuation is frequency independent. For each regime, the microstructure is assumed to be

composed of spherical grains of the same size filling the medium. As the Mason and McSkimin models [18] are very complex, due to multiple scattering, their theoretical treatment is a laborious task. That led other authors to restrict the scattering analysis to a limited frequency range [5], [24] to [27]. Later, Hirsekorn [28] and [29], Stanke and Kino [30], and Weaver [31] developed general solutions valid for all grain size to wavelength ratios and for cubic symmetry polycrystalline materials. However, other researchers demonstrated that those general solutions failed in more complex microstructures, such as the commercial aluminium alloys [32] and two-phase sintered powders [33]. More recent works deal with advances related to the interaction of ultrasound waves with polycrystalline materials. Yang et al. [34] studied the shape effect of elongated grains on attenuation. Arguelles and Turner [35] evaluated the errors resulting from neglecting the grain size distribution and using an average grain size. Rzyzy et al. [36] used a semi-analytical attenuation model that considers the grain morphology and incorporates an exact spatial two-point correlation function. The authors concluded that the grain shape has a strong effect on attenuation in the Rayleigh-stochastic transition region. Rokhlin et al. [37] presented a model that includes second-order multiple scattering, applicable to all frequency ranges, providing small relative errors on both longitudinal and transverse attenuations for low anisotropy. Sha [38] extended the second-order attenuation (SOA) model for elastic waves in texture-free materials to textured polycrystals with ellipsoidal grains of arbitrary crystal symmetry. The predicted attenuation results of this work agree well with the literature on a textured stainless steel polycrystal.

Numerical or grid-based methods are also powerful tools in the analysis of the propagation in scattering media. The improvement of computational resources makes possible the implementation of those methods to study increasingly complex interactions [39]. Recent works used finite element methods (FEM) to accurately mimic wave propagation through a rather complex material [40] to [44]. Norouzian, et al. [45] use DREAM.3D tool to construct material volumes with lognormal grain-size distributions. The results show that the correlation between attenuation and distribution width can be modelled with a power law, and the frequency dependence of attenuation has been shown to be strongly depend on the distribution width.

The main advantage of these methods, when compared to analytical approaches, is related to the fact that they allow microscopic scale analysis instead

of an averaged effective medium. No assumptions have to be made concerning grain statistics, anisotropy degree, or multiple scattering events [36]. Limitations related to frequency range, model dimensions and grid discretization tend to be strongly reduced, in the future, with the increase of computational resources.

One technique widely established for the generation of polycrystalline material morphologies is the Voronoi tessellation [40] to [50] for 2-D or 3-D dimensions. Details about that approach can be found in [50]. The major drawback of elastic wave models based on low-order finite difference or finite element schemes is the large number of grid points per wavelength required to avoid numerical dispersion. The k-Wave toolbox for Matlab can be used as an alternative to those approaches. It uses a Fourier domain pseudospectral method for the simulation and reconstruction of photoacoustic wave fields in a faster way. Also, it uses less memory and is user friendly [51] and [52]. Fewer spatial and temporal grid points are needed for accurate simulations [51]. Recently many authors have used the k-Wave toolbox in applications related to attenuation in ultrasonic computed tomography [53], guided waves in layered structures [54] and [55] time domain, power law attenuation in breast and liver tissues [56], one-sided ultrasonic non-destructive evaluation [57], high intensity focused ultrasound [58], nonlinear ultrasound propagation in absorbing media [59], ultrasonic transducers field modelling [60], and 3-D ultrasound imaging [61].

In the present work, the authors propose a simulation model to evaluate the scattering attenuation in structures with complex scatterer distributions. It uses the experimental attenuation knowledge in nodular cast iron to infer the scatterer size and its power dependence, as well as the concentration that best mimics the structure behaviour. The k-Wave toolbox is used to construct the simulator. Additionally, the results provided by the theoretical and simulation models are presented and a discussion is also made about the feasibility of the Truell et al. [62] and Papadakis [22] models for attenuation evaluation in nodular cast iron.

1 ULTRASONIC SCATTERING ATTENUATION

1.1 Theory

For the Rayleigh domain, characterized by a wavelength much higher than the scatterer size, Ying and Truell [27] presented a model for the total scattering cross-section, assuming the scatterers are spherical, solid, elastic, isotropic, have uniform size

and are embedded in a solid elastic matrix. Based on those assumptions, Truell et al. [62] presented an expression for the ultrasonic scattering attenuation evaluation of longitudinal waves:

$$\alpha_{sT} = \frac{2\pi}{9} Ng \frac{\omega^4}{v_l^4} r^6, \quad (1)$$

where N is the number of scatterers per unit volume, ω is the angular frequency, v_l represents the longitudinal propagation velocity, r is the scatterer radius, and g denotes a factor related to the elastic properties of the medium. The original g expression in [27] can be changed to encompass the longitudinal (v_l) and transversal (v_t) velocities:

$$g = \left(\frac{3 \left(\frac{v_{l1}}{v_{t1}} \right)^2}{3 \left(\frac{v_{t2}}{v_{l2}} \right)^2 - 4} - 1 \right) + \frac{\mu_2}{\mu_1} + 4 + \frac{1}{3} \left(1 + 2 \left(\frac{v_{l1}}{v_{t1}} \right)^3 \right) \left(\frac{v_{l1}\mu_2}{v_{t2}\mu_1} - 1 \right)^2 + 40 \left(2 + 3 \left(\frac{v_{l1}}{v_{t1}} \right)^5 \right) \cdot \left(\frac{\frac{\mu_2}{\mu_1} - 1}{2 \left(3 \left(\frac{v_{l1}}{v_{t1}} \right)^2 + 2 \right) \frac{\mu_2}{\mu_1} + \left(9 \left(\frac{v_{l1}}{v_{t1}} \right)^2 - 4 \right)} \right)^2, \quad (2)$$

where μ is the shear modulus given by $v_t^2 \rho$; ρ is the density, and the indexes 1 and 2 are related to the surrounding medium and scatterer, respectively.

Based on a similar work developed by Bathia and Moore [63], Papadakis [6] and [22] introduced a correction factor for the attenuation considering the effect of the grain size distribution. The author also showed a way of obtaining the correction factor in solids using two-dimension micrographs. The scatterer volume correction factor (T) is:

$$T = \frac{4\pi}{3} \frac{\overline{r_n^6}}{\overline{r_n^3}}, \quad (3)$$

where $\overline{r_n^6}$ and $\overline{r_n^3}$ are the average scatterer radius values of the sixth and third power, respectively, obtained from micrograph samples. From Eq. (3), it is clear that the scattering effects of a single large

scatterer are much greater than the effect summation of many small scatterers having the same volume. Taking the correction factor into account, the scattering attenuation is given as [6]:

$$\alpha_{sp} = NT^2 \frac{4\pi^3 f^4}{v_l^4} \cdot \left(\left(\frac{1}{3} \frac{\Delta\rho^2}{\rho_1^2} + \frac{\Delta L^2}{L_1^2} - \frac{8}{3} \frac{\Delta L \Delta\mu}{L_1^2} + \frac{32}{15} \frac{\Delta\mu^2}{L_1^2} \right) + \left(\frac{L_1}{\mu_1} \right)^{3/2} \left(\frac{2}{3} \frac{\Delta\rho^2}{\rho_1^2} + \frac{8}{15} \left(\frac{L_1}{\mu_1} \right) \frac{\Delta\mu^2}{L_1^2} \right) \right), \quad (4)$$

where f is the frequency, L_1 and μ_1 are the longitudinal and the shear modulus of surrounding medium, respectively, and $\Delta\rho$, ΔL , $\Delta\mu$ are the differences in density, longitudinal modulus, and shear modulus between scatterers and the surrounding medium, respectively. Although Eq. (4) uses a different approach for the attenuation calculation, the results are similar to those provided by the Truell model for uniform scatterer sizes.

1.2 Experimental Attenuation

The attenuation is the result of acoustic wave interactions with the propagation medium. Experimentally, the attenuation can be calculated by collecting the front surface reflection and two back wall ultrasound pulses from samples with parallel faces, as illustrated in Fig. 1. Following that approach, the authors studied different methodologies to calculate the ultrasonic attenuation in nodular cast iron samples [64], and concluded that Eq. (5) provided more reliable results, because it does not take into account the reflection coefficient between water and sample, which is frequency-dependent, as demonstrated experimentally [64]:

$$\alpha = \frac{1}{2L} \ln \left(\frac{A_1 A_s D_c}{A_1^2 + A_2 A_s} \right). \quad (5)$$

In Eq. (5), L is the sample thickness, A_1 and A_2 are the first and second back wall signals from samples, A_s is the reflected signal on the sample front face, and D_c is the diffraction correction coefficient for the sample path. The attenuation (α) unit is [Np/m], where Np is the symbol of neper, which is a logarithmic unit for ratios of measurements of physical quantities. The neper and dB are related by the following relationship: 1 Np = 8.686 dB

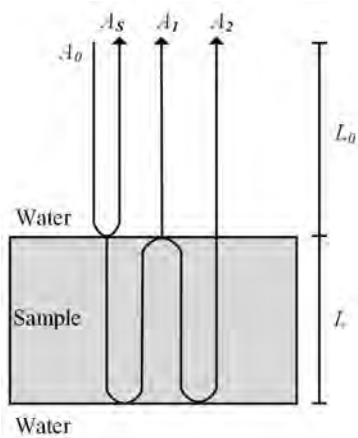


Fig. 1. Setup for attenuation measurement

1.3 k-Wave Simulation

The simulation of elastic wave propagation has many applications in ultrasonic non-destructive testing. The open-source k-Wave toolbox is an easy-to-use time-domain forward model based on a k-space pseudospectral time-domain solution to couple first-order acoustic equations for homogeneous and heterogeneous media in one, two, and three dimensions [51]. That tool makes possible the modelling of arbitrary sources, detecting surfaces with directional elements, and wave propagation that can account for nonlinearity, acoustic heterogeneities and power-law absorption. It is also possible to use optional input parameters to adjust the visualization and performance, including making wave propagation movies for presentations and running simulations on graphics processing units (GPU). The available geometry creation functions allow both Cartesian and grid-based geometries, such as circles, arcs, disks, spheres, shells or balls. As an example, Fig. 2 shows a three-dimensional model for a circular transducer (top) radiating into a sample with randomly distributed scatterers. The main drawback of k-Wave in ultrasonic non-destructive testing is related to the existence of large contrasts between media. Fourier-based methods such as k-Wave are effective when everything is smooth, and the contrasts are not too big (the code was originally designed for modelling ultrasound in biological tissue, where the contrasts are low). For large contrasts, oscillatory errors can be accumulated, because the steep edges cannot be represented very well by a small number of Fourier components. Some caution must be taken in such situations. Smoothing the density and propagation velocity or keeping the time step small will help to reduce that effect.

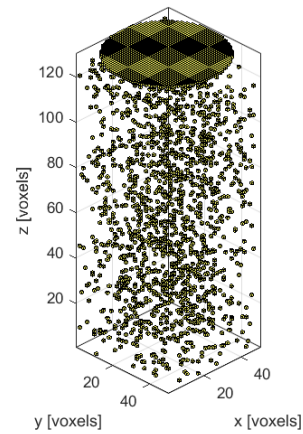


Fig. 2. Three-dimensional simulation model

2 RESULTS AND DISCUSSION

2.1 Samples Granulometry Analysis

Six machined cast iron samples with parallel surfaces within 15 μm and 13.1 ± 0.05 mm in thickness were used in this study (Fig. 3). The samples were prepared using standard metallographic methods. First, they were mechanically polished using metallographic carbon silicate sandpaper with decreasing granulometry (P180, P1000, and P2500) and a final polishing by a 3- μm diamond suspension. The microstructure of the processed zones was examined using optical microscopy, with a 200 \times magnification.

The typical microstructure of such samples is shown in Fig. 4, where the precipitated graphite nodules (in black) embedded in a ferrite/perlite matrix are easily identified, as are the grain boundaries. It was observed in all samples that the nodules present non-uniform shapes and have a large wide range of sizes that can vary from less than one micrometre to several tens of micrometres (the larger measured radius was 35 μm).

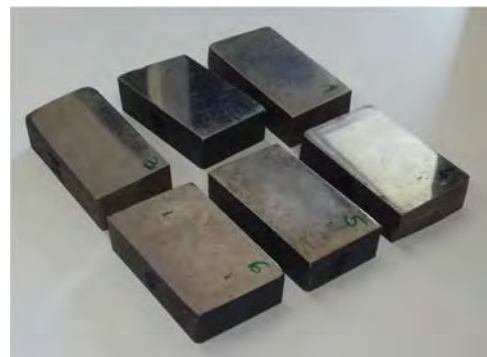


Fig. 3. Cast iron samples

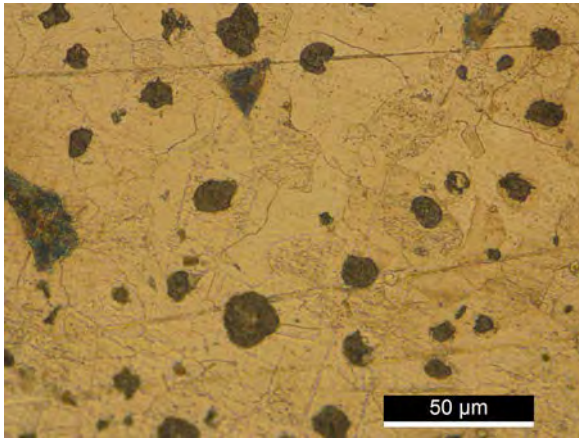


Fig. 4. Optical micrograph of a nodular cast iron sample

Ten micrograph images were taken from each sample, corresponding to an analysis section of 10.32 mm². The ImageJ free package software was used to determine the nodules' size [65], which were considered to be spherical. The nodule size distribution is presented in Fig. 5. Most of the nodules have a radius less than 10 μm. The spatial variability of the nodules' size in different regions of each sample is low. The standard deviation of the nodule size divided by its average value obtained for all images of each sample was about 8.9 %. Also, for each sample, the total number of nodules was used to obtain the lognormal distribution parameters. The mean and standard deviation are presented in Table 1.

2.2 Experimental Attenuation Evaluation Setup and Results

The experimental immersion setup for the attenuation evaluation is presented in Fig. 6. A pulser/receiver is used to excite a broadband 8 MHz central frequency transducer and to collect, amplify, and filter the reflected signals. Then, the acquired signals are displayed in an oscilloscope and transferred to a

computer for further processing. The transducer is moved using a computer-controlled micro-positioning system.

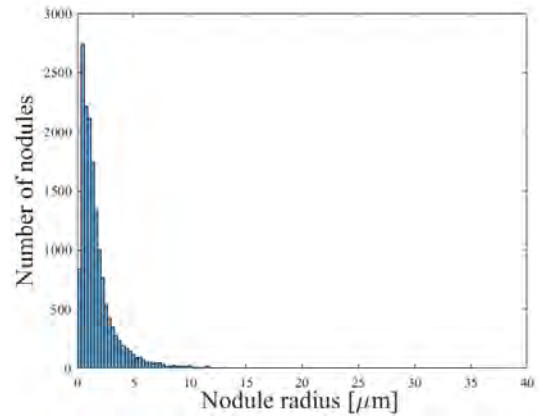


Fig. 5. Samples' nodule size distribution



Fig. 6. Experimental setup for attenuation evaluation

All samples characterized in section 2.1 were used in the study. Four acquisitions were accomplished in different regions of each sample to take into account their structural variation. Then, using Eq. (5) an average attenuation of 11.96 Np/m was obtained,

Table 1. Lognormal fitting parameters for each sample

Sample	1	2	3	4	5	6
Mean [μm]	0.2335	0.1178	0.4310	0.1007	0.0671	0.2420
Standard deviation [μm]	0.9492	0.8026	0.7862	0.8751	0.8545	0.9161

Table 2. Acoustic properties of cast iron components

	Longitudinal velocity [m/s]	Transversal velocity [m/s]	Longitudinal modulus [GPa]	Shear modulus [GPa]	Density [kg/m ³]
Pearlite/Ferrite	5830	3090	269	75.6	7920
Graphite	4210	2030	38	8.9	2170

with a standard deviation (SD) of 1.22 Np/m and root mean square error (RMSE) of 1.10 Np/m for the 24 measurements.

2.3 Attenuation Evaluation Using Analytical Models

In order to evaluate the scattering attenuation in the cast iron samples, important parameters as longitudinal and shear velocities, longitudinal and shear moduli, and density related to the matrix (pearlite/ferrite) and scatterers (graphite), must be known. They are represented in Table 2 [24].

The scatterer concentration knowledge is also required to estimate the attenuation. Thus, the two-dimensional scatterer concentration is obtained relating the total amount of nodules (n_T) with the area of micrograph (A_T), as follows:

$$N_A = \frac{n_T}{A_T}. \tag{6}$$

Then, the scatterer concentration (N) can be calculated using the following equation [66]:

$$N = \left(\frac{\pi}{6f_g} \right)^{\frac{1}{2}} (\alpha_N N_A)^{\frac{3}{2}}, \tag{7}$$

where f_g is the graphite nodule fraction that is about 3.33 % for the analysed images, and $\alpha_N = 1.25$, which is a parameter related to the width of the nodule size distribution.

To infer the real scatterer effect in the attenuation, the scatterers that contribute to the attenuation should be determined. To do so both the Truell and Papadakis models will be used and their performance analysed.

Due to the wide nodular distribution observed in the samples, the Truell model given by Eq. (1) cannot be used to calculate the scattering attenuation, because it uses the same size scatterers. Thus, in order to encompass all scatterer sizes found in the samples, the authors present a new equation for the attenuation, based on Eq. (1), which takes into account the summation of the different classes of scattering nodules:

$$\alpha_{sTm} = \sum_{i=1}^n \frac{2\pi}{9} g \frac{\omega^4}{v_l^4} N_i r_i^6. \tag{8}$$

In Eq. (8), n is the total number of scatterer classes, r_i and N_i are the average radius and scatterer concentration of each class, respectively. Thirty-five scatterer classes with 1 μm step were considered for the attenuation calculation. The scatterer concentration of each class is a fraction of the one obtained by Eq.

(7). From Eq. (8), resulted an attenuation of 2.67 Np/m.

The different attenuation values provided by the Truell model and experimental approach could be due to the fact that the Truell model makes use of spherical, solid, elastic, and isotropic scatterers. In addition, the referred model assumes that there is a sharp variation of the properties between the scatterers and the surrounding media. However, the nodular cast iron structure is rather more complex, especially its matrix, which is usually formed by ferrite around the nodules, with a homogeneous structure, and by pearlite in other regions with a lath-type structure with some degree of heterogeneity [24]. Therefore, the boundary effect between scatterers and matrix is certainly more complex than the one presented in the model.

The Papadakis model takes into account the scatterer size distribution, which is an improved version of the Truell theory. Generally, in practical materials, such as cast iron, the scatterers are not all the same size and follow a distribution that can be evaluated. The correction factor (see Eq. (3)), requires the knowledge of $\overline{r_n^6}$ and $\overline{r_n^3}$:

$$\overline{r_n^6} = \sum_{i=1}^n \frac{r_i^6}{n}, \tag{9}$$

$$\overline{r_n^3} = \sum_{i=1}^n \frac{r_i^3}{n}. \tag{10}$$

Using the experimental scatterer distribution (see Fig. 5), resulted $\overline{r_n^6} = 3.5 \times 10^5 \mu\text{m}^6$ and $\overline{r_n^3} = 54.73 \mu\text{m}^3$ for the whole sample area analysed, and $T = 2.68 \times 10^{-14} \text{m}^3$.

Then, using Eq. (4), where the right side is essentially composed by the physical parameters of matrix and scatterers presented in Table 1, resulted $\alpha_{sP} = 370 \text{ Np/m}$, which is very large compared to the experimental attenuation. In addition to the explanation made for the Truell model deviation in relation to the experimental attenuation, which also applies to the Papadakis model, other important factors can contribute to the observed discrepancy between theory and experiment. Perhaps, the most important one deals with the scatterer distribution obtained from the micrograph images. In the Papadakis model, all scatterer sizes and shapes were taken into account for the attenuation calculation that originates a very high scatterer concentration value. However, the authors concluded the attenuation decreases sharply if the lower size scatterers are discarded, as illustrated in Fig. 7. That tendency is expected because the

scatterer concentration (N) falls more quickly than the correction factor (T). For instance, if all scatterers lower than $7 \mu\text{m}$ are discarded, the attenuation is 15 Np/m , which is very close to the experimental one.

Finally, the Papadakis model also presents limitations in predicting the scattering attenuation in the cast iron samples.

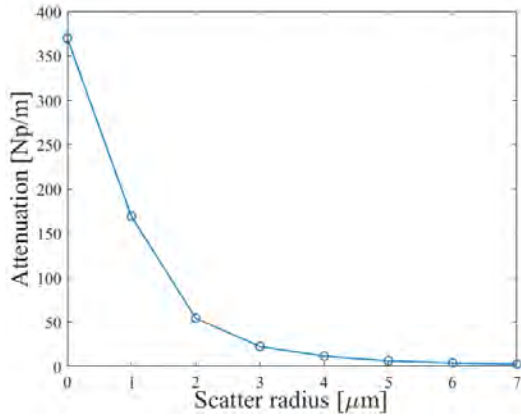


Fig. 7. Attenuation behaviour versus scatterer size

2.4 Simulation Model

The simulation of the cast iron nodular structure shown in Fig. 4 is difficult due to the non-uniformity of scatterers in size and shape. Here, the authors propose a simplified model assuming all scatterers with the same size and known concentration that leads to the experimental attenuation value. For that purpose, it is assumed that the attenuation is directly proportional to the scatterer concentration (see Eq. (11)), which is in accordance with the theoretical models [21] and [60] and is a common-sense assumption. The equivalent scatterer size (r_{eq}) and its exponent dependence (n) will be determined based on the inverse problem, using the experimental attenuation. The simplified attenuation expression based on the previous assumptions is as follows,

$$\alpha = CN(r_{eq})^n, \tag{11}$$

where C and n are constants to be determined. Eq. (11) follows the Rayleigh scattering model, and only the geometric parameters are of interest, such as the scatterer concentration, equivalent scatterer size, and its exponent dependence. The frequency dependence and the matrix and scatterer properties are included in the constant C , because the study related to the attenuation variation with these parameters is outside the scope of this work.

Using k-Wave, a simulation model with dimensions $x = 6 \text{ mm}$, $y = 6 \text{ mm}$ and $z = 40 \text{ mm}$, similar to the one illustrated in Fig. 2, was defined by a computational grid for simulation purposes. For the scattering attenuation evaluation, an ultrasonic probe is located at the bottom of the model ($z = 0$), working as a transmitter and another at $z = 30 \text{ mm}$ working as a receiver. Two signals are collected at the receiver: a reference signal a_0 (model without scatterers) and a signal a_s considering the model filled up with randomly distributed scatterers. The scattering attenuation is then given by:

$$\alpha_{sim} = \frac{1}{d} \ln \left(\frac{a_0}{a_s} \right), \tag{12}$$

where $d = 30 \text{ mm}$ is the distance between the probes. The two probes are 6 mm in diameter, like the one used in the experimental measurements (see section 2.2). The probe excitation was made by a tone burst, as shown in Fig. 8, to mimic the experimental transducer response illustrated in Fig. 9.

The first set of simulations were carried out for scatterers of $r_{eq} = 60 \mu\text{m}$, corresponding to a grid point spacing of $120 \mu\text{m}$ (cuboid edge). The scatterer concentration was varied from 0 mm^{-3} to 100 mm^{-3} with a step of 10 mm^{-3} , and ten signals a_s were collected for each concentration value. The corresponding attenuation was then calculated using Eq. (12), and the results are shown in Fig. 10. An average attenuation value for each concentration was considered. As expected, the attenuation increases as N also increases. That behaviour is more pronounced for lower concentrations. In the boxplot of the simulated data (Fig. 10) the central mark is the median, the edges of the box are the 25th and 75th percentiles, the whiskers extend to the most extreme data points (not considering the outliers, which are plotted individually as crosses).

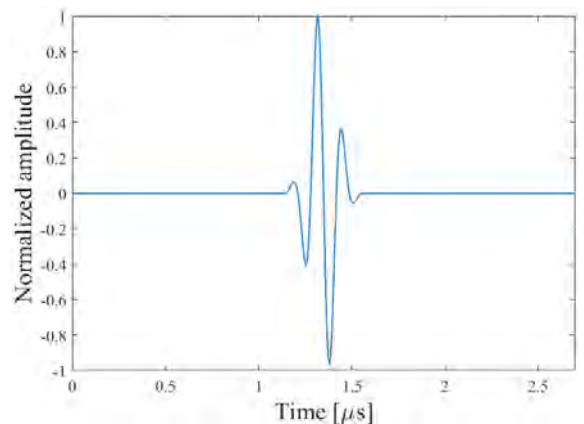


Fig. 8. Simulated tone burst

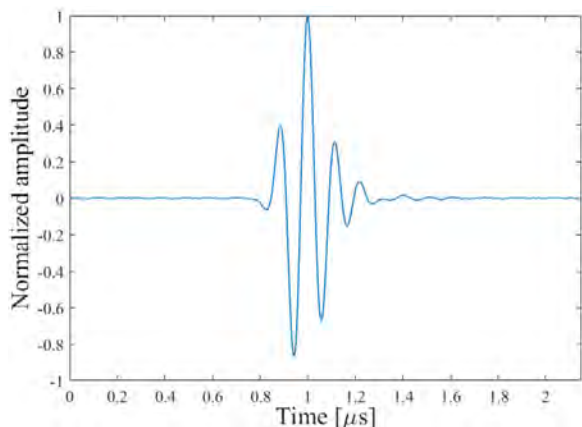


Fig. 9. Experimental transducer response

For the simulated scatterer size, the concentration value that leads to the experimental attenuation calculated in section 2.2 ($\alpha_{exp} = 11.96$ Np/m), is about 3.2 mm^{-3} . This result is more clearly observed in Fig. 11, which corresponds to an expansion of Fig. 10.

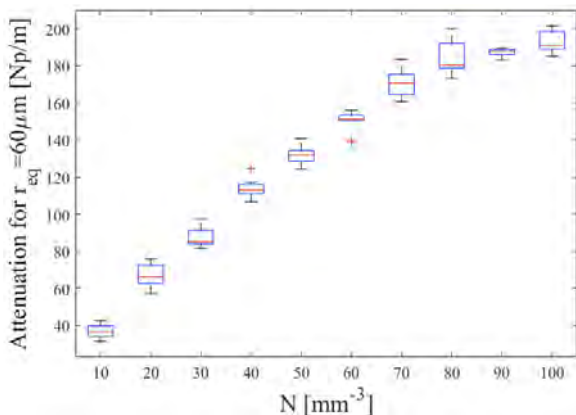


Fig. 10. Simulated attenuation for $r_{eq} = 60 \mu\text{m}$ versus scatterer concentration

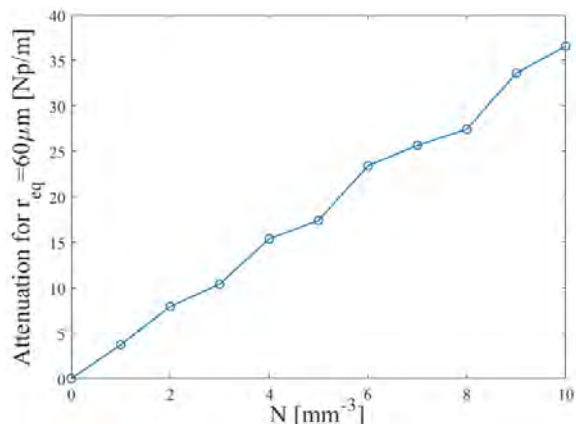


Fig. 11. Zoomed version of Fig. 10 for lower concentrations

In order to establish a correlation between the scatterer concentration (N) and size (r_{eq}), additional attenuation simulations were carried out using different scatterer sizes, resulting in the concentrations (presented in Table 2) that lead to the experimental attenuation. A power fitting applied to the values in Table 2 gave rise to Eq. (13), whose behaviour is illustrated in Fig. 12, where a remarkable goodness of fit with $R^2 = 0.99$ is observed:

$$N = 4.527 \times 10^{-13} (r_{eq})^{-5.173} \tag{13}$$

Table 2. Scatterers dimensions and concentration for $\alpha_{exp} = 11.96$ Np/m

$r_{eq} [\mu\text{m}]$	40	50	60	70	80
$N [\text{mm}^{-3}]$	25.5	8.0	3.2	1.4	0.68

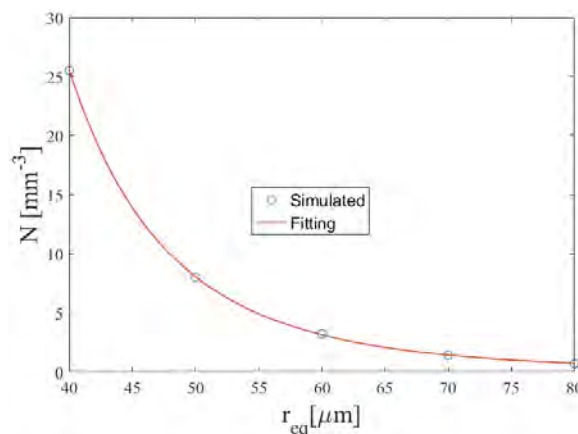


Fig. 12. Power fitting of scatterer concentration N as function of r_{eq}

Solving the proposed simplified model given by Eq. (11), in order to extract the scatterer concentration:

$$N = \frac{11.96}{C} (r_{eq})^{-n} \tag{14}$$

and comparing the Eqs. (13) and (14), results $C = 2.65 \times 10^{13}$ and $n = 5.173$, which allows writing Eq. (11) as:

$$\alpha = 2.65 \times 10^{13} N (r_{eq})^{5.173} \tag{15}$$

where the attenuation [Np/m] is a function of the fifth power of the scatterer size, which is not far from the trend of Eq. (1), as expected. It should be pointed out that Eq. (15) resulted from the studied cast iron samples illustrated in Fig. 4, whose attenuation was experimentally calculated by Eq. (5). Thus, based on the previous knowledge of experimental

attenuation, simulation models can be created varying the parameters N and r_{eq} , according to Eq. (14). This allows mimicking structures with complex scatterer distributions in terms of uniform scatterer concentration and size.

The presented simulation model considers the geometrical factors (size and concentration of scatterers) and can be used to mimic the microstructure of cast iron for a specific excitation frequency. The scatterer equivalent size can be defined a priori and the concentration calculated by Eq. (14) to establish the simulation model.

3 CONCLUSIONS

This work aimed to evaluate the ultrasonic scattering attenuation of structures with complex scatterer distributions. The idea was to predict the attenuation behaviour for such complex structures by developing a simulation model based on the experimental attenuation results. That was motivated by the limitations presented by the Truell and Papadakis theoretical models that provide ultrasonic scattering attenuation results quite inconsistently when compared with the experimental one. That conclusion was observed for nodular cast-iron structures, and the authors believe that the same results are expected for materials having inhomogeneous scatterer sizes and concentrations. The k-Wave simulation model developed in this work allows an easy and customized implementation, where the scatterer size and concentration can be varied to mimic the experimental ultrasonic scattering attenuation measured in inhomogeneous structures.

4 ACKNOWLEDGEMENTS

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Operating Performance of External Non-Involute Spur and Helical Gears: A Review

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In practical use, most gears have an involute shape of tooth flanks. However, external involute gears have some drawbacks, such as unfavourable kinematic conditions at the beginning and end of meshing, a limited minimum number of teeth, and the highly loaded convex-convex (i.e., non-conformal) contact. Researchers have developed and analysed various non-involute forms of tooth flanks, but they have not been widely accepted. The main reasons are higher manufacturing costs and sensitivity to manufacturing and assembly errors. Analyses of non-involute forms of teeth are mostly theoretical (analytical and numerical), while there is a lack of experimental confirmations of theoretical assumptions. This paper reviews external non-involute shapes, their operating characteristics and possibilities of use compared to involute gears. Established criteria, such as Hertzian pressure, oil film thickness, bending stress at the root of the tooth, contact temperature, and gear noise, were used for assessment. The results of analytical studies and experimental research on S-gears are presented in more detail. S-gears have a higher surface durability and a lower heat load when compared to involute gears. The usability of non-involute gears is increasing with the development of new technologies and materials. However, the advantages of non-involute shapes are not so significant that they could easily displace involute gears, which are cheaper to manufacture.

Keywords: non-involute gears, tooth profile, path of contact, Novikov gears, S-gears, gearing load capacity

Highlights

- The advantages and disadvantages of established involute gears are analysed.
- A systematic overview of non-involute spur and helical gears is presented.
- A comparison between involute and non-involute gears is performed based on established criteria.
- Results of research on S-gears are given in more detail.

0 INTRODUCTION

To ensure smooth transmission of rotation between two gears, the flanks of the teeth must be designed to comply with the law of gearing: the common normal of the contacting flanks of teeth must pass through the pitch point C (Fig. 1b) at each point of contact. Involute gears with involute-shaped flanks of teeth are predominantly used in industry. Since non-involute gears will be compared with involute ones, the main properties of involute gears and definitions of the basic concepts will be summarised in Chapter 1. Our overview is limited to external gears. Involute gears are described in more detail in gear-books [1] to [3].

The principal disadvantages of external involute gears are:

- limited minimum number of pinion teeth,
- convex-convex contact of the tooth flanks and therefore a higher contact load than in the concave-convex contact, which occurs in most non-involute gears,
- unfavourable kinematic conditions (high sliding speed) at the beginning and end of meshing.

Many researchers have attempted to eliminate these disadvantages by changing the shape of the tooth profile. Non-involute gears can be designed so

that the shape of the tooth flanks of the basic rack is defined first. The path of contact and the tooth flanks of pinion and wheel are defined using the known methods described in the literature [1]. In this way, the S-gears [4] and cosine gears [5] are defined. Another possibility is to define the shape of the tooth flanks based on a pre-selected path of contact. This procedure is used when defining a gearing with a parabolic path of contact [6].

In the follow-up, the theoretical criteria for assessing the load capacity of gears, their efficiency, and noise will be presented first. Next, an overview of non-involute tooth flanks and their comparison with involute teeth will be made. The results of theoretical analyses and tests from the cited literature will be given. The results of research on S-gears that were developed and tested at the Faculty of Mechanical Engineering, University of Ljubljana, will be presented in more detail [7] and [8]. The purpose of this paper is to give a systematic overview of non-involute gears and their advantages and disadvantages. This reference-backed review is useful for both researchers of non-involute gears and for users in practical applications. The geometry and properties of individual non-involute forms are given in Chapter 3. A summary of properties of different non-involute

gears compared to involute gears, areas of application and references are given in Tables 2 and 3 in Chapter 4.

1 INVOLUTE GEARS

In involute gears, the tooth flank profile has a shape of an involute, which is a curve drawn by a point on a line, which is rolled on the so-called base circle. The involute is mathematically defined by the involute function $\text{inv } \alpha$ (Fig. 1a). The curvature of the involute decreases when the radius of the base circle increases. If the radius of the basic circle is infinite, the involute is a straight line. The rack cutter for involute gears has a trapezoidal shape. Tools for manufacturing gears, therefore, have a simple straight-tooth shape, which is one of the most significant advantages of involute gears. The creation of the involute and the meshing of the teeth are shown in Fig. 1b, while the meshing of rack and pinion is shown in Fig. 2. The curve that connects the points of contact of the tooth flanks is called the path of contact. If the shape of the path of contact is defined, the shapes of the rack cutter profile and the flanks of teeth are also defined. In the case of involute gears, the path of contact is a line segment between the initial contact point A and the final contact point E. The extension of the path of contact runs tangentially to the base circle of the gears. The angle between the direction of the normal force to the tooth flank and the direction of the tangent through the pitch point defines the pressure angle α . In involute gearing, the direction of the force on the tooth does not change during meshing, which is favourable with regard to dynamic forces. Involute gears are not sensitive to centre distance errors. If the centre distance in the gearbox is greater than the theoretical one, the backlash increases, the pressure angle increases slightly, while kinematic conditions in the contact of the tooth flanks do not change significantly.

The operating properties of involute gears can be influenced by profile shift (Fig. 2). In a gear without a profile shift, the datum line of the rack runs tangentially to the pitch circle. The profile shift V can be negative or positive displacement of the tooth profile with respect to the axis of rotation of the gear. An undercut appears in an involute gear with a small number of teeth. The undercut limit depends on the helix angle, pressure angle, and height of the flat part of the rack. In the case of spur gears with a pressure angle of 20° and a rack as in Fig. 2, the undercut limit is at 17 teeth. To make a functional gear with a smaller number of teeth, a positive profile shift is needed. The choice of profile shifts is limited by the sharpness of

teeth, so the number of teeth of an involute pinion has its lowest limit.

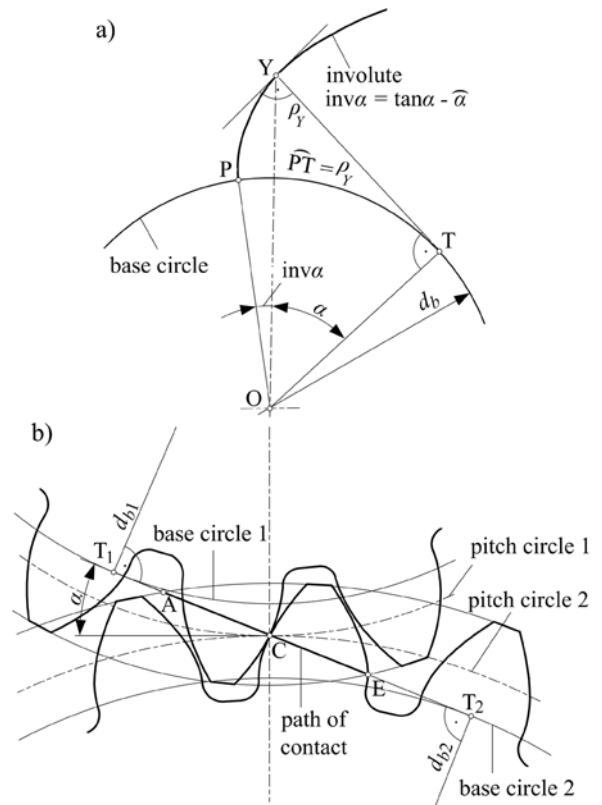


Fig. 1. Involute gears; a) the formation of an involute, and b) meshing of teeth

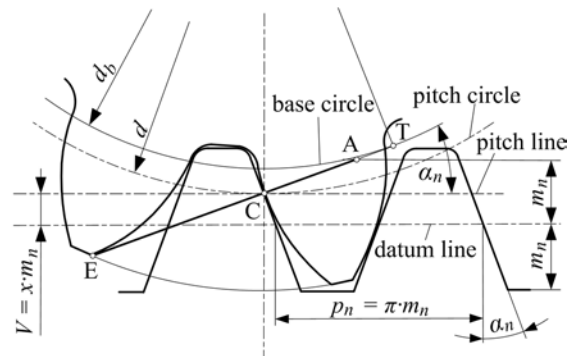


Fig. 2. Profile shift at involute gears

The sum of profile shifts is essential for the operating properties and the centre distance of the gear pair. When the sum of profile shifts is positive, the centre distance, pressure angle and the root load capacity of gears increase, while the contact ratio of gears decreases and the noise of gears increases [1] and [2]. The load capacity of involute gears can be increased, and noise and vibrations can be reduced by a tip relief profile modification [9] to [11] and crowning

[12] to [14]. Special software (e.g., KISSsoft) enables the optimization of operating properties of involute gears.

A special design of involute gearing is asymmetric involute gearing. Its geometry and gear design are discussed by Kapelevich [15]. The gearing is not sensitive to centre-distance errors. The rack cutter is not symmetrical and not standardized. The pressure angle at the loading side is greater than on the unloading side, so the asymmetrical gearing has a higher tooth surface durability and higher root load capacity than the symmetrical gearing [16] to [18]. Sekar [19] discusses the friction between the tooth flanks as a function of geometric parameters and load. The author proves that power losses are smaller than with classical involute gears.

2 CRITERIA FOR ASSESSING THE OPERATING PROPERTIES OF GEARS

2.1 Surface Durability of Gears

During the expected lifespan of gears, there should be no tooth fractures or damages to the flanks of teeth. Flank damages include pitting, scuffing, micro pitting and abrasive wear [20]. Pitting is damage made due to fatigue of the tooth flank material. It starts with a crack that can be on the surface or underneath. Because of the cyclic load, the crack widens until a chip is formed, which is flushed out by the oil. Due to the pits formed on the surface of the flanks, the bearing surface of the flanks is reduced, and the contact load on the remaining surface increases. The criterion for the occurrence of pitting is the contact load, defined by the Hertz equation [2]. Evaluation of involute gears according to the ISO 6336-2 standard [21] is based on the following form of the Hertz equation:

$$\sigma_H = \sqrt{\frac{F_n}{l_d \cdot \rho \cdot \pi \cdot \left(\frac{1-\nu_1^2}{E_1} + \frac{1-\nu_2^2}{E_2} \right)}} \quad (1)$$

At each point of contact, the line contact of teeth can be simulated with two cylinders with radii ρ_1 and ρ_2 or with a cylinder and a ring if one of the radii is negative and the contact is concave-convex. The relative radius of curvature is:

$$\rho = \frac{\rho_1 \cdot \rho_2}{\rho_1 + \rho_2} \quad (2)$$

The Hertzian pressure between two cylindrical surfaces depends on the normal force F_n , material properties (modulus of elasticity E , Poisson number ν), contact length l_d and relative radius of curvature.

The relative radius of curvature depends on the shape of the tooth flanks. When designing non-involute gears, the trend is to achieve the largest possible relative radius of curvature, which consequently means a reduction of the Hertzian pressure. In the case of S-gears [4], the relative radius of curvature in the area at the beginning and end of meshing is significantly larger than in involute gears due to the concave-convex contact. However, in the vicinity of the pitch point C, the convex-convex contact also exists in S-gears and, therefore, the relative radius of curvature is approximately the same as in involute gears. It depends on the pressure angle at point C. The same is true for cosine gears [5] and gears with a parabolic path of contact [6]. An analytical method for calculating the curvature radii for gears with a curved path of contact was developed in [7].

For the test conditions, materials, and geometry of the test gears, Hertzian pressures are defined at which pitting occurs after a certain number of tooth flank meshes. Values for involute gear materials are listed in the ISO 6336-5 standard [22]. When evaluating involute gears, the differences between the test conditions (lubricant, speed and the geometry of the test gears) and the actual conditions and geometry are considered by applying additional coefficients [21].

Due to the friction between the tooth flanks, the oil temperature increases, but the oil viscosity and the thickness of the oil film decrease. When the oil temperature exceeds the allowable value, the oil film breaks. When the protective layer is crushed, the two surfaces weld. Due to the sliding of the flanks, the joint collapses, and a gap is formed in the sliding direction. This type of damage is called "scuffing of gears". The evaluation of involute gears on scuffing is based on temperature criteria. Two methods are available in ISO/TS technical specification: the flash temperature method [23] and the integral temperature method [24]. The calculation of the flash temperature is based on Blok's formula [25] and [26].

$$\vartheta_{fla} = 0.62 \cdot \mu \cdot (w_n)^{0.75} \cdot \left(\frac{E'}{\rho} \right)^{0.25} \cdot \frac{|v_{t1} - v_{t2}|}{\left(\sqrt{B_{M1} \cdot v_{t1}} + \sqrt{B_{M2} \cdot v_{t2}} \right)} \quad (3)$$

Heat generation is influenced by the coefficient of friction μ , load ($w_n = F_n/l_d$), relative radius of curvature and sliding speed $v_g = v_{t1} - v_{t2}$. The denominator of the last term in Eq. (3) takes into account the heat dissipation from the contact. The substitute elasticity

modulus E' depends on the elasticity modulus of both gears and their Poisson numbers.

$$\frac{1}{E'} = \frac{1}{2} \cdot \left(\frac{1-\nu_1^2}{E_1} + \frac{1-\nu_2^2}{E_2} \right). \quad (4)$$

The frequently cited equation [1] and [27] is used for calculating the coefficient of friction.

$$\mu = 0.045 \cdot \left(\frac{w_n}{(v_{r1} + v_{r2}) \cdot \rho} \right)^{0.2} \cdot \eta_0^{-0.05} \cdot X_R. \quad (5)$$

The coefficient of friction depends on load, relative radius of curvature, sum of tangential velocities $v_{r1} + v_{r2}$, viscosity of oil η_0 and surface roughness (taken into account by the X_R coefficient). Non-involute gears with a curved path of contact, such as S-gears [4] and [7], cosine gears [5], and gears with a parabolic path of contact [6], have larger relative radii of curvature and lower sliding speeds at points that are critical for scuffing (at the beginning of meshing). The thermal load is lower at these points, so they are less susceptible to scuffing than involute gears are. Xue et al. [28] combined the dynamic load and transient thermal elastohydrodynamic lubrication (TEHL) to study the scuffing load capacity of spur gears. The hydrodynamic pressure, oil film thickness and flash temperature were calculated with a numerical method. The comparison between the TEHL and Blok's theory was implemented.

The minimum thickness of the oil film at the contact point is crucial for the occurrence of micropitting and abrasive wear of the tooth flanks. In general, the thickness of the oil film is calculated by solving the Reynolds differential equation. This procedure was used by Xu et al. [29] in the case of harmonic gear transmission. Typically, the Dawson-Hamrock equation [30] is used for the calculation of the minimum oil-film thickness. The general notation of the equation applies to the elliptical contact.

$$\frac{h_0}{\rho} = 3.63 \cdot \left(\frac{v_{r1} + v_{r2}}{2} \cdot \frac{\eta_0}{E' \cdot \rho} \right)^{0.68} \cdot (\alpha \cdot E')^{0.49} \cdot \left(\frac{F_n}{E' \cdot \rho} \right)^{-0.073} \cdot (1 - 0.61e^{-0.68k}). \quad (6)$$

Coefficient $k=a/b$ is the ellipticity parameter where a and b stand for ellipse axes. If the contact of the tooth flanks is a straight line, the expression in the last bracket of Eq. (6) is equal to 1. The equation can then be rearranged to the following form.

$$h_0 = \left(3.63 \cdot \eta_0^{0.68} \cdot \alpha^{0.49} \cdot E'^{-0.117} \right) \cdot \left(\frac{v_{r1} + v_{r2}}{2} \right)^{0.68} \cdot \rho^{0.466} \cdot F_n^{-0.073}. \quad (7)$$

Velocities, the relative radius of curvature, and load vary along the path of contact. Non-involute gears with a curved path of contact, such as S-gears [4] and [7], have a larger relative radius of curvature and a greater sum of tangential velocities at the beginning and end of meshing than the involute ones. The greater thickness of the oil film has a positive effect on both the damage of the flanks and the dynamic forces.

The evaluation of involute gears regarding micropitting is covered by the ISO/TS 6336-22 standard [31]. To assess the hazard of the micropitting, the relative thickness of the oil film is used, defined as:

$$\lambda_{GF,F} = \frac{h_y}{0.5 \cdot (R_{a1} + R_{a2})}, \quad (8)$$

where h_y is the local film thickness on the assumption of smooth surfaces, while R_{a1} and R_{a2} are the arithmetic mean roughness values. Bergstedt et al. [32] have demonstrated experimentally that the probability of micropitting is lower on a smoother surface. Clarke et al. [33] and Liu et al. [34] provided an overview of the results of the latest micropitting research.

2.2 Root Load Capacity of Gears

The bending stress at the root of a tooth is an indication of load; the stress depends on the tooth thickness and the tooth fillet at the root. When evaluating involute gears for root load capacity, the bending stress at the root of the tooth is compared with the root fatigue strength, which is determined by gear tests. Analytical evaluation of involute gears is defined in the ISO 6336-3 standard [35]. Numerical methods are predominantly used to calculate the stress at the root of the teeth of non-involute gears, and the finite element method (FEM) is the prevailing numerical method. Analytical methods (such as beam-like models for cycloid gearings) are rare. Numerical analyses are used in many papers dealing with non-involute gears [36] and [37].

2.3 Power Losses and Gear Efficiency

Friction between the flanks of teeth causes power losses. The friction power at the contact point i depends on the normal force F_{ni} , the coefficient of friction μ and the sliding speed $v_g = v_{r1} - v_{r2}$.

$$P_{fi} = F_{ni} \cdot \mu_i \cdot |v_{r1i} - v_{r2i}|. \quad (9)$$

In non-involute gears with a curved path of contact, such as S-gears [4] and [7], cosine gears [5], and gears with a parabolic path of contact [6], the

sliding speeds are lower at the beginning and end of meshing; therefore, the power losses are also lower. The friction work during the meshing of one pair of teeth between the initial contact point *A* and the final contact point *E* is:

$$W_{fr} = \sum_{i=1}^n P_{fri} \cdot \Delta t_i = \int_{t_A}^{t_E} P_{fr} \cdot dt. \quad (10)$$

With the known speed of the pinion n_1 in min^{-1} and the number of teeth of the pinion z_1 , the average friction power can be determined.

$$\bar{P}_{fr} = \frac{z_1 \cdot n_1 \cdot W_{fr}}{60}. \quad (11)$$

For involute gears, there is a theoretical method for determining power losses during the meshing of teeth. The procedure is described in the literature [1] and [2]. In the previous decade, an elastohydrodynamic lubrication (EHL) -based friction coefficient was developed and applied to the prediction of gear efficiency [38] and [39]. Ziegler et al. [40] compared the simulated load-dependent gear power losses of a transient thermal EHL model with the experimental results in the Forschungsstelle für Zahnrad und Getriebebau (FZG) gear test rig. Results show a very strong correlation when taking into account the mixed lubrication.

2.4 Vibration and Noise of Gears

The main geometric parameter that affects gear noise is the contact ratio. Mesh stiffness depends on the contact ratio, and it has a decisive influence on vibration and noise level [28]. The gearing is more silent and quieter if the contact ratio is higher. For helical gears, the total contact ratio is the sum of the transverse contact ratio and the face contact ratio.

$$\varepsilon = \varepsilon_\alpha + \varepsilon_\beta. \quad (12)$$

The transverse contact ratio is the ratio between the arc length of action and the angular pitch.

$$\varepsilon_\alpha = \frac{C_{1A} C_{1E}}{p_n} = \frac{C_{1A} C_{1E}}{m_n \cdot \pi}. \quad (13)$$

For involute gears (with points of contact on a line), it is defined as the ratio between the length ACE and the base pitch (Fig. 1). The distance between the initial contact point *A* and the pitch point *C* is greater in involute gears, so their transverse contact ratio is higher. Non-involute gears with a curved path of contact, such as S-gears [4] and [7], cosine gears [5], and gears with a parabolic path of contact [6], have a

shorter distance between the initial meshing point *A* and the pitch point *C*, so the contact ratio is lower than that of involute gears. The contact ratio of involute gears can be changed by making a profile shift. If the sum of profile shifts is negative, the contact ratio is higher than if the sum is equal to 0 or positive. The overlap ratio is higher if a gear is wider and if the helix angle is larger.

The force on the tooth acts perpendicularly to the tooth flank profile. Because the path of contact is a straight line at involute gears, the direction of the force does not change during meshing, which has a good effect on vibrations. Vibration and noise of gears are decisively affected by tooth deviations [41]. The noise of involute and non-involute gears can only be adequately compared if the gears have been made in the same accuracy class. Greater manufacturing accuracy significantly reduces vibration and noise levels.

3 NON-INVOLUTE GEARS

Various forms of non-involute gears are presented in this chapter. Operating properties are described for each gear type, supported by theoretical and experimental research. A comparison with comparable involute gears is given.

3.1 Cycloid Gearing

The flank of the cycloid gear tooth consists of a hypo- and an epicycloid [3]. A cycloid is a curve produced by tracing a path of a chosen point on the circumference of a circle that rolls around a fixed base circle. The shape of the addendum flank of the tooth is obtained by rolling the circle on the outer side of the base circle, while the shape of the dedendum flank of the tooth is obtained by rolling a circle inside the base circle (Fig. 3). During meshing, the hypocycloid and the epicycloid touch, the touch is concave-convex, and the path of contact has a shape of circular arcs. The rack has flanks curved in the shape of an orthocycloid, which is obtained by rolling a circle along a straight line. The shape of the hypocycloid depends on the ratio of diameters of the rolling circle and the base circle (k_c). The hypocycloid is a straight line if the value of $k_c=0.5$. The contact of the hypo- and the epicycloid is concave-convex when $k_c<0.5$, so the contact pressure is lower than that of involute gears. In these cases, the relative sliding is also lower than with comparable involute gears. The critical point of the gearing is the pitch point *C*, where a transition from the epicycloid to the hypocycloid occurs. In the

area of the pitch point, the relative curvature radii are very small, while the contact pressure is higher than in involute gears.

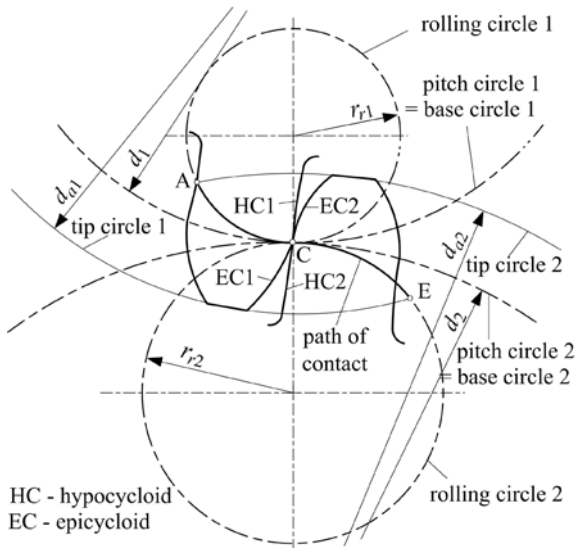


Fig. 3. Cycloid gears

A pinion can have a very small number of teeth (theoretically only two or three). Cycloid gears with two or three teeth are used for rotors in Roots blowers. Gearing is very sensitive to centre distance errors and does not allow a profile shift. If there is an error in the centre distance, vibrations increase. Manufacturing gears is a complicated and expensive process. This type of gears is used in precision mechanics (e.g., mechanical clocks).

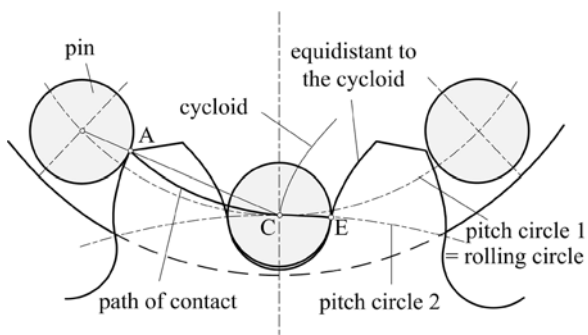


Fig. 4. Lantern gears

One special example of cycloid gearing is lantern gearing (Fig. 4) with pins on one of the gears. The flank of the gear tooth has the shape of an equidistant cycloid. The contact point on the pin slides along the flank of the gear tooth during meshing. The lantern gearing is only suitable for small circumferential speeds, e.g., drives with large driven gears in transport

devices. A special shape of the cycloid gear is installed in the rotary vector cycloidal-pin drive [42], used in robots. The main feature of such a gearbox is a large gear ratio.

Peng et al. [43] developed a new form of cycloid gearing named arc-tooth-trace cycloid gear (Fig. 5).

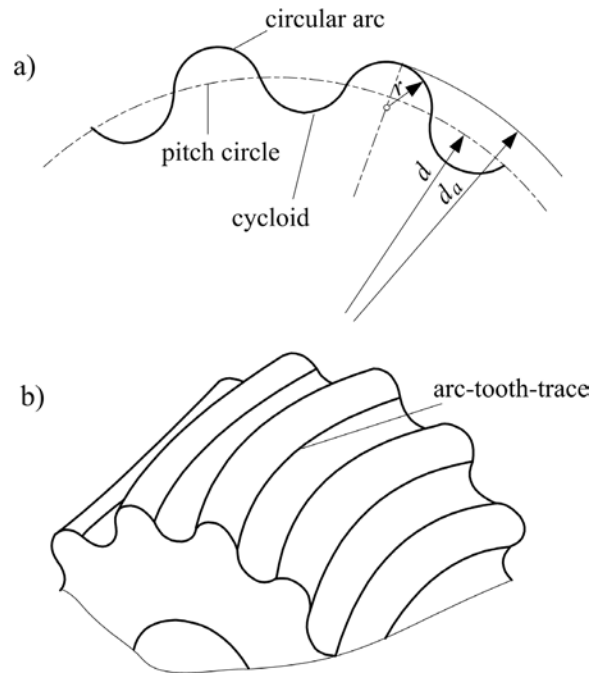


Fig. 5. Arc-tooth-trace cycloid gear; a) tooth profile, and b) arc-tooth-trace; adapted from [43]

The tooth trace of the gear is a segment of a circular arc. The profile of the tooth flank consists of two parts: above the reference circle is a circular arc segment, while a cycloid segment is below the circle. Equations for describing the geometry of the gearing and calculation of the contact ratio are derived in the paper. The transverse contact ratio is a constant value (0.5), while the overlap ratio depends on the module and the radius of a circular arc. 3D models of the new gears and comparable involute gears were made. The results of the numerical analysis show that the new gearing has 17 % higher tooth surface durability and 35 % higher root load capacity than involute gearing.

3.2 Helical gears with circular arc teeth

In 1926, Wildhaber [44] invented helical gearing with circular arc teeth (in a normal cross-section). Independently, a similar shape (a tooth profile with a circular arc in the front face) was invented 30 years later by Novikov [45]. Due to a similar shape, researchers in the past had named the gearing with the

sides of teeth in the form of circular arcs as Wildhaber-Novikov gearing, or shorter W-N gearing. Radzevich [46] and [47] clarifies differences between the two systems and explains why Wildhaber and Novikov gearing should be addressed separately. In his opinion, the name “W-N gearing” is incorrect. The geometry of the original Novikov gearing is shown in Fig. 6. This kind of gearing was applied in the final reduction stage of a Westland Lynx helicopter gearbox [48].

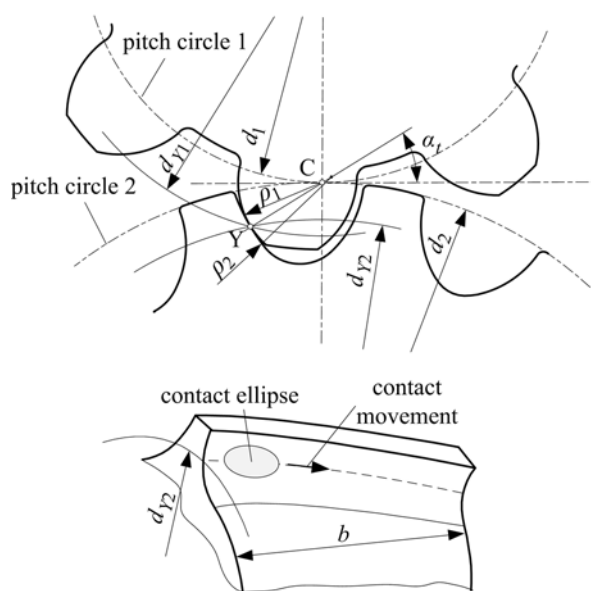


Fig. 6. Original Novikov gearing

The touch of the teeth is concave-convex. The contact surface has (in theory) an elliptical shape under load. When two gears rotate, the contact surface travels in the axial direction, parallel to the axis of the gear rotation. Transmission of rotation is continuous if the overlap ratio is higher than 1. To ensure the lateral backlash, the radius of the pinion's circular arc must be slightly larger than the circular arc radius of the gear wheel.

Dyson et al. [49] and [50] discuss the gearing geometry and kinematics in detail and present design instructions. Contact stress during meshing was analysed in more detail by Coulbourne [51]. Due to the concave-convex contact, contact stress is lower than in involute gearing. According to Niemman's research [52], W-N gearing has 1.5 to 3 times higher surface durability than comparable involute gearing. These values apply to a gear shape when $\rho_1 = \rho_2$. Differences decrease with an increasing helix angle. The influence of pressure angle and root fillet on the stress in the tooth root is discussed by Tsay [53]. His analysis was performed using FEM. Results of analyses indicate

that the W-N gearing has a slightly lower root load capacity than the involute gearing. The conditions for the formation of the oil film are better than in involute gearing [54].

Over the previous decade, Markovski and Batsch [55] have conducted numerical and experimental research on Novikov gearing. They incorporated errors that occur during manufacturing and assembly into their mathematical model. On the basis of numerical contact analysis, they analysed the impact of errors on the wear of tooth flanks. It turned out that the contact surface is not a real ellipse as it was theoretically assumed. They developed a method for measuring and displaying the contact surface of Novikov gearing [56]. They performed fatigue tests on Novikov gears ($m_n = 3$ mm, $z_1 = 30$, $z_2 = 47$, $b = 30$ mm) and comparable involute gears [57]. With the same torque on the pinion, the Novikov gears lasted three times longer till the first signs of pitting. This confirms that the surface durability of the Novikov gearing is about three times higher than the durability of involute gearing. They also measured vibration amplitudes on the housing. They were approximately five times higher than in the involute ones. The main reasons for this are slightly worse manufacturing accuracy and a small contact ratio. Batsch extended the research of Novikov gears to bevel gears [58]. A mathematical model of meshing is derived. Contact stresses and maximum bending stresses in Novikov gears and comparable spiral bevel gears are calculated numerically. In Novikov gears, the contact area is larger, so the calculated contact stresses are lower by approximately 22 %. Root stresses are higher by 27 % on the Novikov pinion.

Litvin et al. [59] developed a new shape of the W-N helical gearing. The rack cutter has a parabolic shape instead of a circular shape (Fig. 7). The contact ellipse is longer and narrower than in the basic Novikov gearing. Advantages of the new W-N gearing compared to the basic Novikov gearing are lower noise and vibration, lower sensitivity to manufacturing and assembly errors, the possibility of grinding the gears, lower contact pressure and lower bending stress. The advantages were proven by a simulation of meshing and by numerical calculations of contact stresses and bending stresses.

W-N gearing can also be made with two meshing zones (Fig. 8a) [60]. Compared to gearing with a single-zone meshing, this one has lower meshing stiffness. The jerks at the beginning of the meshing are smaller, and the distribution of force on the teeth is more favourable. A rack cutter with two circular arcs for manufacturing such gearing is shown in Fig. 8a.

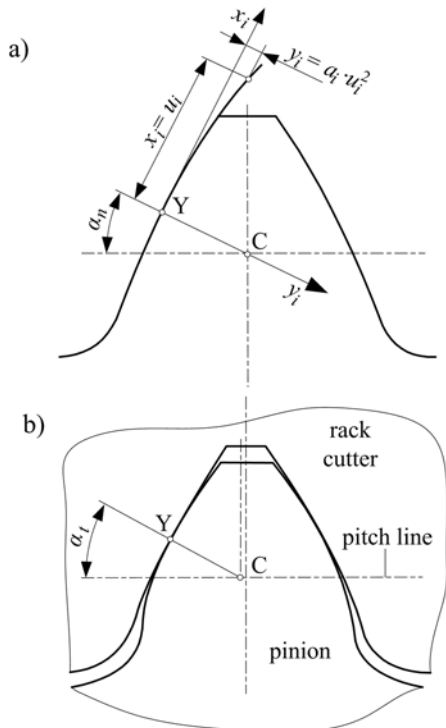


Fig. 7. New version of the W-N gear; a) rack-cutter in normal section, and b) transverse profiles; adapted from [59]

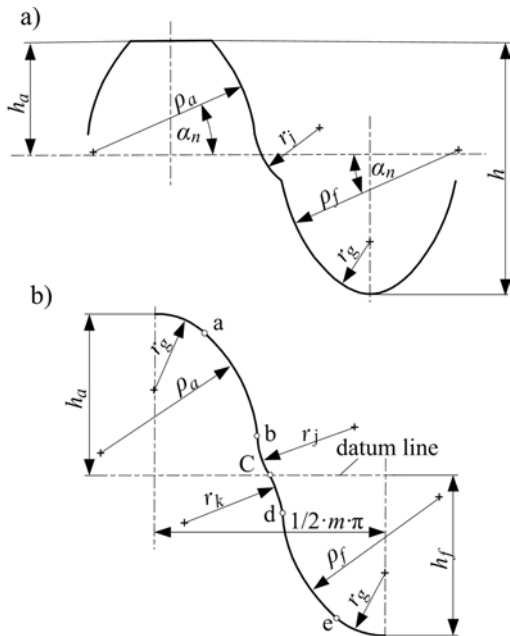


Fig. 8. Profiles of rack-cutter; a) W-N gears with two zones of meshing, and b) stepped triple circular-arc gears; adapted from [61]

The generated tooth flank has a singular point due to the sharp transition between the radii. To eliminate it, Yang [61] proposed a rack cutter with three circular

arcs, shown in Fig. 8b. He named it a stepped triple circular-arc gearing. He developed a mathematical model of the gearing and performed a stress analysis using FEM.

Ariga and Nagata [62] also suggested an improvement of Wildhaber-Novikov gearing. The top of the new tooth profile has the shape of a circular arc, and the dedendum consists of two involutes. Tests have shown that these gears are less sensitive to centre-distance errors than conventional W-N gears.

The gearing with a gear profile composed of circular arcs was also developed by Hlebanja and Hlebanja [63] (Fig. 9), called uniform power transmission gears (UPTG). A theoretical comparison with involute gearing is made. To confirm its functionality, a model of aluminium gears was made.

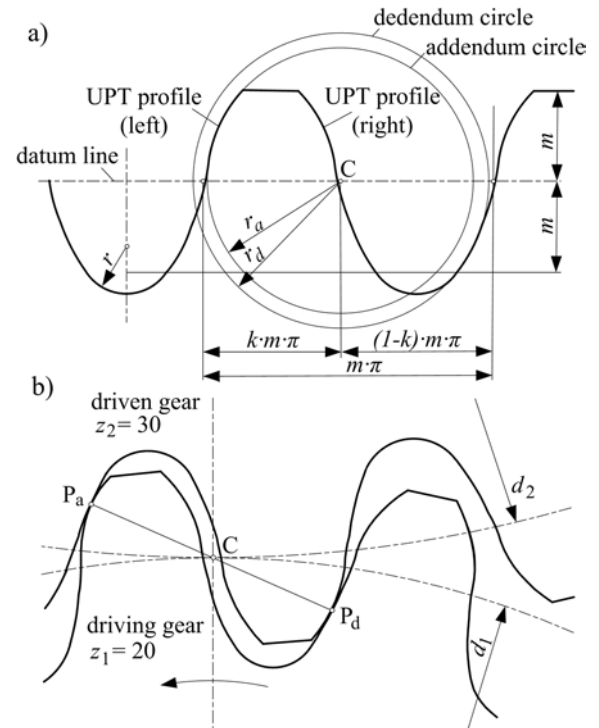


Fig. 9. UPT gears [63]; a) shape of the rack-cutter, and b) pair of UPT

Liang et al. [64] proposed gear transmission with a double circular arc-involute tooth profile. Three components are important in the generation of a tooth profile: the upper convex circular arc, the middle involute and the lower convex circular arc. Based on the equations of tooth surfaces, the software for calculating the data points of tooth surfaces was developed. The results were exported to the 3D drawing software. A numerical comparison of contact stresses and sliding coefficients between the involute

gearing and the new gearing was made; both are lower in the new gearing. Proposed gears and involute gears were also compared experimentally. The transmission efficiency under different operating conditions is presented. The measured efficiency of the new gear was lower due to the lower manufacturing quality.

3.3 Pure Rolling Helical Gears

Chen et al. [65] developed novel circular arc helical gears (CAHGM) with pure rolling. Based on the meshing principle of space curve meshing, the parameter equations of the concave-convex circular arc profiles were established. A pair of CAHGM was manufactured via rapid prototyping technology. The contact ratio equation and the theoretical kinematic performance were validated. The gears have a large backlash which is harmful to the forward and reverse drives. Chen et al. [66] and [67] also developed pure rolling helical gears with convex-to-convex meshing type and pure rolling rack and pinion mechanism. Geometric design, meshing performance, and mechanical behaviour were presented. Parametric equations for contact curves and for the tooth surfaces were derived. A numerical comparison of contact stresses and bending stresses between the modified involute gearing and the new gearing was made. With new gearing, the maximum bending stresses are significantly lower, and the contact stresses are 1.5 to 2 times higher. The maximum contact stress can be decreased by increasing the tooth number and face width and decreasing the helix angle. Properties of pure rolling gearing have not yet been experimentally researched.

3.4 Convoloid Gearing

Berlinger and Coulbourne [68] developed a new type of gearing, named convoloid gearing. Its tooth's dedendum has a concave shape, while its addendum has a convex shape (Fig. 10). The transverse contact ratio is between 1.1 and 1.3, while the overlap ratio is an integer. A theoretical and experimental comparison with equivalent involute gears is made. An important geometric parameter is the ratio between the width of the gear b and centre distance a . If $b/a < 0.2$, the convoloid gearing has no advantages. The load capacity ratio between convoloid and involute gears increases with increasing b/a ratio. Test results show that 20 % to 35 % higher load capacity can be achieved with convoloid gears.

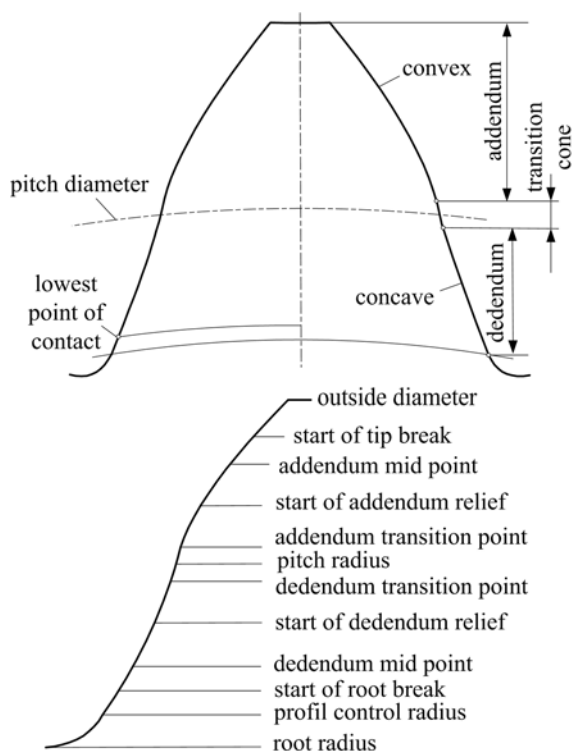


Fig. 10. Convoloid gearing; adapted from [68]

3.5 Gears with Curved Paths of Contact

3.5.1 S-Gears

Development of S-gears (invented by Hlebanja [69] and developed by Hlebanja et al. [70]) began in the 1970s. This type of gear was used in a rolling mill with a power of 1500 kW [71]. A path of contact in the shape of the letter S is a characteristic of this gearing. The first gearing had a small pressure angle at the pitch point C compared to today's shape. In the 1990s, the basic rack tooth profile equation was defined [72].

$$y = a_p \cdot (1 - (1 - x)^n). \quad (14)$$

The following parameters were selected in [7]: $a_p = 1.3$ and $n = 1.9$. The shape of the tooth profile, the path of contact, and both tooth flanks are shown in Fig. 11. A tool for shaping S-gears with a module of 4.575 mm was made. Research studies of load capacity of spur S-gears made of steel and comparable involute spur gears were performed. Involute gears had a module $m_n = 4.5$ mm, width $b = 20$ mm, number of teeth $z_1 = 16$ and $z_2 = 24$, and the profile shift coefficients $x_1 = 0.233$ and $x_2 = 0.12$. Before the tests, the geometry of all gears was measured on a coordinate-measuring machine. S-gears were

manufactured in accuracy classes 7 and 8, and involute gears in class 6 (according to DIN 3962 [73]). The roughness of the tooth flanks was approximately the same in both types of gears.

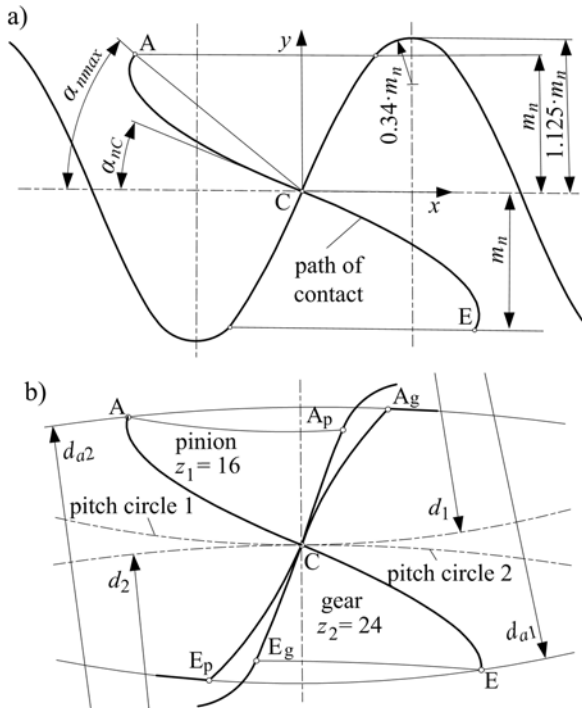


Fig. 11. S-gears [7]; a) basic rack tooth profile, and b) tooth flanks of S-gears

The load capacity tests were made on a standard FZG test rig [74] with a centre distance of 91.5 mm. Theoretical calculations of Hertzian pressures, flash temperatures and oil film thicknesses were made for the geometry of the test gears and the test conditions. Figs. 12 and 13 show a comparison of Hertzian pressures and flash temperatures along the path of contact. Theories about the load capacity of S-gears were defined on the basis of the criteria described in Chapter 2.

The tests showed a significantly higher resistance of S-gears to scuffing, which is the expected result on the basis of the calculation of contact temperatures and oil film thickness at the beginning of meshing. In fatigue tests, pitting occurred in the pitch point area of S-gears, and in the dedendum of involute gears. S-gears withstood a slightly higher number of load cycles than involute ones (by approx. 20 %) [7] and [75]. The wear of S-gears was about one half lower during the same operating time due to a thicker oil film.

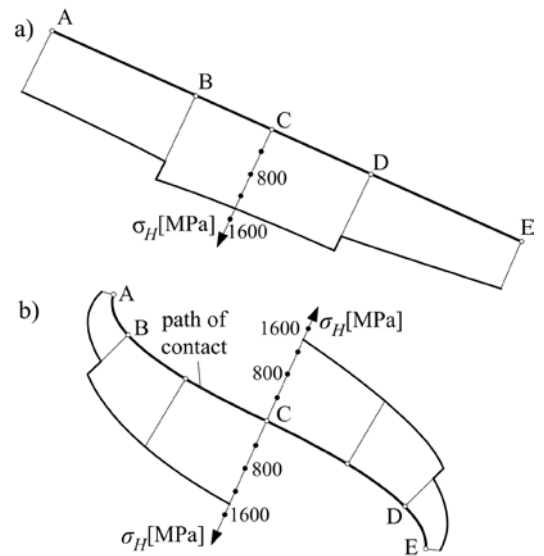


Fig. 12. Hertzian pressures during tests; a) involute gears, and b) S-Gears

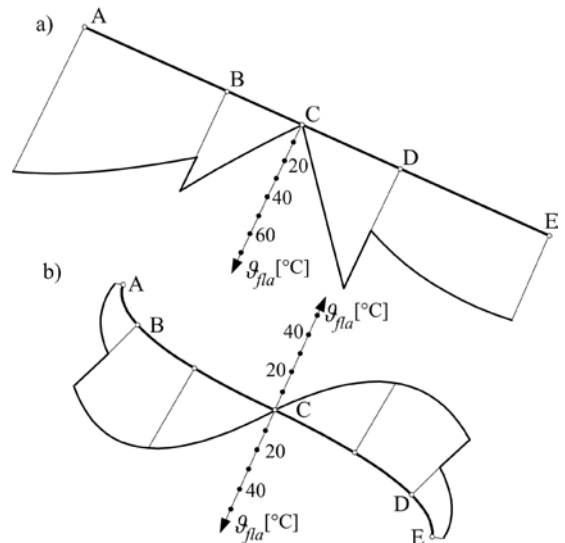


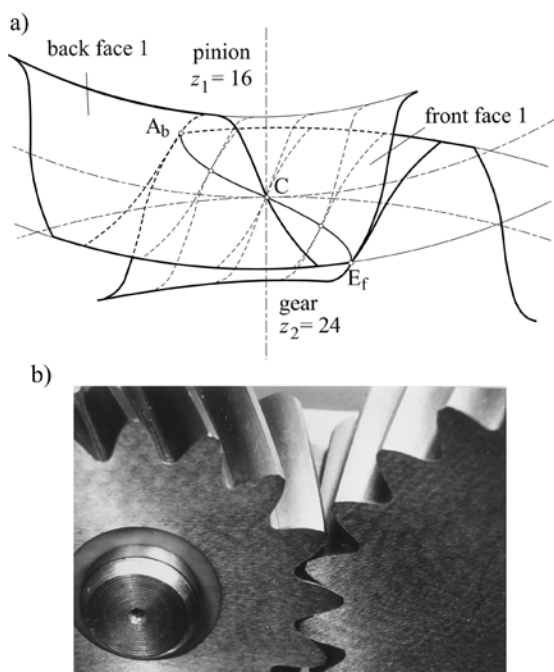
Fig. 13. Flash temperatures during tests; a) involute gears, and b) S-Gears

The heat load of the gears was estimated on the basis of oil temperature measurements [76] and [77]. S-gears with a modulus of 4.575 mm and involute gears with a modulus of 4.5 mm, and the number of teeth of 16 and 24 were used in the tests. The oil temperature at the beginning of the test was 90 °C. Oil temperatures after 15 minutes of operation at a given load in the standard FZG scuffing test are shown in Table 1. The measured temperatures prove that the heat load in S-gears is lower than in involute ones.

Table 1. Oil temperature in °C at the end of the test at each load step

Load step	7	8	9	10	11	12
Torque T_1 [Nm]	183	239	302	372	450	543
S-gears	94	97	102	106	113	118
Involute gears	99	103	107	112	120	127

One of the disadvantages of S-gears is a lower contact ratio than with involute gears. This weakness can be eliminated by using helical S-gears depicted in Fig. 14a. A gear pair with helical S-gears is shown in Fig. 14b. These gears were manufactured with the same shaping tool as the test spur S-gears. No experimental research on these gears has been made yet, but it is expected that operating characteristics will be better with helical gears than with spur S-gears.

**Fig. 14.** Helical S-gears; a) contact of the teeth, and b) a gear pair

The S-gears can also be used in internal gears, which are widely used in planetary gears. The analysis of the load capacity of the internal S-gears is discussed by Hlebanja and Hlebanja [78]. Since the path of contact is a straight line in the pitch point region (which is similar to involute gears), the S-gears are significantly less sensitive to centre distance errors [79] than classic cycloid gearing. The operating properties of the S-gears depend on parameters a_p and n . The influence of these parameters on the shape of teeth and operating properties is discussed by Kulovec and Duhovnik [80]. The S shape of the path of contact

was also used in worm gears. Theoretical analysis of the situation and demonstration of the operation is given by Hlebanja et al. [81].

In the previous decade, the concept of using polymer S-gears has emerged. Polymer gears have a low modulus of elasticity, so the deformations are significantly higher. Most gears are made by injection moulding, with which it is not possible to achieve high precision. The contact temperature (which depends on the friction coefficient) has a significant impact on damage to polymer gears. The theoretical contact temperature of polymer S-gears is lower than that of involute gears with comparable geometry (same modulus and number of teeth), so the expected load-bearing capacity of S-gears is higher. In the first phase of research, the gears were made by injection moulding. The advantages of S-gears have not been demonstrated due to poor manufacturing accuracy and large deformations. In the second phase of research, the gears were made by hobbing. In this case, a higher load capacity of the S-gears was measured. Results on research concerning polymer S-gears were published in several papers [82] to [85]. A comparison of the efficiency of S-gears and involute gears was made. The results show better efficiency of S-gears [86]. Problems of polymeric S-gears are large deformations, small size of teeth (test gears had a module of 1 mm) and manufacturing accuracy. The problem is general and specific to plastic S-gears. In steel gears, the deformations do not have such an effect on meshing. The accuracy class of the geometry is also higher for steel gears. Advantages of S-gears can be seen only if manufacturing accuracy is high enough, which can be achieved by hobbing. The question is, whether energy savings and higher load capacity justify the higher cost of manufacturing polymer S-gears.

3.5.2 Cosine Gearing

The shape of the flank of a pinion with a cosine gearing is described by a cosine function. Luo et al. [5] derived equations for the description of the tooth flank and the path of contact. A 3D model of the gearing was made, and a numerical analysis was performed with FEM. Contact ratio and sliding coefficient were calculated. Compared to involute gearing, the cosine gearing, which is discussed in [5], has the following characteristics: lower contact pressure (by approx. 22 %), lower root stress (by approx. 35 %), lower contact ratio (by 20 %) and considerably smaller sliding coefficient, both on gear and pinion. The stated properties of the gearing are valid generally, but the values in parentheses depend on the geometric

parameters of the gears. Wadagaonkar and Shinde [87] confirmed lower contact stresses (by approx. 30 %) and lower bending stresses (by approx. 50 %) for their geometry of cosine gears.

3.5.3 Gearing with a Parabolic Path of Contact

Wang et al. [6] worked on gears with a parabolic path of contact. A mathematical model of the gearing was derived on the basis of a known path of contact. The influence of the parabola parameters on the shape of teeth of the pinion and the gear was analysed. The number of teeth in which dedendum undercut occurs is significantly smaller than in the case of involute gearing. A numerical comparison of contact stresses and bending stresses between the involute gearing and the new gearing was made; both are lower in the new gearing. This general statement is valid for any form of a parabola, while the quantitative values depend on the parabola parameters that can be optimized. Properties of a gearing with a parabolic path of contact have not yet been experimentally researched.

3.5.4 Gears with a Constant Relative Radius of Curvature

The relative radius of curvature has a decisive effect on the Hertzian contact pressure (Eq. (1)). Liu et al. [88] developed a new form of non-involute gearing based on a controlled relative radius of curvature. It was named constant relative curvature (CRC) gears. Equations were developed to calculate the profile of the rack cutter and gears. A mathematical model for the gearing was defined with a constant relative radius of curvature along the path of contact. The tooth profile is shown in Fig. 15. CRC and involute gears were compared analytically. The contact ratio was calculated for both types of gears; it was slightly lower for CRC gears. Contact stress, sliding coefficients, and oil film thicknesses were determined. The gearing has similar advantages over the involute gears as the

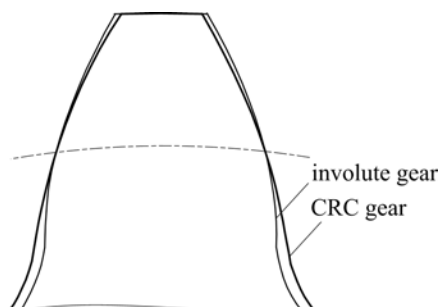


Fig. 15. Tooth profile of CRC gear and involute gear; adapted from [88]

S-gears: lower contact stress, thicker oil film and less sliding of flanks. The advantages of CRC gears have not yet been confirmed experimentally.

4 DISCUSSION

Gears in modern gearboxes must have the best possible efficiency, high load capacity, and low noise level combined with the smallest possible size. In recent decades, research has focused mainly on improving the properties of involute gears. The effects of geometry, lubrication, manufacturing accuracy and new materials on gearing load capacity and noise were researched in detail. New findings in tribology have significantly contributed to the improvement of the operational characteristics of involute gears.

Research on non-involute forms is, therefore, less frequent and less up-to-date. The development of software increased the possibilities of accurate analyses of non-involute forms. Mathematical models for defining geometry parameters have been developed to describe various non-involute shapes mentioned in this paper. A mathematical model of the gearing enables the creation of a 3D model of the gear pair, which can be analysed numerically. Analyses are relatively inexpensive and quickly feasible on modern computers. Most analyses are performed for the ideal shape of the tooth flanks. The accuracy of manufacturing and deformations of teeth significantly affect the shape of teeth and the operating characteristics. Involute and non-involute gears can only be compared if their manufacturing accuracy is comparable.

Based on the research of gears with a curved path of contact and the results of research on non-involute gears conducted and published by other authors, Table 2 assesses the properties of non-involute gears. The comparison refers to a comparable involute gear (the same number of teeth, helix angle, modulus, and accuracy class).

The three black dots in Table 2 indicate that the non-involute gearing is slightly better than the involute one in terms of the given property, while four and five dots indicate that it is significantly better than the involute one. At two black dots or less, the non-involute gearing is worse than the involute one.

At the Faculty of Mechanical Engineering, University of Ljubljana, research was carried out on steel gears with a progressively curved path of contact [7]. Prototypes of S-gears and comparable involute gears were made. A theoretical comparison was performed according to the criteria described in Chapter 2. The theoretical assumptions were

Table 2. Evaluation of the properties of non-involute gears compared to involute gears (●●●● best; ○○○○ worst)

	Surface durability	Root load capacity	Gear efficiency, Heat load	Contact ratio, Vibration and noise	Sensitivity to manufacturing and assembly errors	References
Cycloid gears	●●●○	●●●○	●●●○	●●○○	○○○○	[1], [3], [42]
Lantern gears	●●○○	●●○○	○○○○	●●○○	●○○○	[1], [3]
Arc-tooth-trace cycloid gears	●●●○	●●●○	●●●○	●●○○	●●○○	[43]
Original Novikov gears	●●●○	●●●○	●●●○	●○○○	○○○○	[44] to [58], [60]
New W-N gears	●●●○	●●●○	●●●○	●●○○	●●○○	[36], [59], [62]
Stepped triple circular-arc gears	●●●○	●●●○	●●●○	●●○○	●○○○	[61]
UPT Gears	●●●○	●●●○	●●●○	●●○○	●●○○	[63]
Double circular arc-involute tooth profile	●●●○	●●●○	●●○○	●●○○	●●○○	[64]
Pure rolling helical gears	●●●○	●●●○	●●●●	●●○○	●●○○	[65], [66], [67]
Convoloid gears	●●●○	●●●○	●●●○	●●○○	●●○○	[68]
S-Gears	●●●○	●●●○	●●●○	●●○○	●●○○	[4], [7], [8], [69] to [72], [75] to [86]
Cosine gears	●●●○	●●●○	●●●○	●●○○	●●○○	[5], [87]
Gears with a parabolic path of contact	●●●○	●●●○	●●●○	●●○○	●○○○	[6]
CRC-Gears	●●●○	●●●○	●●●○	●●○○	●●○○	[88]

confirmed by tests at the FZG test rig. In the last decade, extensive research has also been conducted on plastic S-gears [8]. In papers on cosine gears [5], gears with a parabolic path of contact [6] and CRC gears [87], researchers listed similar properties as were found in S-gears.

Non-involute gears are relatively well researched analytically and numerically in terms of load capacity, while vibration and noise analyses remain lacking. We also lack experimental confirmations of theoretical assumptions regarding load capacity, efficiency, and noise.

Table 3. Areas of use of some non-involute gears

	Areas of use
Cycloid gears	watches, roots blowers, gear pumps
Cycloidal-pin drive	robots (large gear ratio)
Lantern gears	transport devices (large gear ratio, small speeds)
W-N gearing	aircraft industry (helicopters), oil industry
Convoloid gearing	wind turbines
S-Gears	rolling mill, wind turbines, planetary gearboxes, precision mechanics
Cosine gearing	gear pumps
Gearing with a parabolic path of contact	gear pumps

Areas of use of some non-involute gears are listed in Table 3. The gears presented in recent years are still in the research phase and have not yet been used in practical applications.

5 CONCLUSIONS

The paper presents both non-involute gears with a long history (cycloid gears, classical Novikov gears) and gears first presented in the last decade. New technologies (3D printing) enable easy and fast prototyping and testing of the functionality of new designs. Unfortunately, prototypes are not enough to convince users that new designs are useful.

Different gears with a curved path of contact, which show similar operating properties compared with involute gears (higher root load capacity and surface durability, lower tooth sliding and heat load, lower contact ratio), are discussed in more detail. Theoretical findings for spur S-gears have also been experimentally confirmed. Vibration and gear noise research is lacking. A higher contact ratio can be achieved with helical S-gears, and it would make sense to research them experimentally in the future.

Experimental studies on gears manufactured in real-life accuracy and surface roughness are rare. We believe that it is not possible to succeed in the market with a gearing that is not supported by experiments.

Therefore, the geometry and manufacturing of involute gears are much simpler and, above all, cheaper than that of non-involute gears. The evaluation of involute gears is carried out according to standards and recommendations. The references only list ISO standards and ISO technical specifications, but national standards (e.g., AGMA, DIN, etc.) and recommendations also exist (e.g., VDI).

Because it is much more difficult to achieve the same accuracy with non-involute gears, they are still limited to specific applications. The current situation in development and research indicates that non-involute gears cannot displace involute gears on a large scale.

6 ACKNOWLEDGMENT

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Vsebina

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Razširjeni povzetki (extended abstracts)

- Andraž Maček, Janez Urevc, Miroslav Halilović: Prepoznavna oblike planih preizkušancev z uporabo optičnih metod merjenja celotnih polj deformacij in poravnava oblike z uporabo minimizacije odstopanja preslikave SI 29
- Abdulbasit Mohammed, Hirpa G. Lemu, Belete Sirahbizu: Določitev optimalne oblike lopatic za sistem črpanja vode na vetrno energijo na izbranih lokacijah SI 30
- Mingxing Han, Yinshui Liu, Yitao Liao, Shucui Wang: Raziskava modela in dinamičnih lastnosti novega proporcionalnega hidravličnega ventila, ki ga poganja zvočniška tuljava SI 31
- Mohsen Soori, Mohammed Asmael: Zmanjšanje preostalih napetosti in napak odklona z virtualnim petosnim rezkanjem turbinskih lopatic SI 32
- Mário Santos, Jaime Santos: Slabitev ultrazvočnih valov v nodularni litini z razpršitvijo: eksperimentalna študija in simulacija SI 33
- Ivan Okorn, Marko Nagode, Jernej Klemenc: Obratovalne lastnosti zunanjih neevolventnih valjastih zobnikov z ravnimi in poševnimi zobmi: pregled SI 34

Prepoznavna oblike planih preizkušancev z uporabo optičnih metod merjenja celotnih polj deformacij in poravnava oblike z uporabo minimizacije odstopanja preslikave

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Raziskava predstavlja razvito metodologijo za prostorsko poravnavo oblike merjenega planega preizkušanca z obliko računskega modela. Prostorska poravnava predstavlja ključen del najnovejših postopkov inverzne identifikacije parametrov konstitutivnih modelov trdnin, ki temeljijo na meritvah celotnega polja deformacij preizkušancev s heterogenim odzivom. Poravnava je namreč nujno potrebna, saj moramo med postopkom inverzne identifikacije primerjati izmerjene in izračunane odzive preizkušancev v istih (materialnih) točkah.

Razvita metodologija obsega tako prepoznavo oblike merjenega preizkušanca iz fotografij merilnega sistema korelacije digitalnih slik (DIC) kot tudi postopek prostorske poravnave. S praktičnega vidika je to ključnega pomena, saj moramo prepoznati in poravnati obliko preizkušanca, ko je ta že na lokaciji in pripravljen za meritve deformacij. Torej ima preizkušane v tem stanju že nanešen naključni vzorec, ki je potreben za izvedbo meritev deformacij z merilnim sistemom DIC in je vpet v trgalni stroj. Razvita metoda omogoča prostorsko poravnavo meritev, kjer je preizkušane opazovan bodisi z eno kamero (ravninsko 2D merjenje), bodisi z več. Prepoznavna oblike merjenega preizkušanca temelji na primerjavi dveh fotografij preizkušanca. Prva fotografija je zajeta na klasičen način, tako kot fotografiramo preizkušane za potrebe postopka DIC. Pred zajemom druge fotografije pa dodatno osvetlimo preizkušane še iz zadnje smeri. Tako zajamemo dve fotografiji, ki se razlikujeta v osvetlitvi ozadja preizkušanca. S prepoznavo slikovnih točk, katerih sivinska vrednost se med slikama ne spremeni veliko, določimo obliko merjenega preizkušanca. Prostorska poravnava te oblike z obliko preizkušanca uporabljenega v izračunu temelji na preslikavi pozicije slikovnih točk na zajetih fotografijah v prostor, kjer se nahaja oblika računalniškega modela preizkušanca. Ta preslikava je pravzaprav sestavljena iz dveh preslikav. Prva predstavlja umerjanje merilnega sistema in preslika pozicijo slikovnih točk (vrstica in stolpec na zajeti fotografiji) v pozicijo koordinatnega sistema preizkušanca (X in Y os ležita v ravnini preizkušanca in predstavljata realne mere, npr. milimeter). Druga preslikava pa nadalje preslika XY pozicije v prostor računalniškega modela preizkušanca (xy). Ta v osnovi predstavlja le transformacijo togega telesa, torej premik in zasuk točk v ravnini preizkušanca. Tako lahko preslikamo lokacije vseh slikovnih točk fotografij v prostor računalniškega modela preizkušanca. Točke, ki na fotografiji predstavljajo preizkušane in istočasno v prostoru modela predstavljajo ozadje ali obratno, predstavljajo točke neujemanja. Večje kot je število točk, večje je neujemanje med oblikama preizkušancev. Sama poravnava je izvedena v obliki optimizacijskega problema, ki išče parametre omenjene transformacije togega telesa s ciljem minimizacije števila točk neujemanja.

Robustnost metode je prikazana na primeru preizkušanca, katerega oblika spominja na strukturo metamateriala, in primerjana s splošno znano iteracijsko metodo poravnave najbližjih točk (ICP). Izkaže se, da novo razvita metoda potrebuje manj iteracij za zagotovitev poravnave preizkušanca. Uspešno pa uspe izvesti poravnavo tudi pri različnih stopnjah začetnega neujemanja in ob prisotnosti geometrijskih napak ter šuma, kjer ICP lahko konvergira k napačni poravnavi preizkušanca.

Uspešnost metode poravnave je prikazana tudi na primeru realnih meritev heterogenega preizkušanca, kjer smo obravnavali primer merjenja z le eno kamero in primer merjenja s štirimi kamerami. V obeh primerih je razvita metoda zagotovila odlično poravnavo oblik preizkušanca.

Razvita metoda predstavlja veliko praktično uporabnost v vseh primerih inverzne identifikacije na osnovi meritev celotnega polja deformacij planih preizkušancev in predstavlja novost na omenjenem področju, saj avtomatsko prepozna obliko preizkušanca in na edinstven način poskrbi za poravnavo z upoštevanjem celotne oblike preizkušanca.

Ključne besede: meritve polja deformacij, korelacija digitalnih slik, prepoznavna oblike preizkušanca, poravnava oblik, inverzna identifikacija, heterogeno polje deformacij

Določitev optimalne oblike lopatic za sistem črpanja vode na vetrno energijo na izbranih lokacijah

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Pričujoči članek obravnava rezultate optimizacije oblike in študije zmogljivosti rotorskih lopatic, ki so primerne za delo v šibkem vetru. Vetrnice so namenjene za neposredni pogon vodnih črpalk na treh izbranih ruralnih lokacijah v Etiopiji.

Rotorske lopatice vetrnic za tri izbrane lokacije so bile oblikovane na podlagi aerodinamičnega profila SG6043 ter podatkov o hitrosti vetra na teh lokacijah. Izračunani so bili aerodinamični profili rotorskih lopatic, ki zagotavljajo maksimalen koeficient moči oziroma razmerje med dejansko močjo vetrnice in razpoložljivo močjo vetra na izbranih referenčnih območjih. Optimizacija lopatic s ciljem doseganja maksimalnega izkoristka energije vetra je bila opravljena z različnimi parametri, kot so oblika lopatic, tetive, razmerje med obodno hitrostjo in hitrostjo vetra, geometrije, nastavljeni koti itd. Optimizacija lopatic je potekala na osnovi teorije gibalne količine lopatic (BEM) v programski opremi QBlade, ki je bila razvita posebej za načrtovanje lopatic vetrnih turbin. Za iteriranje oblike je bila uporabljena koda v programskem paketu MATLAB®, ki je bila razvita za ta namen.

Na podlagi raziskav so bili določeni optimizirani profili lopatic za vetrne razmere na lokacijah v Abomsi, Metehari in Ziwayju na jugovzhodu Etiopije. Sistemi za črpanje vode na treh lokacijah lahko učinkovito delujejo z vetrnicami s tremi lopaticami premera 10,74 m, 7,34 m oz. 6,34 m. Največje vrednosti koeficienta moči $C_p(r_i)$ znašajo 0,4512, 0,4587 in 0,4627 pri največjih vrednostih razmerja med obodno hitrostjo in hitrostjo vetra $\lambda(r_i)$ 3,0, 3,5 oz. 4,0.

Članek obravnava načrtovanje, analizo in optimizacijo rotorskih lopatic za določitev optimalnih parametrov zmogljivosti pri specifičnih lokalnih vetrnih razmerah, vključno z velikostjo rotorja in profili. Rezultate numerične analize bo treba še dodatno preveriti in validirati v okviru teh raziskav. Na podlagi konstrukcijskih parametrov bo razvit model celotne vetrne turbine in zgrajene bodo prototipne turbine za izvajanje preskusov na treh lokacijah.

Glavni prispevek dela je v praktičnih možnostih izvedbe. Razvite metode projektiranja rotorskih lopatic za učinkovito izkoriščanje vetrnih zmogljivosti za pogon sistemov za črpanje vode, zlasti na območjih z nizko hitrostjo vetra, imajo velik pomen za izboljšanje kakovosti življenja na ruralnih območjih v Etiopiji, ki nimajo dostopa do električne energije za oskrbo s pitno vodo. To je zelo pomembno za velik delež prebivalstva te države.

Ključne besede: metoda robnih elementov, dimenzioniranje vetrnic, vetrna energija, hidravlična moč, koeficient moči, sistem za črpanje vode

Raziskava modela in dinamičnih lastnosti novega proporcionalnega hidravličnega ventila, ki ga poganja zvočniška tuljava

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Proporcionalni hidravlični ventil kot glavna krmilna komponenta v vodni hidravliki ima pomembno vlogo pri delovanju krmiljenja sistemov.

Konstruiranje visokozmogljivih vodnih proporcionalnih hidravličnih ventilov je težavna naloga zaradi nizke viskoznosti in oksidativnih lastnosti medija. V procesu konstruiranja je treba razrešiti veliko težav, kot sta npr. slabo mazanje in visoko trenje. Upor na krmilnem batu zaradi nizke viskoznosti in slabe mazalnosti vode ni le velik, ampak tudi nelinearen. Delovanje oljnih razvodnih ventilov drsniškega tipa je idealno, medtem ko se pri krmilnih batih podobnih vodnih ventilov v pogojih vodnega mazanja zaradi neoptimalne kombinacije materialov (jeklo/jeklo) pojavljajo oscilacije. Za visoko dinamično zmogljivost je zato potrebna optimizacija zgradbe hidravličnega proporcionalnega ventila in elektromehanskega aktuatorja.

Pričujoča študija predstavlja nov hitroodzivni vodni proporcionalni hidravlični ventil. Aktuator ventila je izveden kot zvočniška tuljava (VCM), katere prednosti so v hitrem odzivu, visoki natančnosti krmiljenja in majhnih izmerah. Za hitro krmiljenje ventila je bil zasnovan ročni ojačevalnik, ki zagotavlja zadostno silo za nastavljanje. Razvit je bil podroben in natančen nelinearen matematični model ventila, ki upošteva parametre konstrukcije ventila in elektromagnetne lastnosti VCM. Izvedene so bile podrobne simulacije delovanja, ki so vključevale elektromagnetne simulacije, analizo lastnosti ročnega ojačevalnika in simulacijo dinamične zmogljivosti ventila.

Rezultati simulacij kažejo, da znaša nastavitveni čas približno 28 ms, največji prenehaj pa približno 5 %. Čas odziva na stopnico znaša približno 15 ms. Zgrajeni sta bili tudi preizkuševališči za ventil in za VCM. Rezultati preskusov prototipa kažejo, da je optimalno območje hoda VCM med 4 in 15 mm. Največji prenehaj ventila znaša okrog 10 %, nastavitveni čas pa je približno 30 ms pri odpiranju in 35 ms pri zapiranju. Rezultati preskusa dokazujejo, da ima ventil dobro statično in dinamično zmogljivost krmiljenja. Trenutno še ni primerjav med oljnimi in vodnimi hidravličnimi ventili glede vpliva pomanjkljivega mazanja na dinamični odziv na stopnico. To vprašanje je zelo zanimivo in bo navdih za boljše raziskave, ki jih še načrtujejo avtorji.

V študiji je podan predlog vodnega proporcionalnega hidravličnega ventila s hitrim odzivom. Prednosti VCM, ki sta izkoriščeni pri elektromehanskem pretvorniku ventila, sta visoka hitrost in natančnost krmiljenja. Nizka viskoznost in mazalnost vodnega medija zahtevata veliko silo za premikanje krmilnega bata. Za ojačevanje sile VCM je bil uporabljen ročni ojačevalnik. Dinamična zmogljivost in odpornost ventila proti motnjam sta bili izboljšani z regulacijo položaja v zaprti zanki s povratno zvezo. Rezultati testov dokazujejo, da ventil ohranja visoko odzivnost pri različnih tlakih. Pri novem hidravličnem ventilu z VCM je bistveno izboljšana dinamična zmogljivost v primerjavi s tradicionalnimi vodnimi proporcionalnimi ventili. VCM je tako lahko dobra zamenjava za elektromagnete pri tovrstnih ventilih.

Ključne besede: proporcionalni krmilni ventil, zvočniška tuljava, optimizacija, dinamična zmogljivost

Zmanjšanje preostalih napetosti in napak odklona z virtualnim petosnim rezkanjem turbinskih lopatic

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Preostale napetosti v delih, izdelanih s postopki odrezavanja, so posledica trenja, oblikovanja odrezkov in nastanka toplote v območju odrezavanja. Sile in povišane temperature med odrezavanjem lahko na obdelanih turbinskih lopaticah povzročijo napake odklona, ki pa morajo biti zaradi zahtevane natančnosti izdelka čim manjše. Cilj pričujoče študije je bil izboljšanje natančnosti in zanesljivosti obdelanih turbinskih lopatic z minimizacijo preostalih napetosti in napak odklona v virtualnem obdelovalnem sistemu.

Z uporabo takega sistema je mogoče napovedati in zmanjšati preostale napetosti in napake odklona po petosnem rezkanju lopatic. Z analizo po metodi končnih elementov (MKE) so bile določene rezalne sile in temperature za vsak položaj rezalnega orodja na poti obdelave. Za optimizacijo parametrov obdelave z vidika preostalih napetosti in napak odklona je bil nato uporabljen genetski algoritem.

Virtualni obdelovalni sistem je izračunal delovno temperaturo in rezalne sile pri danih parametrih v vsaki točki na poti obdelave. Izračunane sile in temperature so bile uvožene v analitično programsko opremo Abaqus R2016X FEM za analizo preostalih napetosti in napak odklona po obdelavi. Raziskovalno delo je bilo nato validirano s primerjavo preostalih napetosti, določenih eksperimentalno po metodi rentgenske difrakcije, in rezultatov virtualnega obdelovalnega sistema. Napake odklona so bile določene z meritvijo obdelanih lopatic na koordinatnem merilnem stroju.

Preiskava fizično in virtualno obdelanih lopatic je pokazala 24,2-odstotno oz. 23,9-odstotno zmanjšanje preostalih napetosti po uporabi optimiziranih rezalnih parametrov. Zmanjšanje napake odklona je bilo 26,3-odstotno oz. 24,1-odstotno.

Predstavljeni pristop bo mogoče uporabiti pri obdelavi površin letalskih aerodinamičnih profilov na petosnih CNC-obdelovalnih strojih za zmanjšanje preostalih napetosti.

Analiza literature je pokazala, da napovedovanje in zmanjševanje preostalih napetosti in napak odklona pri rezkanju z virtualnimi obdelovalnimi sistemi ni dovolj raziskano. Razviti virtualni obdelovalni sistem bo tako lahko pripomogel k izboljšanju natančnosti in zanesljivosti lopatic.

Ključne besede: virtualna obdelava, preostale napetosti, napaka odklona, optimizacija parametrov, turbinska lopatica, petosni CNC-obdelovalni stroji

Slabitev ultrazvočnih valov v nodularni litini z razpršitvijo: eksperimentalna študija in simulacija

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Nodularna železova litina je razširjena zaradi svoje livnosti, visoke toplotne prevodnosti in dobrih mehanskih lastnosti, predvsem natezne trdnosti in duktilnosti. Mehanske lastnosti kovin so v veliki meri odvisne od njihove mikrostrukture. Nodularna litina se proizvaja z dodajanjem manjših količin elementov tik pred strjevanjem, npr. magnezija in cerija. Ti dodatki povzročijo rast grafitnih vozlov, njihova oblika in porazdelitev pa sta odločilni za lastnosti kovine. Neporušne preiskave mikrostruktur so zelo pomembne za identifikacijo osnovne mikrostrukture in vozlov.

Ultrazvočna karakterizacija ponuja vrsto prednosti v primerjavi s porušnimi metalografskimi metodami. Interakcijo ultrazvočnih valov z mikrostrukuro je mogoče spremljati z akustičnimi parametri, kot sta hitrost in slabitev. Obstajata dva glavna mehanizma slabitve ultrazvočnih valov: absorpcija in razpršitev. Absorpcija je povezana s toplotnimi izgubami, histerezo in viskozni izgubami. Do razpršitve prihaja zaradi heterogenosti v obliki mej med zni, praznin, vključkov, delcev drugih faz in poroznosti. Prav ta mehanizem slabitve je najbolj pomemben pri heterogenih materialih, kot je železova litina.

V pričujočem delu je bila eksperimentalno in s simulacijami ocenjena slabitev ultrazvočnih valov zaradi razpršitve na strukturah s kompleksno porazdelitvijo, ki je značilna za železovo litino. Uporabljena je bila eksperimentalna določitev slabitve po impulzni metodi. V ta namen so bili obdelani odboji ultrazvočnih impulzov na prednjih in na zadnjih ploskvah šestih preizkušancev prizmatične oblike. Preizkušanci so bili pripravljani po standardnih metalografskih metodah in analiza optičnih mikroposnetkov je pokazala nizko prostorsko variabilnost porazdelitve velikosti vozlov. Slabitev z razpršitvijo je bila ocenjena po modelih Truella in Papadakisa, rezultati pa so bili primerjani z rezultati eksperimentov. Poudarek je bil na modelu Papadakisa, ki upošteva porazdelitev velikosti sipalcev. Rezultati obeh analitičnih modelov se močno razlikujejo od eksperimentalnih.

Za analizo vpliva velikosti in koncentracije sipalcev v preučevanih strukturah, ki sta pomembni za učinkovit razvoj modelov, so bila pri izvedbi simulacij uporabljena orodja MATLAB k-Wave. Ta simulacijska orodja uporabljajo preprost napovedovalni model na osnovi rešitve v časovni domeni po psevdospektralni metodi v k-prostoru za povezovanje akustičnih enačb prvega reda za homogene in heterogene medije v eni, dveh in treh razsežnostih. Paket k-Wave omogoča modeliranje poljubnih virov in zaznavanje površin z usmerjenimi elementi. Pri širjenju valov je mogoče upoštevati nelinearnost, heterogenosti in zakon absorpcije.

Simulator, ki je bil razvit za preizkušance, je omogočil prediktivni model, v katerem je slabitev sorazmerna s peto potenco velikosti sipalcev, formulacija pa je podobna tisti, ki jo predlagajo analitični modeli. Predhodno eksperimentalno pridobljeno znanje o slabitvi valov omogoča ustvarjanje simulacijskih modelov s spreminjanjem koncentracije sipalcev in enakovredne velikosti. Na ta način je omogočeno posnemanje struktur s kompleksnimi porazdelitvami sipalcev z vidika njihove enakomerne koncentracije in velikosti.

Ključne besede: modeliranje, anizotropija, impulzna ultrazvočna metoda, simulacija, slabitev ultrazvoka

Obratovalne lastnosti zunanjih neevolventnih valjastih zobnikov z ravnimi in poševnimi zobmi: pregled

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Namen preglednega znanstvenega članka je bil podati sistematičen pregled neevolventnih zobnikov, njihovih prednosti in slabosti ter možnosti uporabe v praktičnih aplikacijah. V strokovni in znanstveni literaturi nismo našli celovitega pregleda od starih oblik, kot so cikloidno ozobje in Novikov ozobje, do ozobji, ki so bila razvita v zadnjem desetletju. Želja avtorjev je bila pripraviti kritičen pregled, ki bo uporaben tako za raziskovalce novih oblik neevolventnih zobnikov kot za uporabnike le-teh v industriji.

Večina zobnikov v industrijski praksi ima evolventno obliko bokov zob. Z vidika bočne nosilnosti, obrabe in izkoristka ta oblika ni optimalna. Ključne slabosti evolventnega ozobja so: omejeno minimalno število zob pastorka, visoka kontaktna obremenitev zaradi konveksno-konveksnega dotika bokov zob in neugodne kinematske razmere (velike drsne hitrosti) na začetku in koncu ubiranja. Številni raziskovalci so razvili in analizirali različne neevolventne oblike bokov zob, ki se niso širše uveljavile v praksi. Glavna razloga sta stroški izdelave in občutljivost na točnost izdelave in montaže.

V preteklih desetletjih so bile raziskave večinoma usmerjene v izboljšavo lastnosti evolventnih zobnikov. Podrobno so bili raziskani vplivi geometrije, mazanja, točnosti izdelave in novih materialov na nosilnost in hrupnost ozobja. Ker v članku primerjamo evolventne zobnike z neevolventnimi, so uvodoma povzete glavne lastnosti evolventnih zobnikov. Pojasnjene so možnosti za izboljšanje njihovih obratovalnih lastnosti (uporaba profilnih premikov, profilne korekcije, nesimetrično ozobje).

Ocena lastnosti evolventnih in neevolventnih ozobji temelji na uveljavljenih kriterijih, kot so Hertzov kotalni tlak, temperatura v dotiku, debelina oljnega sloja, upogibna napetost v korenu zoba in hrupnost ozobja. V drugem poglavju so kriteriji za oceno podrobno predstavljeni. Na podlagi enačb, ki popisujejo razmere v dotiku bokov zob, so pojasnjeni trendi, ki jih raziskovalci upoštevajo pri razvoju neevolventnih zobnikov. Pri oblikovanju neevolventnih zobnikov je cilj doseči čim večji srednji krivinski radij, kar posledično pomeni zmanjšanje Herzovega tlaka, zmanjšanje temperature v dotiku in povečanje debeline oljnega sloja. Drugi ključni cilj je čim manjše relativno drsenje v dotiku, kar prispeva k manjši toplotni obremenitvi zobnikov in boljšemu izkoristku. Hrupnost zobnikov je manjša, če je stopnja prekrivanja večja. Večina neevolventnih zobnikov ima manjšo stopnjo prekrivanja, kar negativno vpliva na dinamične sile in hrupnost.

V tretjem poglavju so opisno in s slikami predstavljene različne oblike neevolventnih zobnikov. Povzeti in komentirani so rezultati raziskav iz navedene literature. Podrobneje so predstavljeni rezultati raziskav na S-zobnikih, ki so bili razviti in preizkušani na Fakulteti za strojništvo Univerze v Ljubljani. S preizkusi S-zobnikov in primerljivih evolventnih zobnikov je bila potrjena večja bočna nosilnost in manjša toplotna obremenitev S-zobnikov. Potencial na področju S-zobnikov predstavljajo zobniki s poševnimi zobmi, ki pa še niso bili eksperimentalno raziskani. V četrtem poglavju je združen pregled lastnosti predstavljenih neevolventnih zobnikov v primerjavi z evolventnimi zobniki. Uporabljeni kriteriji so bočna nosilnost, korenska nosilnost, toplotna obremenitev, hrupnost in občutljivost ozobja na napake izdelave in montaže. Večina neevolventnih zobnikov ima večjo korensko in bočno nosilnost ter manjšo toplotno obremenitev.

Za širšo uveljavitev neevolventnih zobnikov v industrijski praksi bodo potrebne nadaljnje eksperimentalne potrditve teoretičnih predpostavk. Z razvojem novih tehnologij in materialov se uporabnost neevolventnih zobnikov sicer povečuje. Ker pa je pri neevolventnih zobnikih mnogo težje doseči enak razred točnosti, so le-ti še vedno omejeni na specifične primere uporabe. Trenutno stanje razvoja in raziskav kaže, da neevolventni zobniki ne morejo izpodrinuti evolventnih v večjem obsegu.

Ključne besede: neevolventni zobniki, profil boka zoba, ubirница, S-zobniki, cikloidno ozobje, Novikov ozobje, nosilnost ozobja

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Contents

Papers

- 203 Andraž Maček, Janez Urevc, Miroslav Halilovič:
Flat Specimen Shape Recognition Based on Full-Field Optical Measurements and Registration Using Mapping Error Minimization Method
- 214 Abdulbasit Mohammed, Hirpa G. Lemu, Belete Sirahbizu:
Determining Optimum Rotary Blade Design for Wind-Powered Water-Pumping Systems for Local Selected Sites
- 223 Mingxing Han, Yinshui Liu, Yitao Liao, Shucui Wang:
Investigation on the Modeling and Dynamic Characteristics of a Novel Hydraulic Proportional Valve Driven by a Voice Coil Motor
- 235 Mohsen Soori, Mohammed Asmael:
Virtual Minimization of Residual Stress and Deflection Error in the Five-Axis Milling of Turbine Blades
- 245 Mário Santos, Jaime Santos:
Ultrasonic Scattering Attenuation in Nodular Cast Iron: Experimental and Simulation Studies
- 256 Ivan Okorn, Marko Nagode, Jernej Klemenc:
Operating Performance of External Non-Involute Spur and Helical Gears: A Review